



*Structural Dynamics and Acoustic Systems Laboratory
University of Massachusetts Lowell*

*Model Reduction/Expansion with Applications for
Linear and Nonlinear Dynamic Response
and Related Topics*

*Peter Avitabile
Professor Emeritus
Mechanical Engineering Department
Structural Dynamics and Acoustic Systems Laboratory (SDASL)
Modal Analysis and Controls Laboratory (MACL)
University of Massachusetts Lowell*





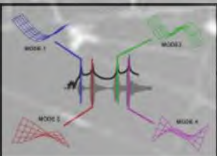
Structural Dynamics and Acoustic Systems Laboratory University of Massachusetts Lowell

Half a Century

How Some Old Stuff Can Still Be Useful



*Peter Avitabile
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Structural Dynamics Thrust Areas - Peter Avitabile

Test/Analysis Correlation Updating

MDM Reduced Order Updated Response Predictions

Field to Laboratory Fixture Neutralization

Model Reduction Expansion

VIKING/SERP Self-Smoothing

FRF Smoothing/Expansion

SERP SMOOTHING

DECAF

System Modeling

Modal Models

Reduced Models

Modal/Physical Models

Impedance Models

4 Bay Frame (Modal Model)

Linear & Nonlinear Response Modeling

Linear Modeling

Nonlinear Modeling

Full Field Optical Techniques

Damage Identification

$$DII = \frac{1}{N} \sum_{i=1}^N (|E_{E,LI}| - |E_{A,LI}|)^2$$

Force Reconstruction

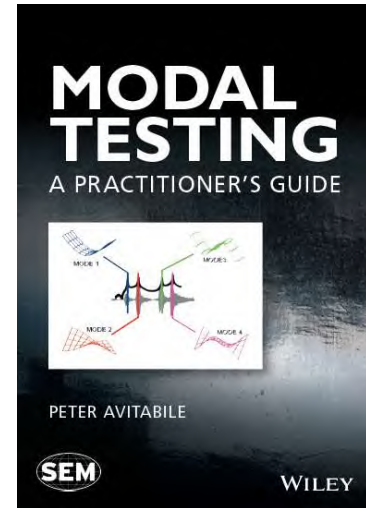
Dr. Peter Avitabile, PE



*Professor Emeritus, Mechanical Engineering
Research Professor, Michigan Technological U
Co-Director, Structural Dynamics
and Acoustic Systems Laboratory*

*President 2016/2017 & Vice President 2014
for Society for Experimental Mechanics
Associate Editor - Handbook of
Experimental Structural Mechanics*

*B.S.M.E., Manhattan College
M.S.M.E., University of Rhode Island
D. Eng., University of Massachusetts Lowell
Professional Engineer, Rhode Island*



Pete has five decades of experience in design and analysis using FEM and experimental techniques. His main area of research is structural dynamics specializing in the areas of modeling, testing and correlation of analytical and experimental models along with advanced applications for developing structural dynamic models. Pete has contributed over 300 technical papers in the area as well as his "Modal Space" article series in the Experimental Techniques magazine published by the Society for Experimental Mechanics for 17 years. He is the 2004 recipient of the prestigious SEM DeMichele Award and elected SEM Fellow in 2020. He is recognized worldwide as an expert in structural dynamic modeling applications. He often provides consulting services for a wide variety industries in these specialty areas of expertise.

Structural Dynamics and Acoustic Systems Laboratory



Modeling Tools



Testing Equipment

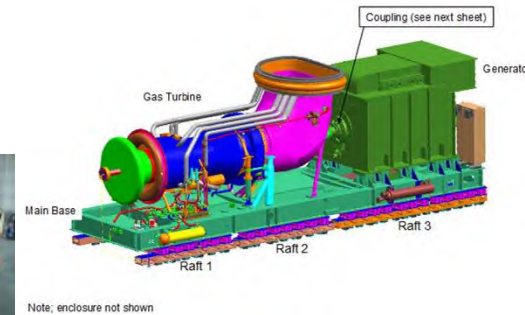


Experimental Tools

The Structural Dynamics and Acoustic Systems Laboratory is one of the best equipped labs in the country to do the work that the lab does.

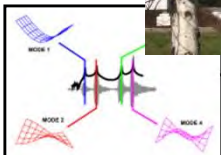
In addition to research efforts, the SDASL has been involved in many industry related structural dynamic tests due to their capability

- Performed detailed modal tests for several 50m+ wind turbine blades
- DDG 1000 Propulsion System for new Navy destroyer
- Optical & radio telescopes (Nobeyama, Gemini, Haystack)



Integrating Test and FEA for Enhanced Structural Dynamic Modeling Applications

Peter Avitabile - Mechanical Engineering
Structural Dynamics And Acoustic Systems Lab



Peter Avitabile Work Spans Many Industries and Applications

Modal/Operating Tests for System Characterization

Telescopes - 45m in Nobeyama, Japan & Gemini 8m in Chile/Hawaii

Historical Developments in MAEL/SDASL
Table of Contents

RADARSAT Experimental Modal Test and Correlation

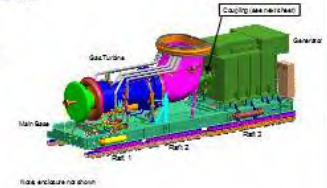
Large multichannel multiple input multiple output modal test

Historical Developments in MAEL/SDASL
Table of Contents

DDG 1000 Destroyer Propulsion System

A very large modal test conducted on DDG 1000 propulsion system - 150 ton, 50' x 20' x 12'

- test conducted in 22 continuous hr
- 150 measurement locations
- 18 separate references used

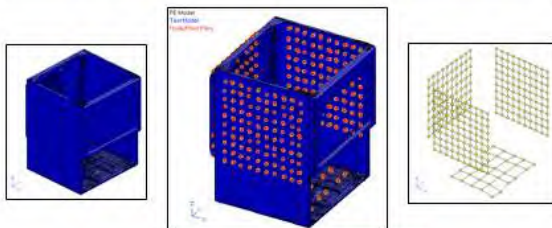


Extracted
two days
imposed

MAEL/SDASL 154 Peter Avitabile - Mechanical Engineering
Structural Dynamics And Acoustic Systems Lab



Test - Analysis - Correlation Whirlpool

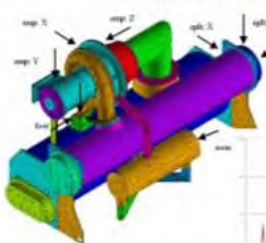


System Disassembly for Cascaded Component Targets

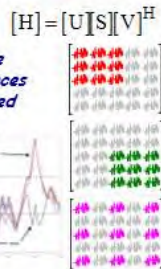
Components may be physical or reduced and may not have elemental topology

Optimal Reference Selection for Modal Testing

SVD of FRF matrix provides insight into optimal reference selection



Multiple references evaluated



Three selected for critical test for correlation with FEM

Historical Developments in MAEL/SDASL 121 Peter Avitabile - Mechanical Engineering
Structural Dynamics And Acoustic Systems Lab



ECF Project Work - Tuned Absorber Application

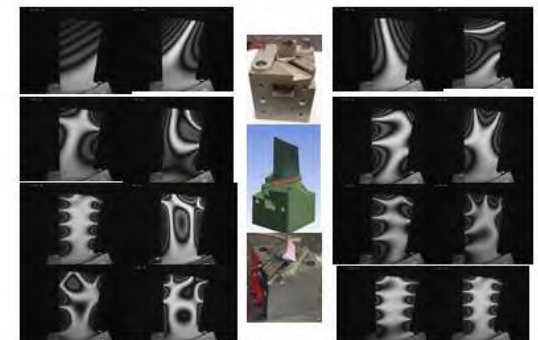
All FRFs synthesized at beam tip

Massingering has significant effect on the main structure

Tuned Absorber

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Structural Dynamics And Acoustic Systems Lab

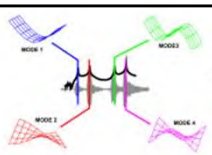
FAA High Frequency Blade Characterization



Historical Developments in MAEL/SDASL 100 Peter Avitabile - Mechanical Engineering
Structural Dynamics And Acoustic Systems Lab



Where is UMASS Lowell ???





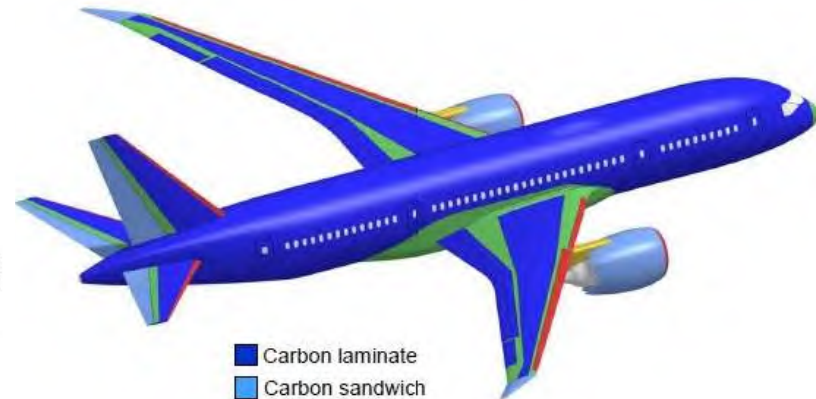
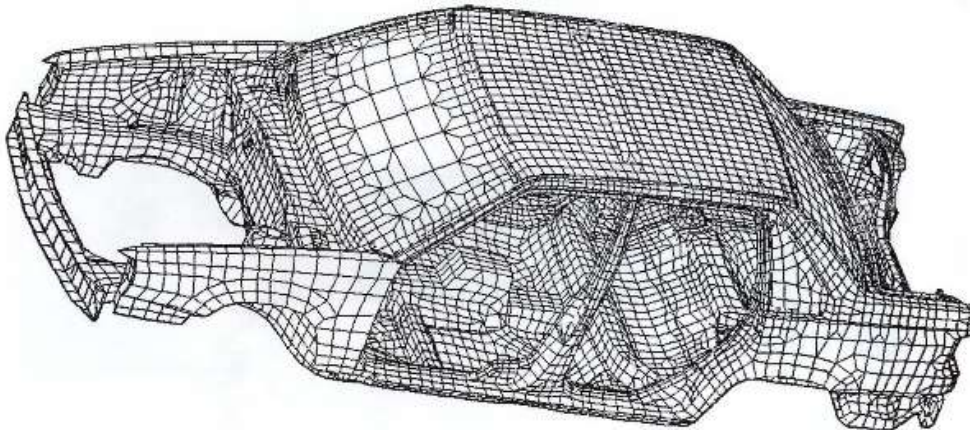
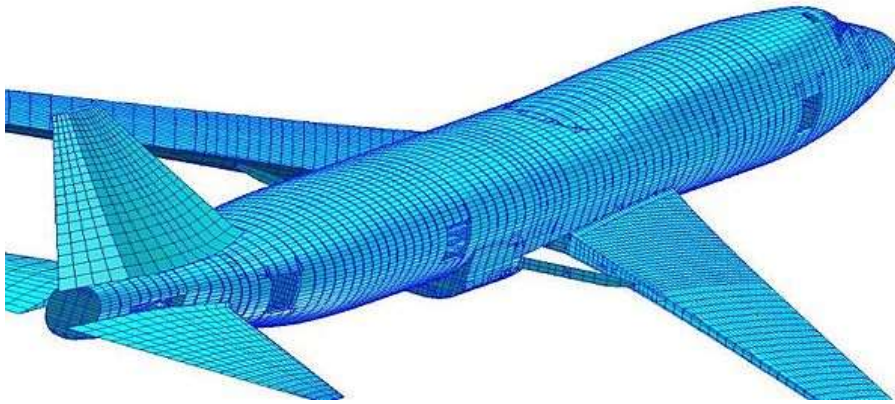
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Some general topics:

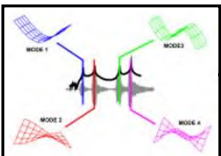
- Model Reduction and Expansion*
- Correlation/updating of structural dynamic models*
- Structural dynamic modification and system modeling*
- Highly reduced order analytical models for efficient response*
- Incorporating measured experimental data (from limited data) for full field dynamic response/strain*
- Utilizing linear modeling techniques to efficiently address nonlinear response applications*
- Damage detection utilizing dynamic response and dynamic strain from real-time operating data*
- Force Reconstruction - Beyond Measured Points*

