#### MONITORING IMPACT DAMAGE IN COMPOSITES WITH LARGE AREA SENSING SKINS

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#### INTRODUCTION

Impact damage detection is a multi-faceted problem

- Can induce a permanent loss in the toughness
- NDT methods commonly incur non-trivial opportunity costs while parts are imaged
  - Ultra-sonic
  - Acoustic Emission



Ultrasonic image of composite blowout damage courtesy of iMAP's lab U of SC.



#### INTRODUCTION

- The soft elastomeric capacitor (SEC) is a flexible electronic capable of monitoring strain over large areas
  - Respond to changes in the sensor geometry
  - Inherits the mechanical properties of an elastomer
  - Functions as a parallel plate capacitor





#### BACKGROUND

 $C = \mu_0 \,\mu_r \,\frac{A}{h}$  $\frac{\Delta C}{C_0} = \frac{\left(\frac{A_1}{h_1} - \frac{A_0}{h_0}\right)}{\frac{A_0}{h_0}} = \frac{A_1 h_0}{A_0 h_1} - 1$ 

Preservation of Mass:

$$\rho V_1 = \rho V_0 \rightarrow A_1 h_1 = A_0 d_0$$







#### BACKGROUND







$$\frac{\Delta C}{C_0} = \left(\frac{A_1}{A_0}\right)^2 - 1$$

$$\approx \left(\frac{A_0(\varepsilon_{11} + \varepsilon_{22} + 1)}{A_0}\right)^2 - 1$$

$$\approx \varepsilon_{11}^2 + \varepsilon_{22}^2 + 2(\varepsilon_{11} + \varepsilon_{22})$$

$$\approx 2(\varepsilon_{11} + \varepsilon_{22})$$

$$A_{1} = A_{0}(1 + \varepsilon_{11})(1 + \varepsilon_{22})cos(\gamma_{12})$$
$$A_{1} = A_{0}(\varepsilon_{11}\varepsilon_{22} + \varepsilon_{11} + \varepsilon_{22} + 1)$$
$$\approx A_{0}(\varepsilon_{11} + \varepsilon_{22} + 1)$$



#### MANUFACTURE



# MANUFACTURE

a) The dielectric is drop cast onto a glass pane

b)

a)

- b) The carbon black SEBS solution is then painted onto the dielectric in progressive layers
- c) Two copper tabs are used for metallic connections to connect to the data acquisition system

contacts

c)



# **PROPERTIES**

- The manufacture of the SEC makes the scaling of the sensor trivial
- The Elastomer matrix can extend up to 500% its original length





# **PROPERTIES**

- The sensor is quite notably resilient to damage
- Removing sections of the sensor hasn't unduly harm them



30 40 Time [s] 60

50

70

-10

10

20



# **MODEL ENERGY**

Assumptions in elastic impact: Initial impact energy is equal to the kinetic energy leaving the sample

Assumptions in plastic impact: Initial impact energy is equal to the kinetic energy minus the retained strain energy





#### **SENSING APPARATUS**



GFRP sample



# **DROP TOWER**

- GFRP nominal proof resilience of 0.678 J to 4.237 J
- Drop tower with a 6.5 kg impactor



# **DROP TOWER**







# **DROP TOWER**

GFRP nominal proof resilience of 0.678 J to 4.237 J

However, testing showed minor crack initiation after 1.32J though not a full failure of the plate

Testing range chosen of 0.25 J to 1.5 J over 5 impact event



#### RESULTS

- First row: the load cells measure for each individual impact
- Second row: the measured capacitance measures and where the data omits the 2 second before and after the impact
- Third row: the normalized change in capacitance of the region before and after.





#### RESULTS





## CONCLUSIONS

The GFRP can be expected to respond in three domains:

- Pure elastic responses that are non-hysteric
- Small plastic deformations due to failure in the resin
- Larger deformations post-fracture where the Glass fiber fails



### CONCLUSIONS



Impacts above the proof resilience are expected to store energy as deformations. These would be registered in the be the SEC as a sustained increase in capacitance where sub fracture impacts would be registered no sustained change in capacitance by the SEC (i.e. state-based damage detection).



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# FURTHER READING ON THE SEC

- Li, Jian, et al. Strain-based Fatigue Crack Monitoring of Steel Bridges using Wireless Elastomeric Skin Sensors. No. FHWA-KS-19-01. 2019.
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# THANKS!

Questions: Alexander Vereen