Finite Element Modeling

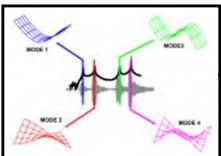
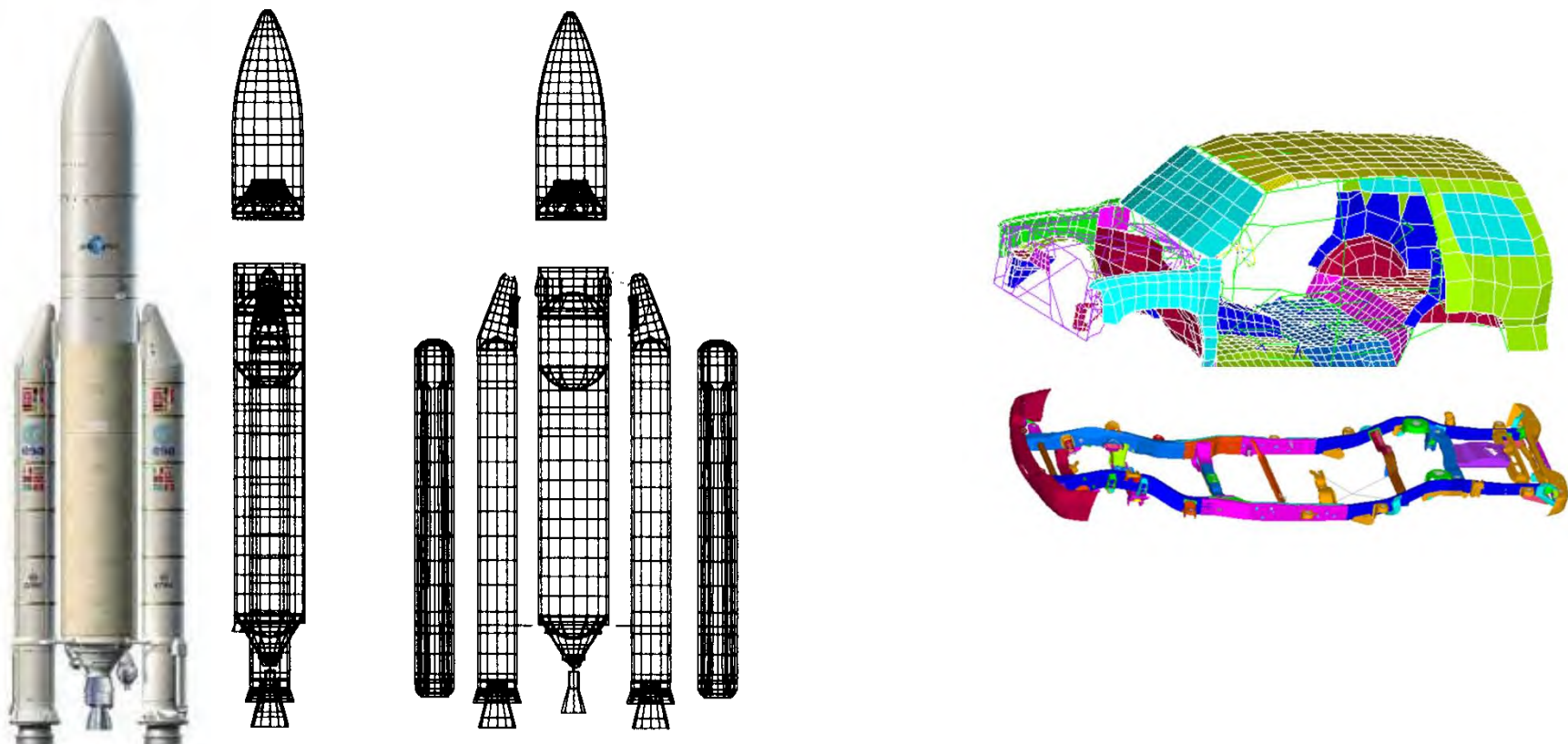
Obsessively Large Finite Element Models Dominate the Landscape Today (only because we think we need them to be this large)



- Carbon laminate
- Carbon sandwich
- Fiberglass
- Aluminum
- Aluminum/steel/titanium pylons

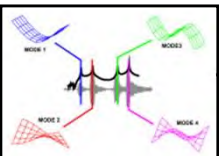


Models so large that reduced order models necessary



Larger Models vs Smaller Accurate Models

- Large models are often difficult to interpret*
- Especially true when multiple perturbations are evaluated*
- Often time smaller reduced modes are needed*
- Models must be accurate for any response studies*



Larger Models vs Smaller Accurate Models

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Sometimes it seems like we are going back to old approaches

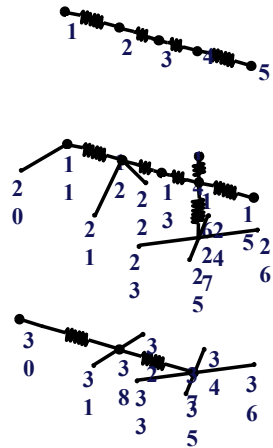


TECHNOLOGY IMPROVEMENTS IN MODELING

(or were the old techniques adequate?)

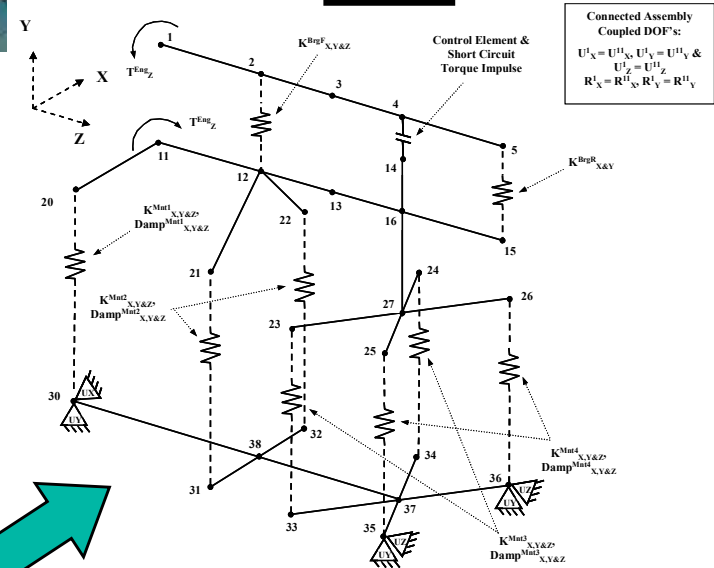
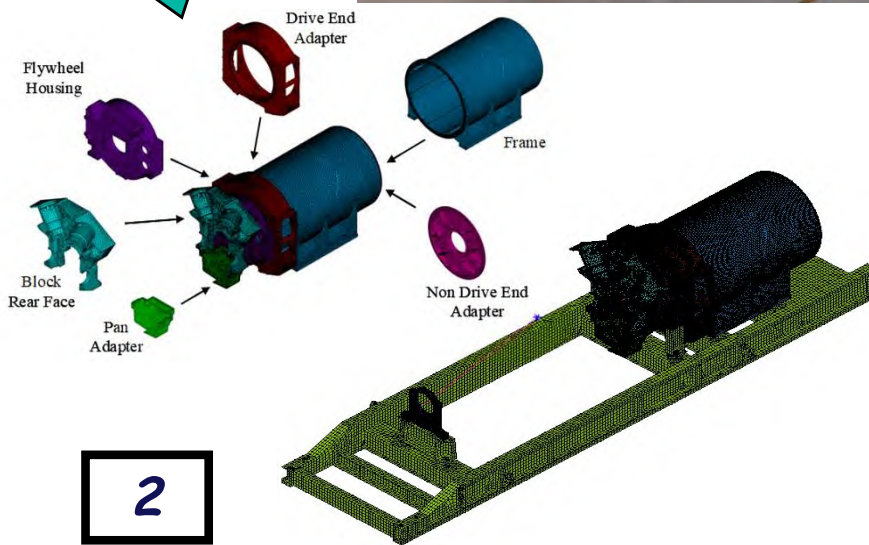
1

Prehistoric models were simple due to lack of computational ability



Simpler models are developed from detailed models to gain better understanding of effects observed from studies made

3



Connected Assembly Coupled DOF's:
 $U^1_X = U^{11}_X, U^1_Y = U^{11}_Y$ &
 $U^{12}_X = U^{12}_X, U^{12}_Y = U^{12}_Y$
 $R^1_X = R^{11}_X, R^1_Y = R^{11}_Y$

2

Improved computers and software allow development of sophisticated models

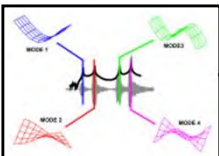


BACK TO THE FUTURE



Integrating Test and FEA for Enhanced Structural Dynamic Modeling Applications

Peter Avitabile - Mechanical Engineering
 Structural Dynamics And Acoustic Systems Lab



TECHNOLOGY IMPROVEMENTS IN MODELING

(or were the old techniques adequate?)

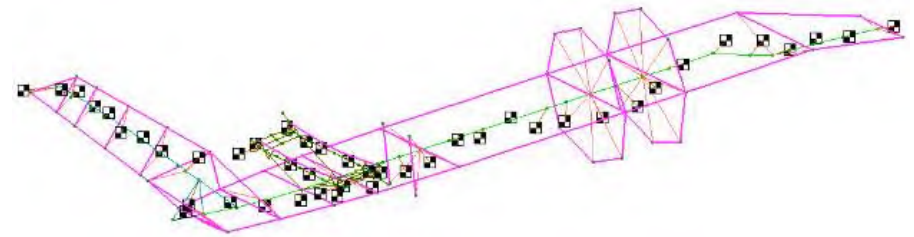
1

Early models were simple due to lack of computational ability



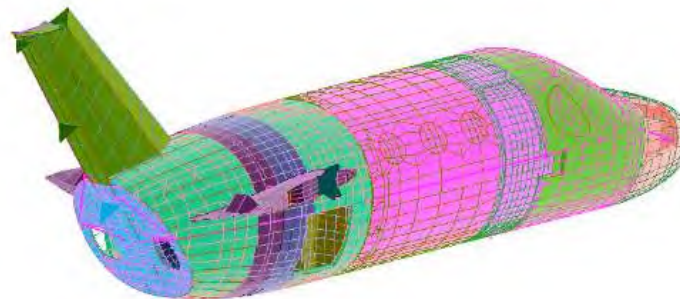
Computational restraints force use of simpler models

3



2

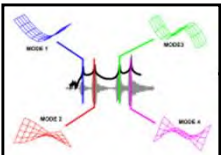
Improved computers and software allow development of sophisticated models



BACK TO THE FUTURE



Integrating Test and FEA for Enhanced Structural Dynamic Modeling Applications



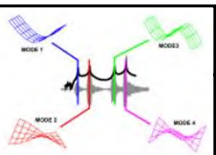
Larger Models vs Smaller Accurate Models

- Large models are often difficult to interpret
- Especially true when multiple perturbations are evaluated
- Often time smaller reduced modes are needed
- Models must be accurate for any response studies

Sometimes it seems like we are going back to old approaches



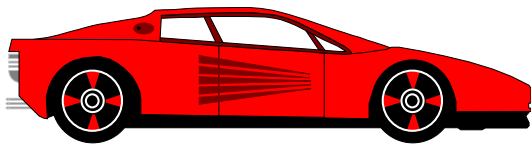
BTTF



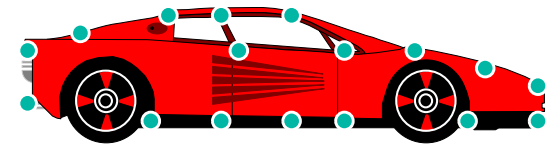
Model Reduction

Model Reduction - An Important Piece

Dynamic reduction means :



X_F = full set of dof's



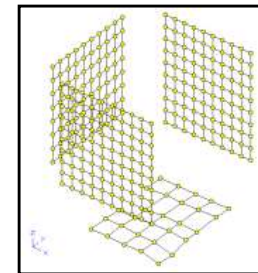
X_A = active set of dof's

*reducing a given dynamic finite element model
to one with fewer degrees of freedom
while maintaining the dynamic characteristics of the system.*

*The transformation T will take on various forms
depending on the transformation technique utilized*

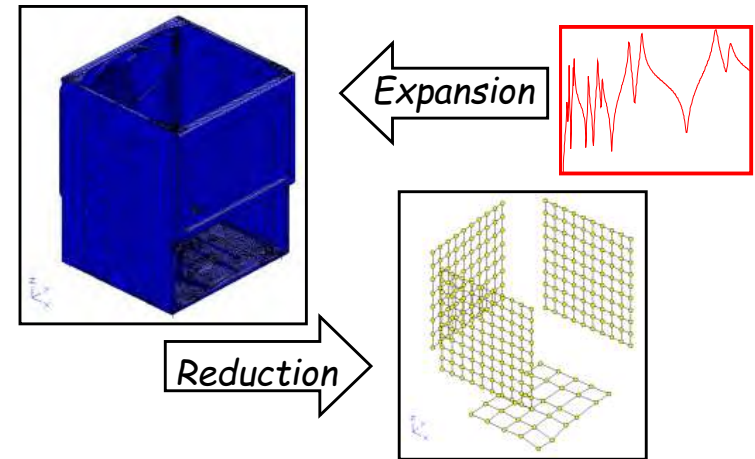


$$\{X_n\} = \begin{Bmatrix} X_a \\ X_d \end{Bmatrix} = [T] \{X_a\}$$



Model Reduction and Expansion

- *Model reduction - capture system dynamics at a subset of points*
- *Model expansion - obtain full field information from limited points*
- *The general transformation between the 'n' full space DOF and 'a' subset of DOF is*

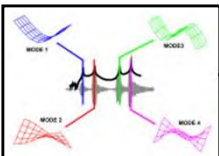


$$\{x_n\} = \begin{Bmatrix} x_a \\ x_d \end{Bmatrix} = [T]\{x_a\}$$

- *And the mass and stiffness matrices are written as*

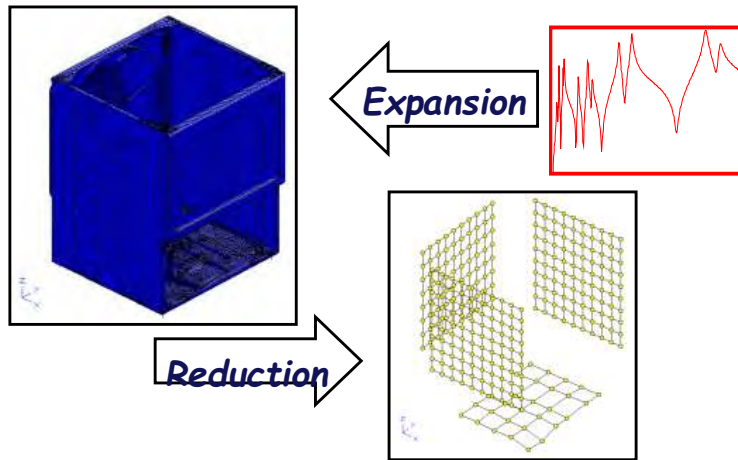
$$[M_a] = [T]^T [M_n] [T]$$

$$[K_a] = [T]^T [K_n] [T]$$



Model Reduction and Expansion

Reduction/Expansion are critical to many structural Dynamic modeling activities - correlation, modification, updating, ...



$$\{x_n\} = \begin{Bmatrix} x_a \\ x_d \end{Bmatrix} = [T] \{x_a\}$$

$$[M_a] = [T]^T [M_n] [T]$$

$$[K_a] = [T]^T [K_n] [T]$$

Guyan Condensation

$$[T_s] = \begin{Bmatrix} [I] \\ [t_s] \end{Bmatrix} = \begin{Bmatrix} [I] \\ -[K_{dd}]^{-1} [K_{da}] \end{Bmatrix}$$

Improved Reduced System

$$[T_i] = \begin{Bmatrix} [I] \\ [t_s] \end{Bmatrix} + [t_i] \quad [t_s] = -[K_{dd}]^{-1} [K_{da}]$$

$$[t_i] = \begin{Bmatrix} [0] & [0] \\ [0] & [K_{dd}^{-1}] \end{Bmatrix} [M_n] [T_s] [M_a]^{-1} [K_a]$$

Dynamic Reduction

$$[D_n] = [[K_n] + f[M_n]]$$

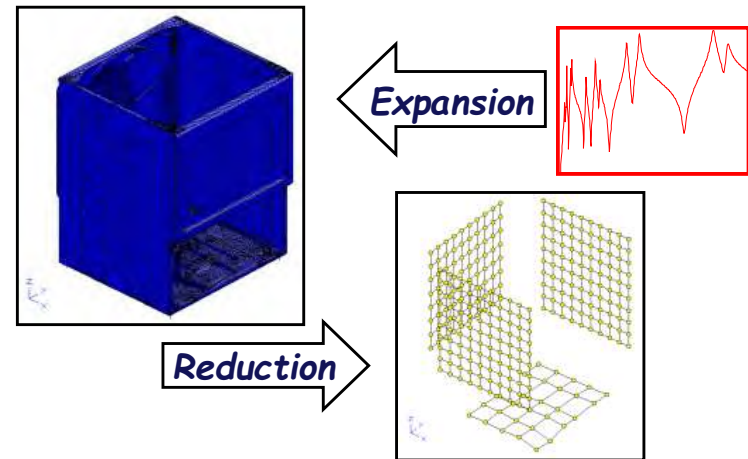
$$[T_f] = \begin{Bmatrix} [I] \\ [t_f] \end{Bmatrix} = \begin{Bmatrix} [I] \\ -[D_{dd}]^{-1} [D_{da}] \end{Bmatrix}$$

Model Reduction and Expansion

System Equivalent Reduction Expansion Process
SEREP is a significant breakthrough in reduction/expansion

$$\begin{Bmatrix} \{X_a\} \\ \{X_d\} \end{Bmatrix} = \{x_n\} = [U_n]\{p\} = \begin{bmatrix} [U_a] \\ [U_d] \end{bmatrix} \{p\}$$

$$\begin{Bmatrix} \{X_a\} \\ \{X_d\} \end{Bmatrix} = \begin{bmatrix} [U_a] \\ [U_d] \end{bmatrix} [U_a]^g \{x_a\}$$



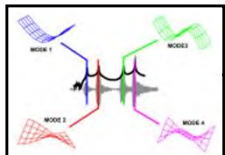
Reduced matrices saw tremendous computational savings

$$[M_a^S] = [T_U]^T [M_n] [T_U] = [U_a^g]^T [U_a^g] \quad [K_a^S] = [T_U]^T [K_n] [T_U] = [U_a^g]^T [\Omega^2] [U_a^g]$$

and breakthrough changes in correlation techniques

$$[U_a]^T [M_a] [E_a] \equiv [U_n]^T [M_n] [E_n] \equiv [U_a]^g [E_a]$$

https://faculty.uml.edu//pavitabile/downloads/IMAC7_SEREP.pdf



Model Reduction and Expansion - SEREP

- *The System Equivalent Reduction Expansion Process (SEREP) transformation is*

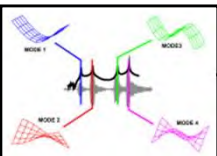
$$[T_U] = [U_n][U_a]^g$$

- *The eigenvectors & eigenvalues are preserved exactly using this technique*
- *The mass and stiffness are written as*

$$[M_a] = [T_U]^T [M_n][T_U] = [U_a^g]^T [U_a^g]$$

$$[K_a] = [T_U]^T [K_n][T_U] = [U_a^g]^T [\Omega^2 [U_a^g]$$

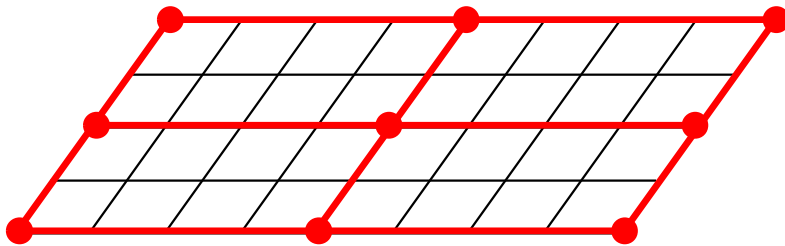
https://faculty.uml.edu//pavitabile/downloads/IMAC7_SEREP.pdf



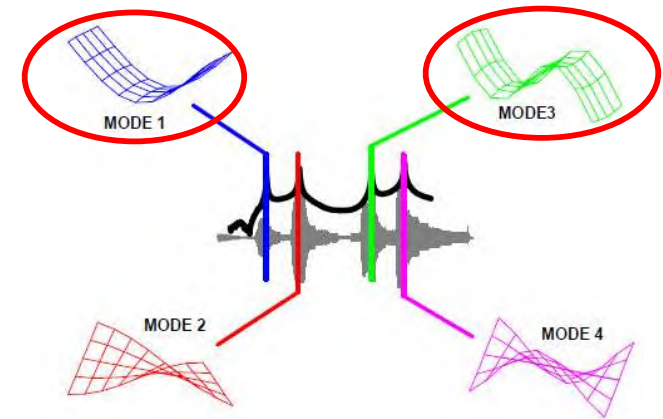
Efficient Time Response Calculation Techniques

- *Equivalent Reduced Model Technique (ERMT)*
- *Modal Modification Response Technique (MMRT)*

ERMT - Physical approach



MMRT - Modal approach

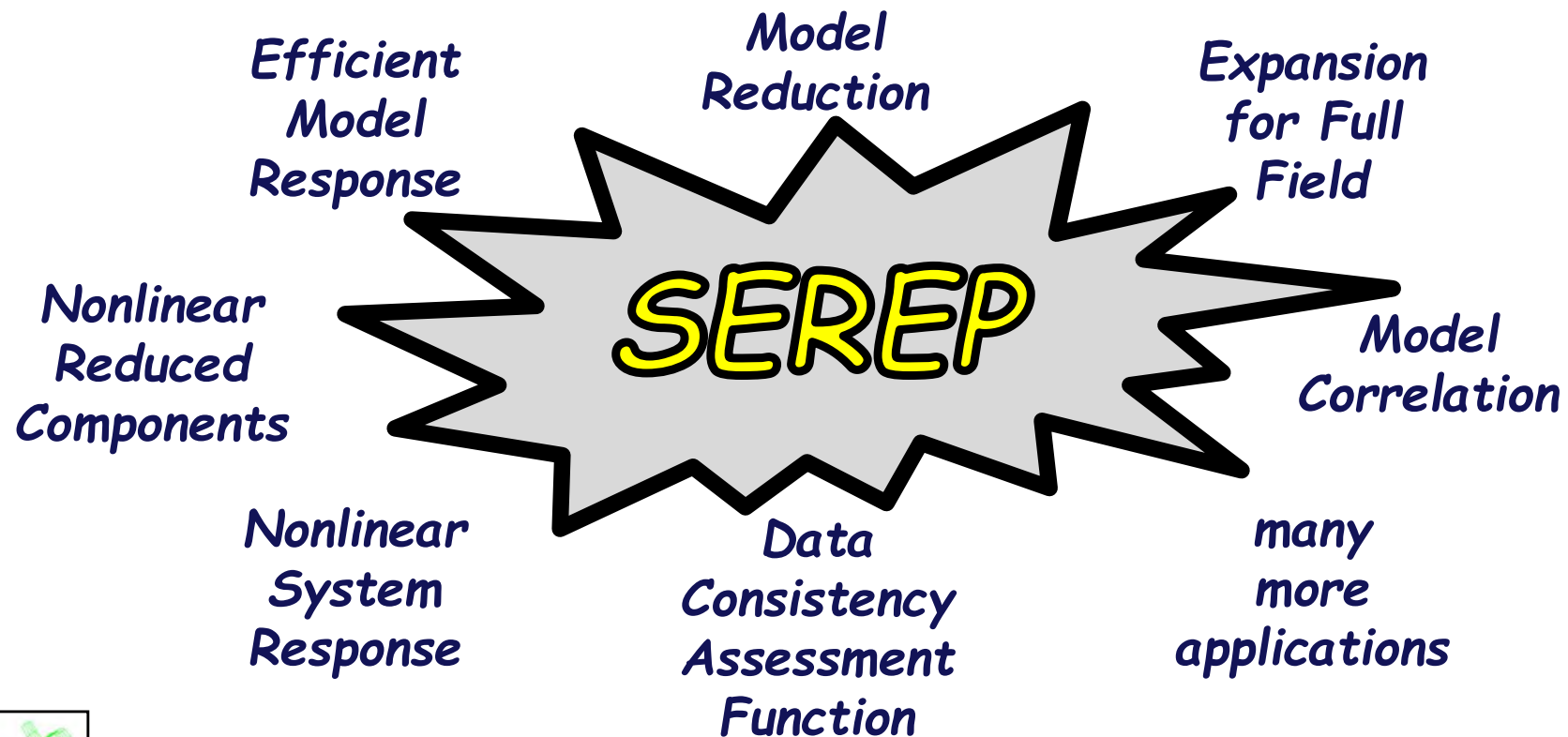


- *Numerical integration performed at each time step for the specific configuration*
 - *Newmark time integration*
 - *Wilson-Penzien damping formulation*

Theory - Model Reduction/Expansion

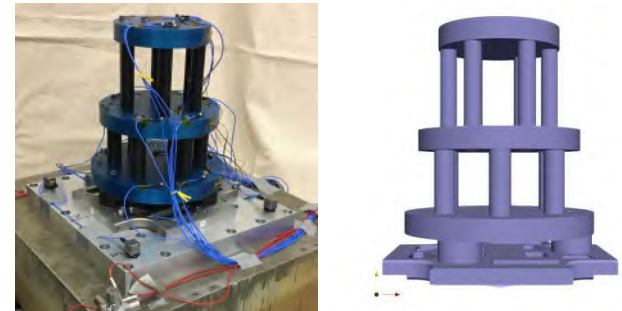
Reduction/Expansion are critical to many structural dynamic modeling activities - correlation, modification, updating, ...

The System Equivalent Reduction Expansion Process (SEREP) has been widely used in many applications

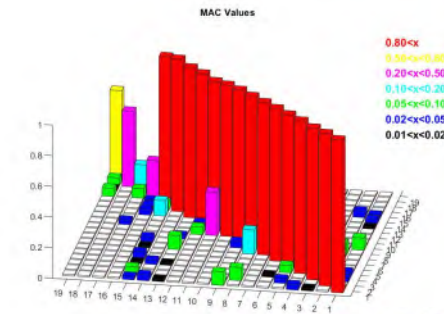


Model Correlation - SEREP-POC-CORTHOG

SEREP is widely used and has shown significant accuracy in many applications.

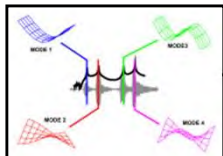
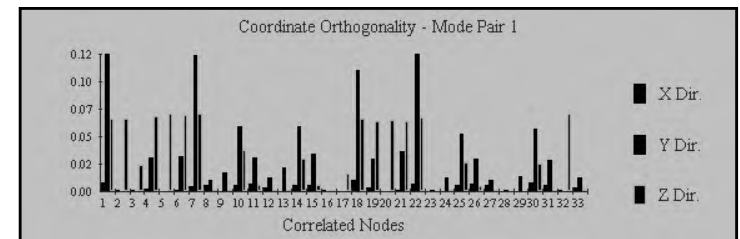


Pseudo Orthogonality Check allows for efficient mass orthogonality check without need for mass or stiffness matrices.



SEREP

CORTHOG is a mass scaled degree of freedom check for comparison to the FEA and indicates where errors exist.



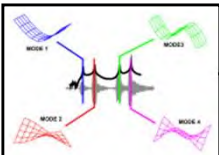
Model Expansion

Expansion as Important as Reduction

*Integrating Test and FEA for Enhanced
Structural Dynamic Modeling Applications*

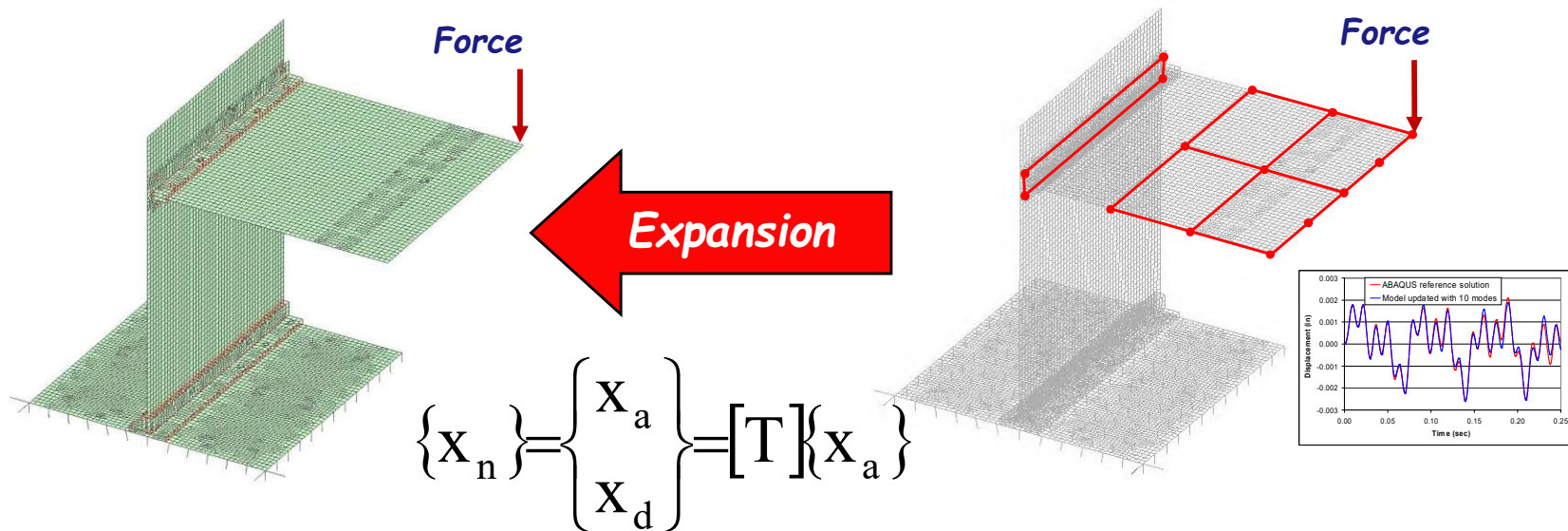
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Expansion Enables Additional Possibilities

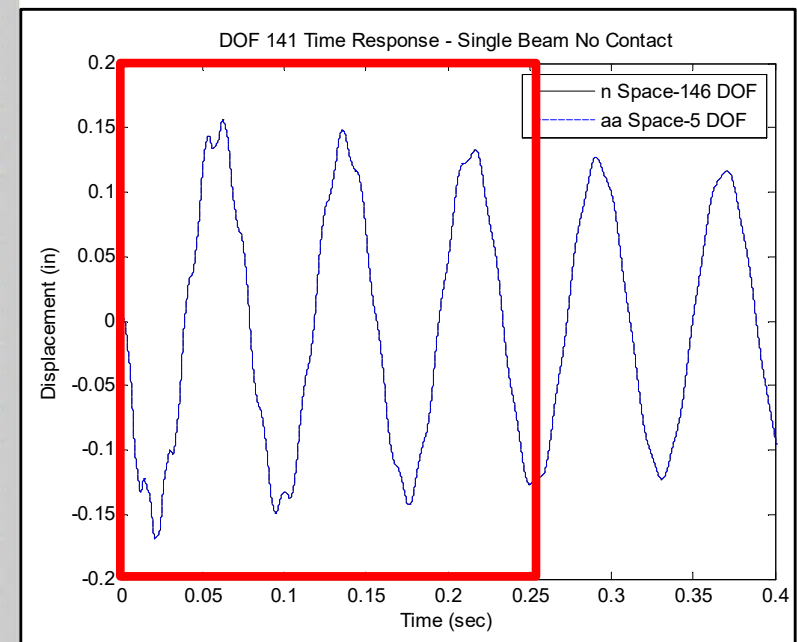
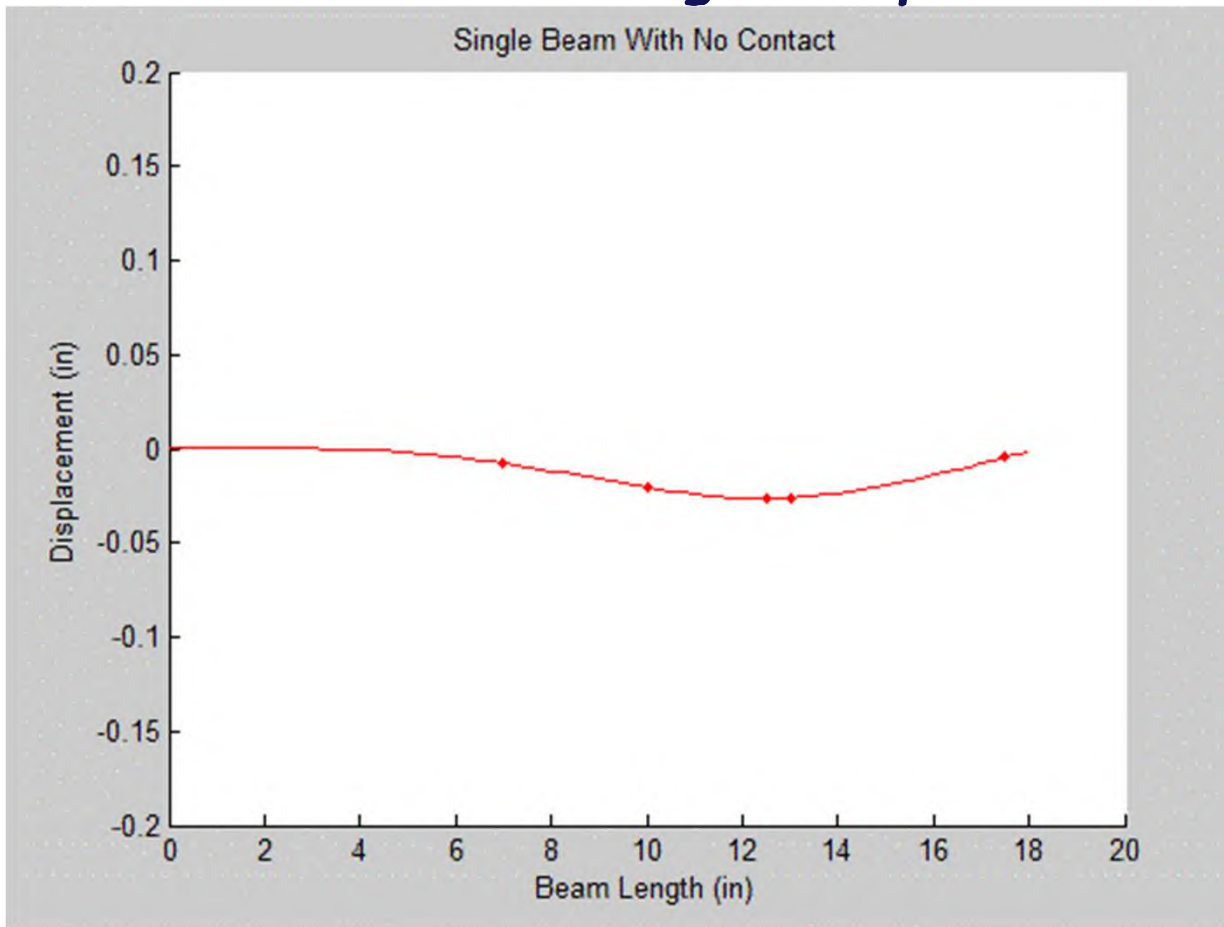
Using force or displacement at attachment points, expansion processes can be applied to identify full model response (dynamic stress/strain)



Formerly only used for linear components
 (but extensions to systems and nonlinear possible)

Full Field Response From Limited Set of Data

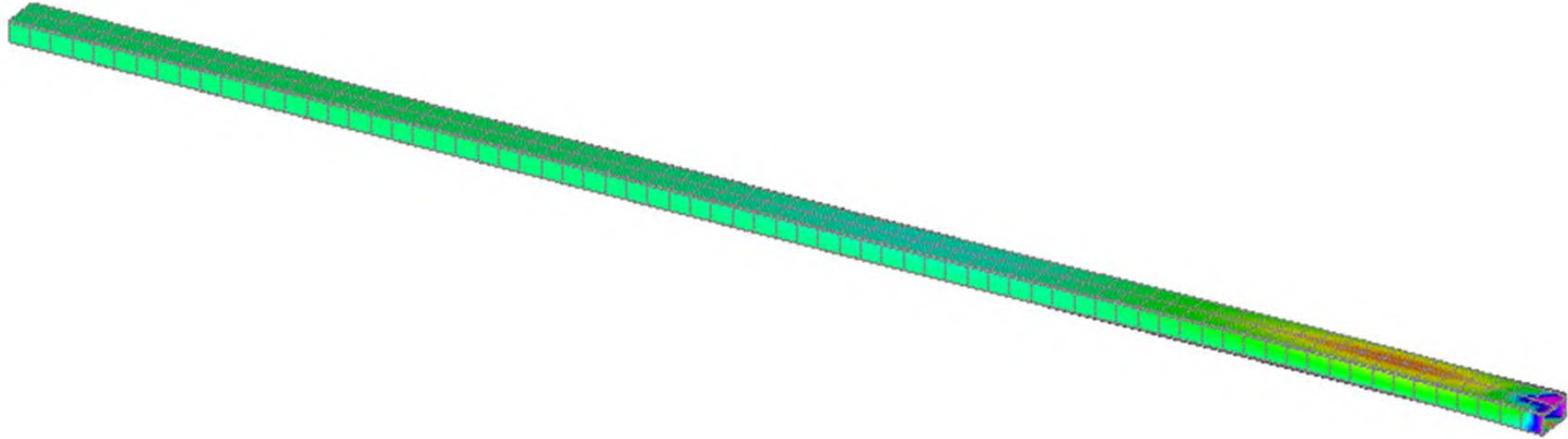
The full field displacement obtained from limited sets of data either for single component or for the nonlinear system



ANIMATION

Expansion Allows Full Field from Limited Data

**Eventual Aim:
Expansion of Nonlinear Displacements
to Compute Dynamic Stress-Strain**

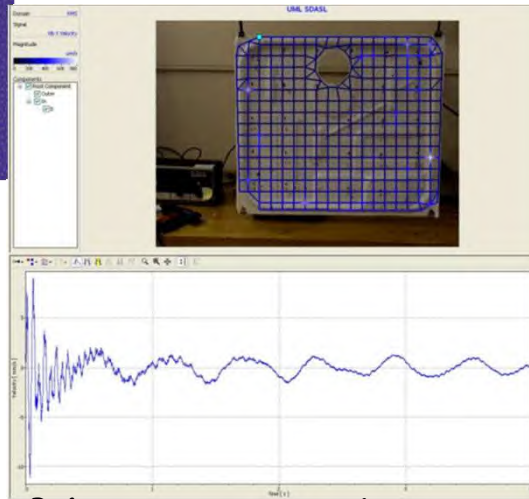
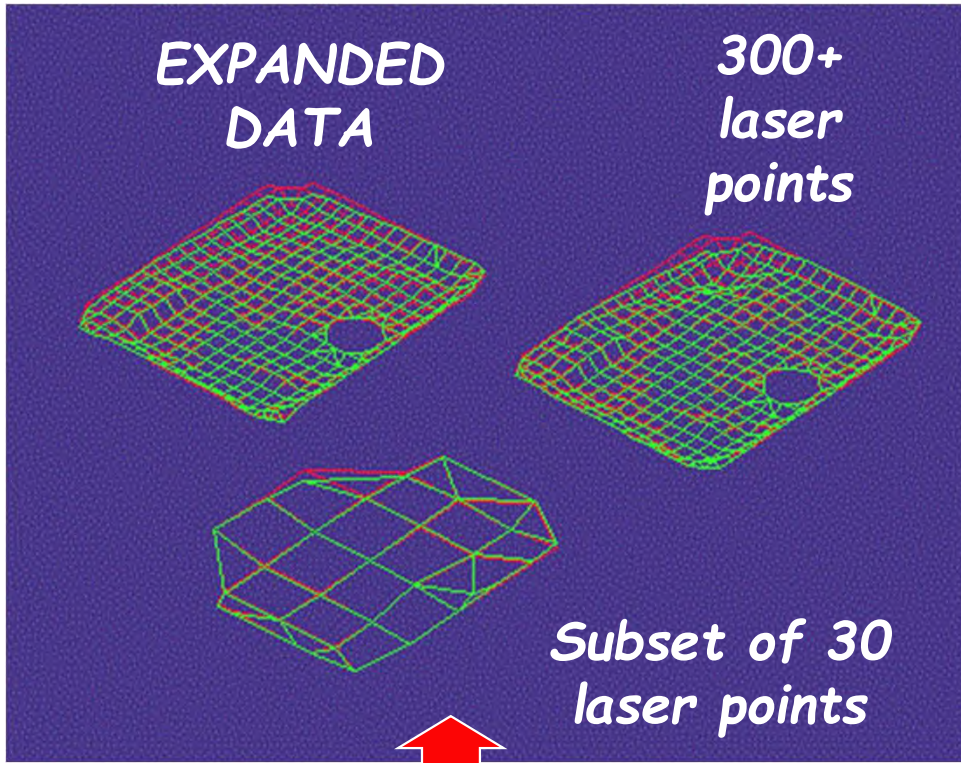
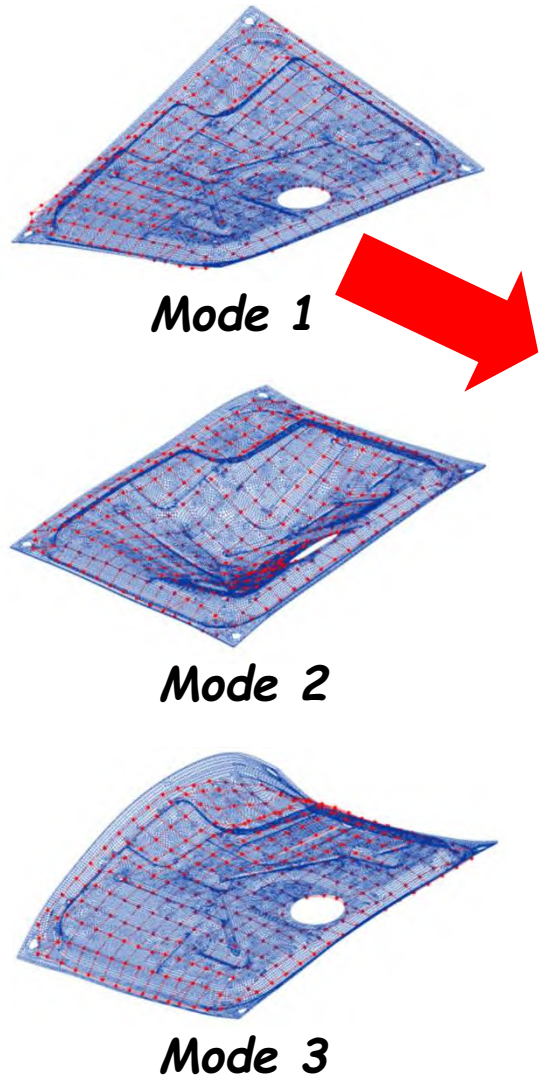


Proved that it can be done for nonlinear system

ANIMATION

Expansion Real Time Operating Data

**MODE
SHAPES
MATTER**



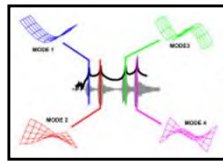
Polytec Scanning Vibrometer

'a' set
Limited Measurement Points

$$[T_U] = [E_n][E_a]^g$$

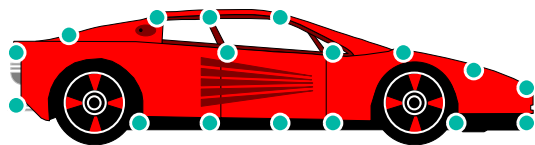
$$[RTO_n] = [T][RTO_a]$$

ANIMATION

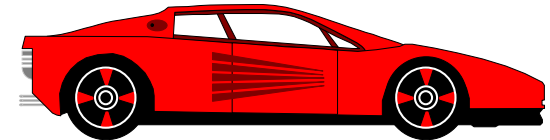
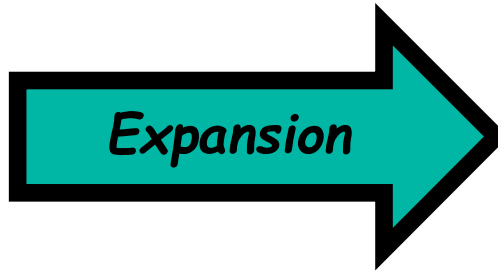


Expansion - Very Important Piece

Dynamic expansion means :



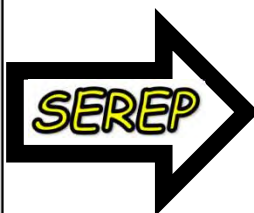
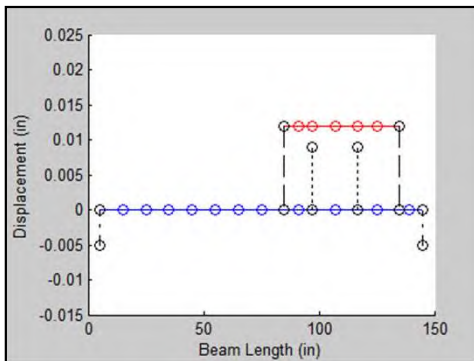
X_A = active set of dof's



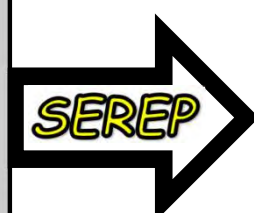
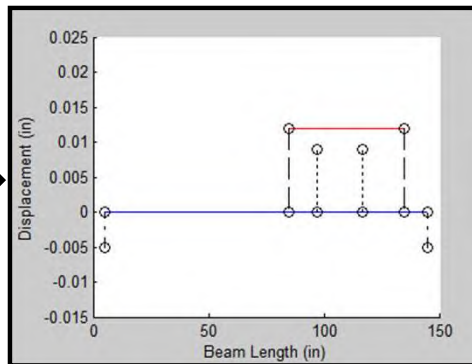
X_F = full set of dof's

Starting with a sparse set of degrees of freedom and "expanding" to a much larger set of degrees of freedom (possibly intermediate space or full field space or FEA dofs)

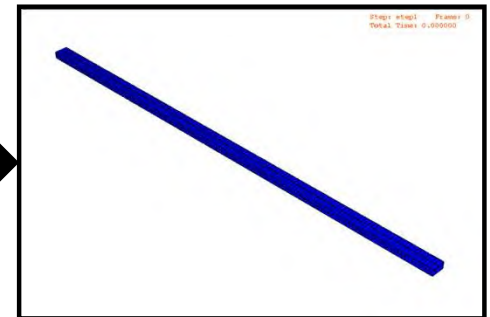
Response at a-space



Expanded Response at N-space

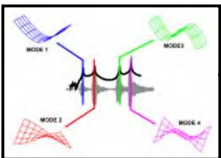


Component Expanded Response Full Field



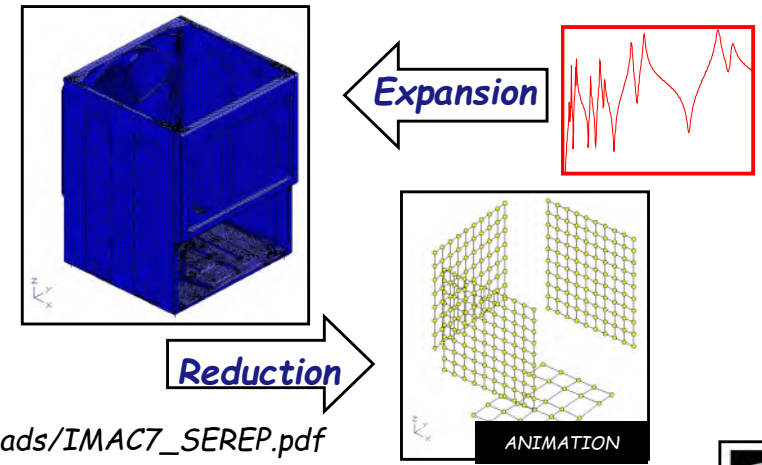
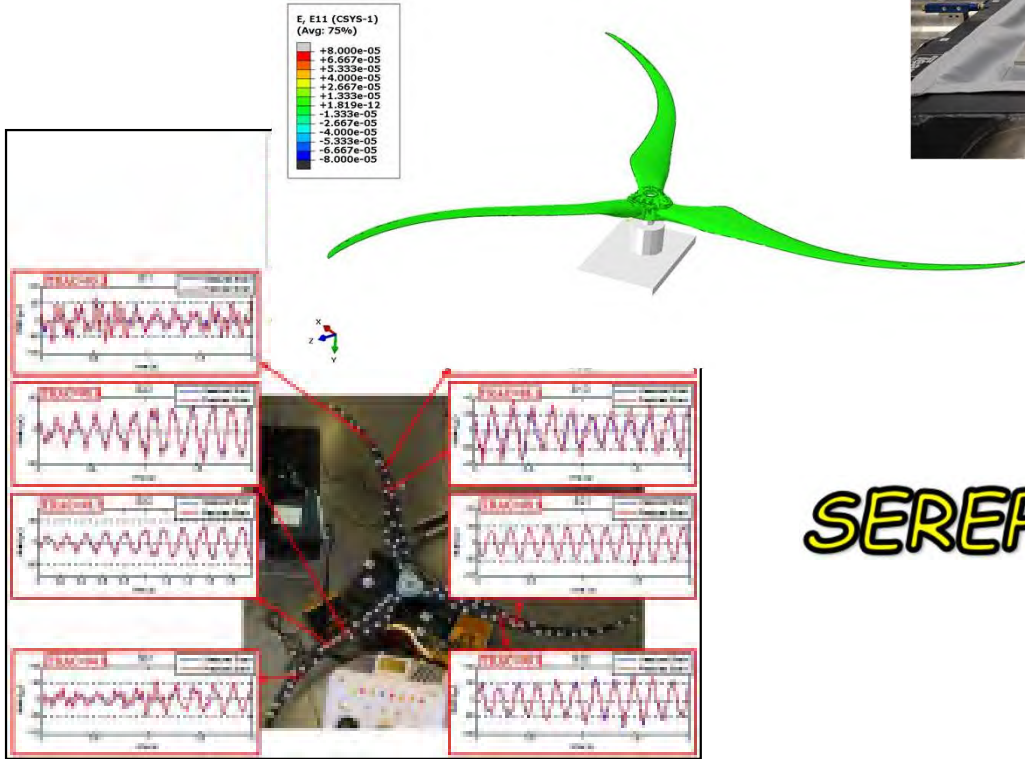
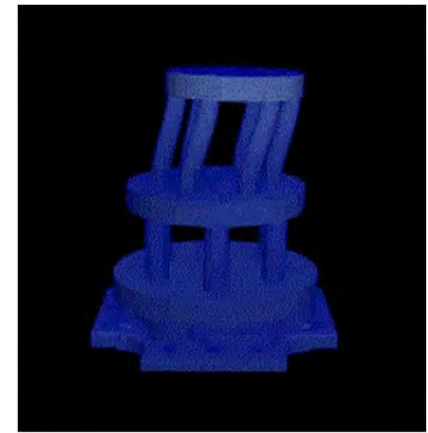
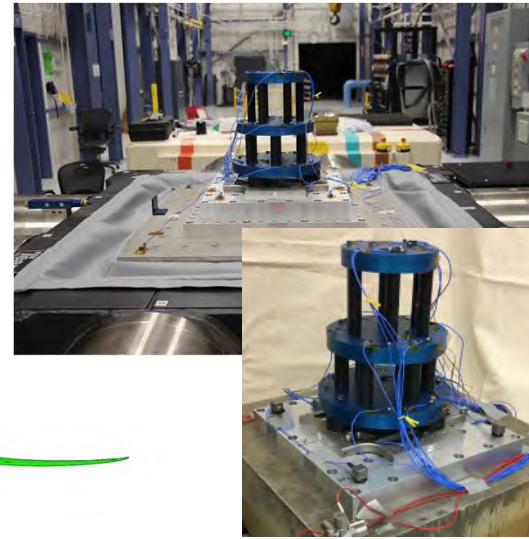
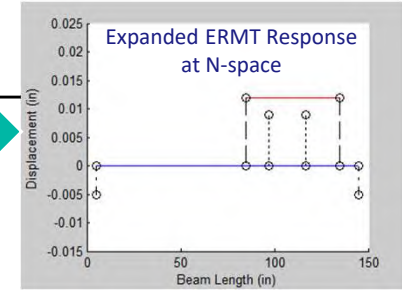
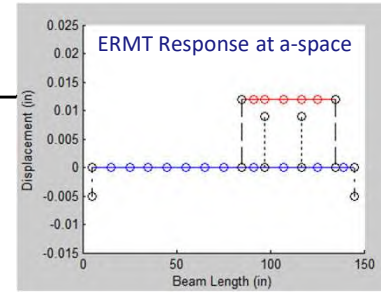
https://faculty.uml.edu//pavitabile/downloads/IMAC7_SEREP.pdf

ANIMATION

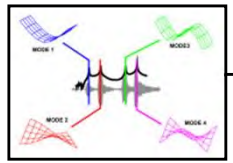


Expansion Possibilities

Expansion useful in a broad range of applications.



https://faculty.uml.edu//pavitabile/downloads/IMAC7_SEREP.pdf

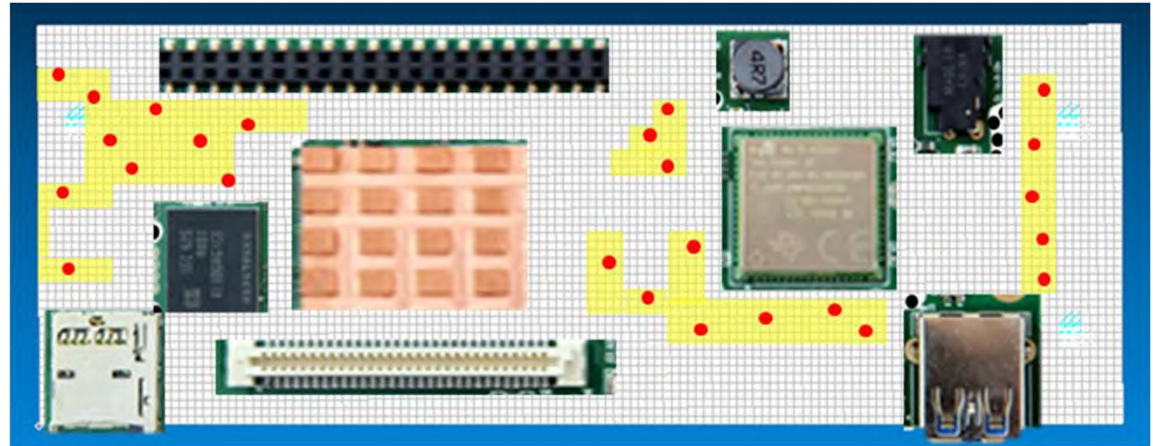


Direct Strain Expansion - Analytical Studies

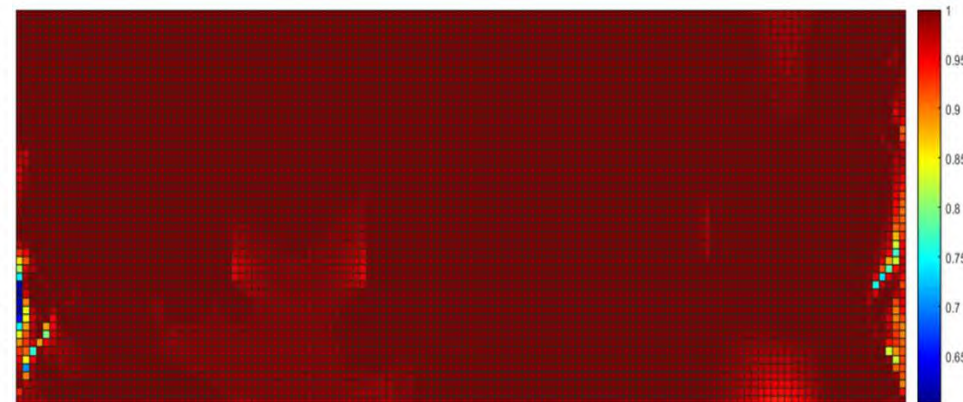
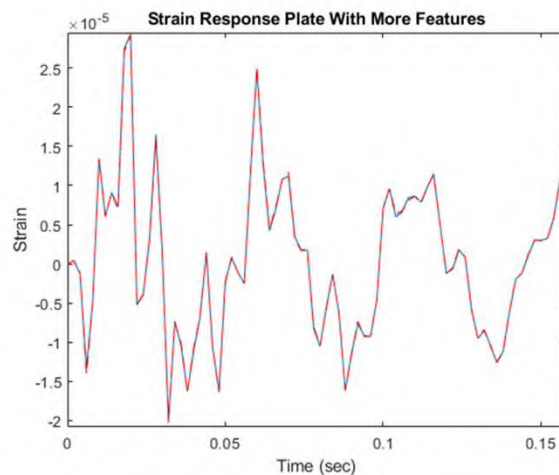
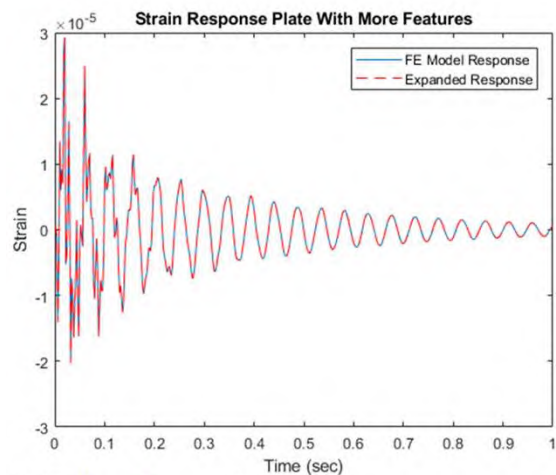
Limited Line of Sight - Strain Expansion offers enhancements:

Structure Speckled for DIC

Corresponding Area on Modified Model

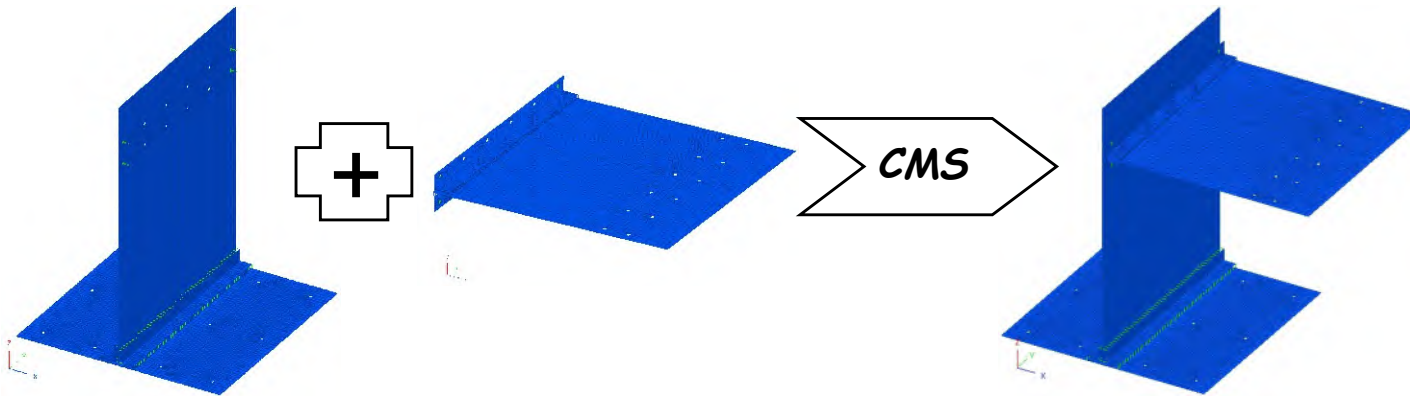
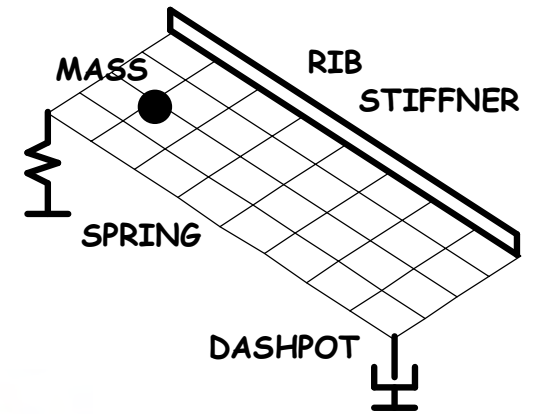


25 Active Elements - Average TRAC = 0.9956

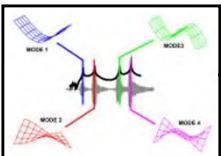


SEREP

Structural Dynamic Modification and System Modeling



https://faculty.uml.edu//pavitabile/downloads/IMAC20_SDM20_MACL.pdf



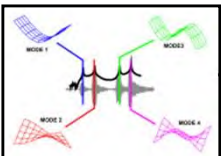
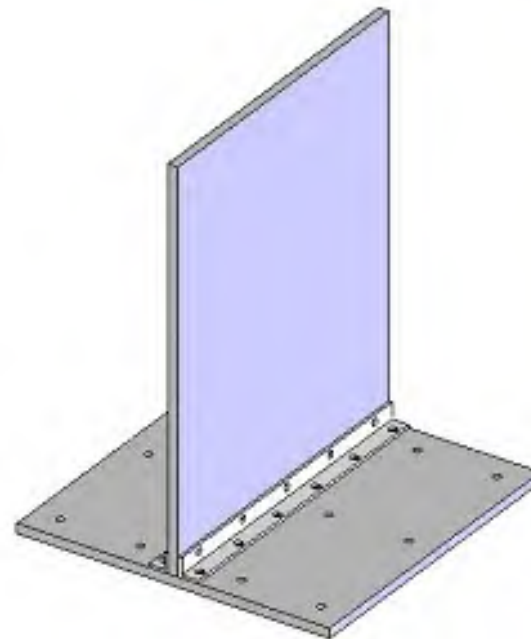
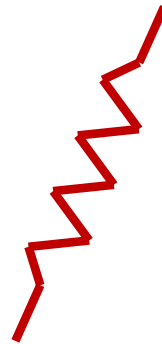
Predict Changes in Mass and Stiffness

Can the existing frequencies and mode shapes be used to predict new frequencies and mode shapes due to changes in mass and stiffness

$$\left[\begin{array}{ccc} \ddots & & \\ & \bar{M}_1 & \\ & & \ddots \end{array} \right] + [\Delta \bar{M}_{12}] \{ \ddot{p}_1 \} + \left[\begin{array}{ccc} \ddots & & \\ & \bar{K}_1 & \\ & & \ddots \end{array} \right] + [\Delta \bar{K}_{12}] \{ p_1 \} = [0]$$

$$[\Delta \bar{M}_{12}] = [U_1]^T [\Delta M_{12}] [U_1]$$

$$[\Delta \bar{K}_{12}] = [U_1]^T [\Delta K_{12}] [U_1]$$



Eigenvalue Modification Technique - BORN

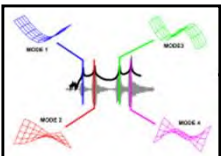
The modal projection is used to recast the equations as

$$\left[\begin{array}{ccc} \ddots & & \\ & \bar{\mathbf{M}}_1 & \\ & & \ddots \end{array} \right] + [\Delta \bar{\mathbf{M}}_{12}] \{ \ddot{\mathbf{p}}_1 \} + \left[\begin{array}{ccc} \ddots & & \\ & \bar{\mathbf{K}}_1 & \\ & & \ddots \end{array} \right] + [\Delta \bar{\mathbf{K}}_{12}] \{ \mathbf{p}_1 \} = \{ \mathbf{0} \}$$

where $[\Delta \bar{\mathbf{M}}_{12}] = [\mathbf{U}_1]^T [\Delta \mathbf{M}_{12}] [\mathbf{U}_1]$ $[\Delta \bar{\mathbf{K}}_{12}] = [\mathbf{U}_1]^T [\Delta \mathbf{K}_{12}] [\mathbf{U}_1]$

An eigensolution of an $(m \times m)$ system is required to uncouple the set of equations

$$\left[\left[\begin{array}{ccc} \ddots & & \\ & \bar{\mathbf{K}}_1 & \\ & & \ddots \end{array} \right] + [\mathbf{U}_1]^T [\Delta \mathbf{K}_{12}] [\mathbf{U}_1] - \lambda \left[\begin{array}{ccc} \ddots & & \\ & \bar{\mathbf{M}}_1 & \\ & & \ddots \end{array} \right] + [\mathbf{U}_1]^T [\Delta \mathbf{M}_{12}] [\mathbf{U}_1] \right] \{ \mathbf{p}_1 \} = \{ \mathbf{0} \}$$



Local Eigenvalue Modification Technique - FAST

If only one change of mass or stiffness is considered then these equations can be reduced to

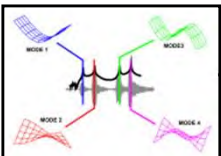
$$\left[\begin{array}{c} \vdots \\ \vdots \\ \bar{\mathbf{K}}_1 \\ \vdots \\ \vdots \end{array} \right] + \{v_{1k}\} \alpha_{1k} \{v_{1k}\}^T - \lambda \left[\begin{array}{c} \vdots \\ \vdots \\ \bar{\mathbf{M}}_1 \\ \vdots \\ \vdots \end{array} \right] \{p_1\} = \{0\}$$

$$\left[\begin{array}{c} \vdots \\ \vdots \\ \bar{\mathbf{K}}_1 \\ \vdots \\ \vdots \end{array} \right] - \lambda \left[\begin{array}{c} \vdots \\ \vdots \\ \bar{\mathbf{M}}_1 \\ \vdots \\ \vdots \end{array} \right] + \{v_{1m}\} \alpha_{1m} \{v_{1m}\}^T \{p_1\} = \{0\}$$

The solution then reduces to a second order equation for each of the 'm' modes of the system

$$\frac{1}{\alpha_k} = \sum_{i=1}^m \frac{\left(\{u^{(i)}\}^T \{t_k\} \right)^2}{\Omega_2^2 - \omega_i^2}$$

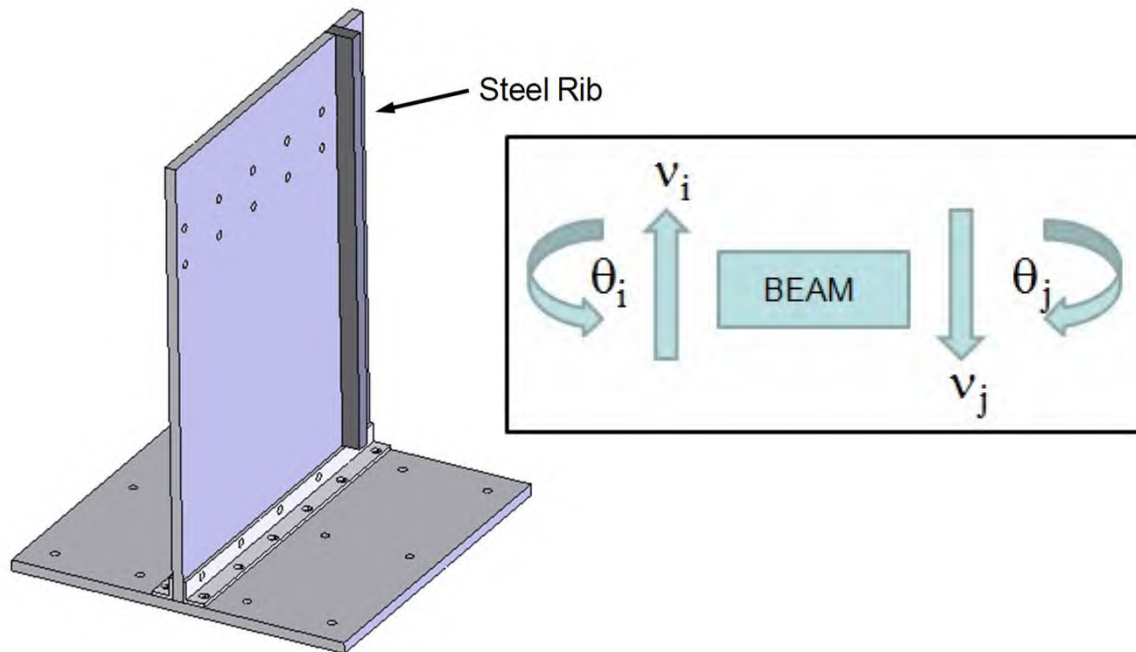
$$\frac{1}{\Omega^2 \alpha_m} = \sum_{i=1}^m \frac{\left(\{u^{(i)}\}^T \{t_m\} \right)^2}{\Omega_2^2 - \omega_i^2}$$



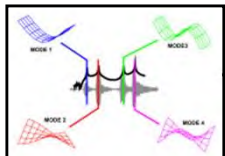
Realistic Beam Modifications Necessary

Realistic modifications require more than just translational degrees of freedom for modifications.

Rotary degrees of freedom necessary.



Expansion is the Tool



Expansion Required

Spline fits of data proved inadequate.

Equivalent Reduction (ER) developed for this.

$$\{X_n\} = \begin{Bmatrix} X_a \\ X_d \end{Bmatrix} = [T] \{X_a\} \quad [T_U] = [U_n][U_a]^g$$

Note:

ER helped SDM for rotary modifications and allowed the path for system models to become realistic.

Expansion Required

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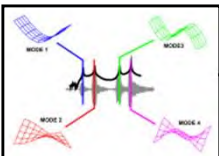
$$\{X_n\} = \begin{Bmatrix} X_a \\ X_d \end{Bmatrix} = [T] \{X_a\} \quad [T_U] = [U_n][U_a]^g$$

Note:

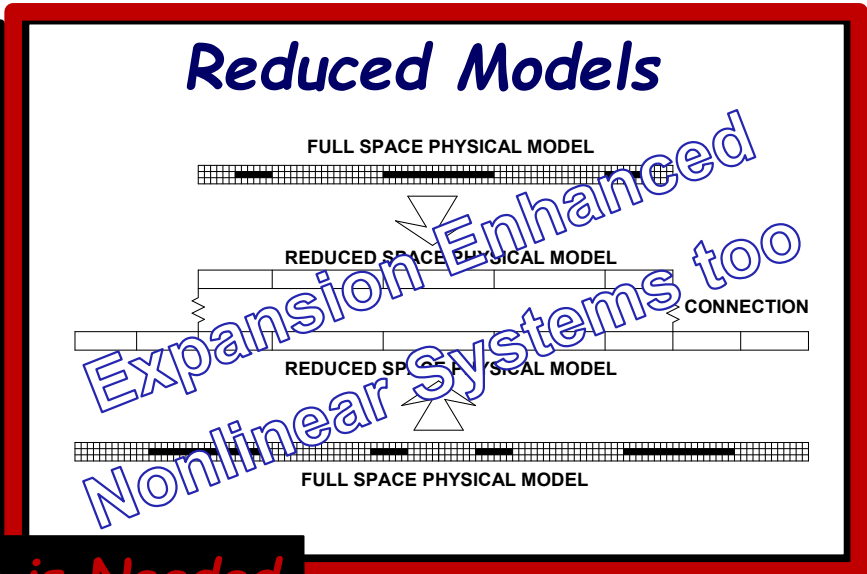
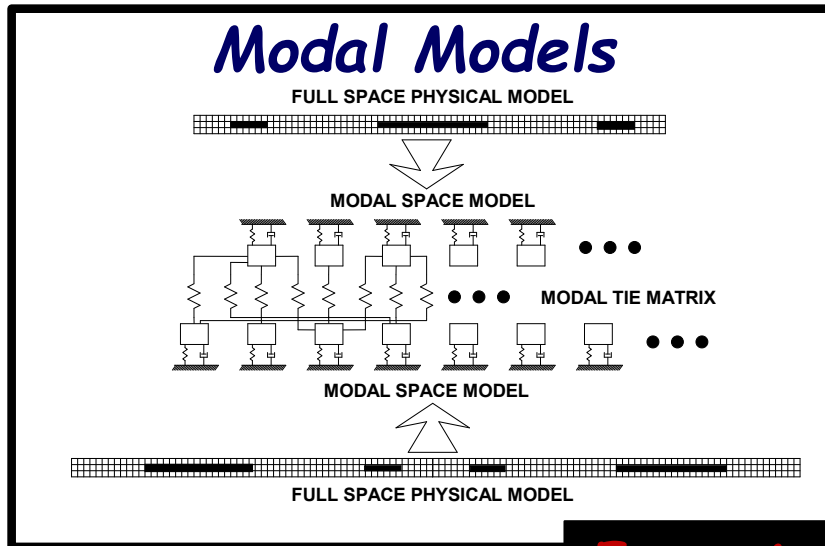
ER helped SDM for rotary modifications and allowed the path for system models to become realistic.

BUT ... not too much later ER grew up and became SEREP.

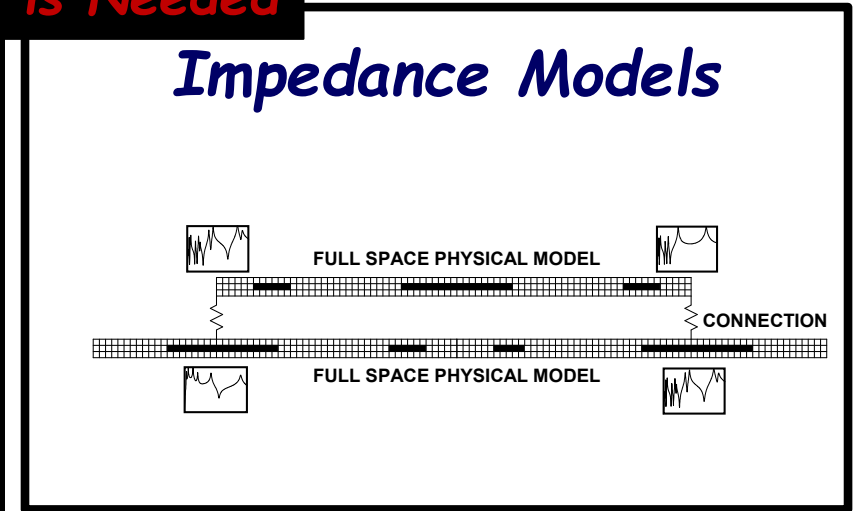
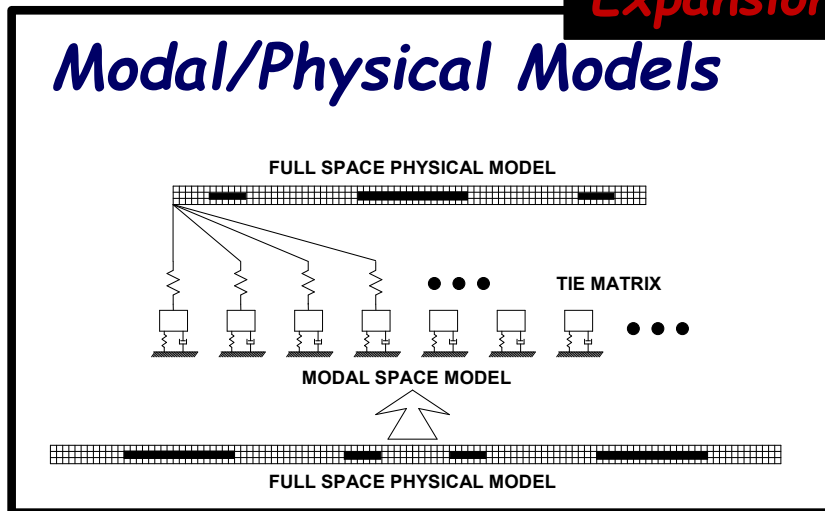
SEREP is an extremely powerful model reduction tool.



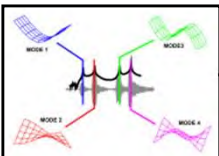
System Modeling Possibility



Expansion is Needed



https://faculty.uml.edu//pavitabile/downloads/Avitabile_Svdig_Vol3_No4_Jul2001_MACL.pdf



LEMP Quickly Outdated

LEMP was an extremely good efficient, computational tool.

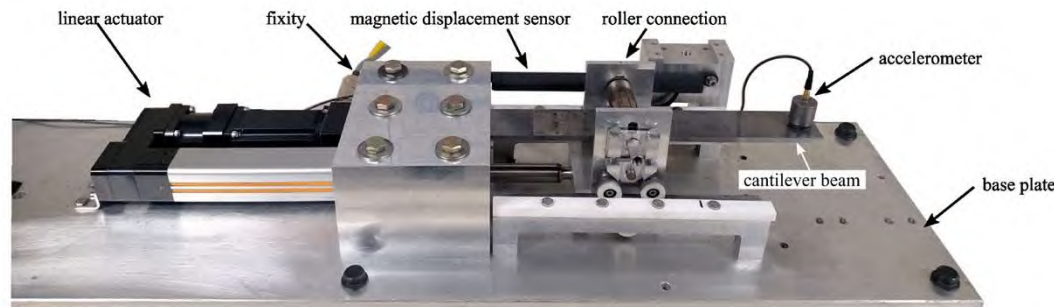
But computers got much more powerful very quickly.

LEMP was quickly retired and sent to the graveyard (early 80s).

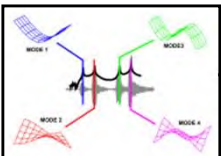


LEMP - brought back from the dead - 2020

LEMP was re-introduced to help speed up eigensolutions.



*“Real-time Structural Model Updating using Local Eigenvalue Modification Procedure for Applications in High-Rate Dynamic Events”, E.Ogunniyi, C.Drnek, S.Hong, Y.Wang, J.D.Bakos, **P.Avitabile**, J.Dodson, Mechanical Systems and Signal Processing, Volume 195, 15 July 2023, 110318, <https://doi.org/10.1016/j.ymsp.2023.110318>*



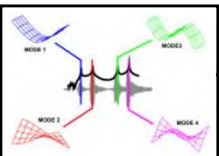
Correlation and Model Updating

*Finite Element Model Correlation, SEM Handbook of Experimental Structural Dynamics,
DOI:10.1007/978-1-4614-4547-0_34, July 2022*

*Integrating Test and FEA for Enhanced
Structural Dynamic Modeling Applications*

45

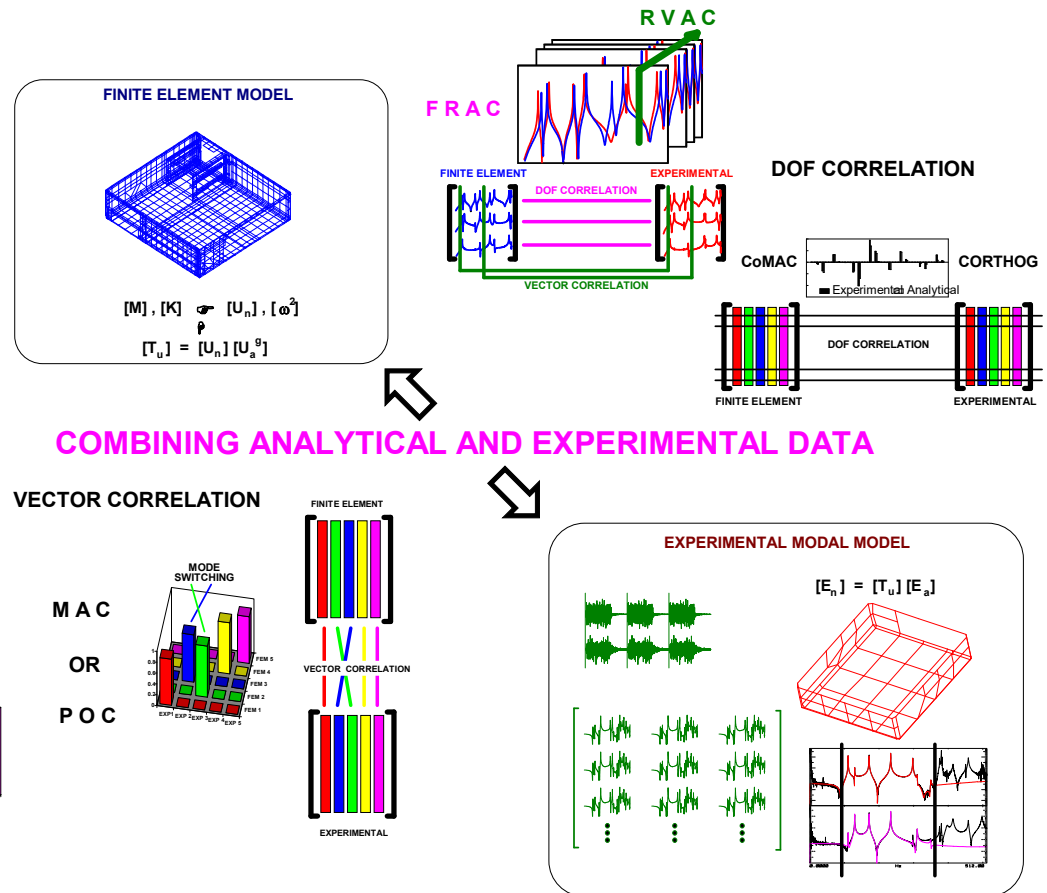
*Peter Avitabile - Mechanical Engineering
Structural Dynamics And Acoustic Systems Lab*



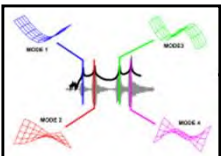
Correlation and Updating Models

Research, development and application of these techniques has been a continuing area of focus for many applications

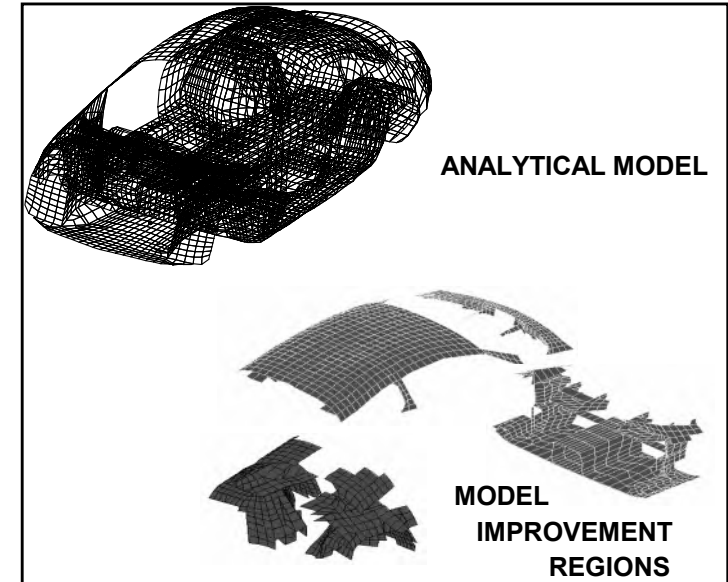
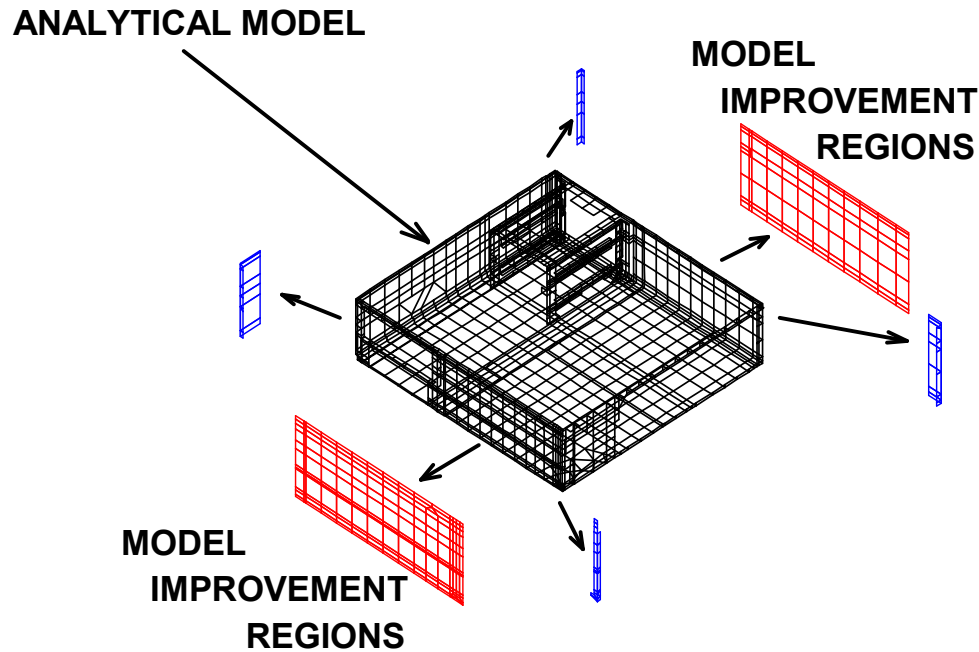
Analytical and experimental models are correlated and adjusted to provide better component and system models



Reduction and Expansion Needed



Updating Models



- *Models can be adjusted to better reflect actual measured system characteristics*
- *Joint stiffness can be more accurately identified*
- *Simplistic modeling assumptions can be modified to reflect the actual system*

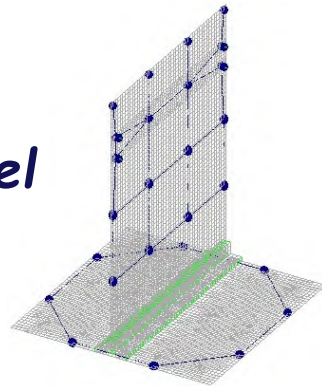
Analytical Model Improvement (AMI)

Seed Mass and Stiffness and **Target Data**

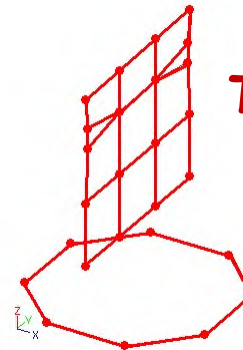
$$[M_I] = [M_S] + [V]^T [I - \bar{M}_S] [V]$$

$$[K_I] = [K_S] + [V]^T [\omega^2 - \bar{K}_S] [V] - [[K_S][E]][V] - [[K_S][E]][V]]^T$$

Seed finite element model

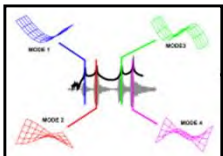


Target test modes



Mass Weighted Generalized Inverse

$$[V] = [[E]^T [M_S] [E]]^{-1} [E]^T [M_S] = [\bar{M}_S]^{-1} [E]^T [M_S]$$



Tools For Evaluation

Correlation Metrics

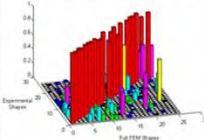
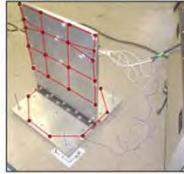
Modal Assurance Criteria - MAC

$$MAC_{ij} = \frac{[\{u_i\}^T \{e_j\}]^2}{[\{u_i\}^T \{u_i\}][\{e_j\}^T \{e_j\}]}$$

$$MAC_{ij} = \frac{[\{e_i\}^H \{e_j\}]^2}{[\{e_i\}^H \{e_i\}][\{e_j\}^H \{e_j\}]}$$

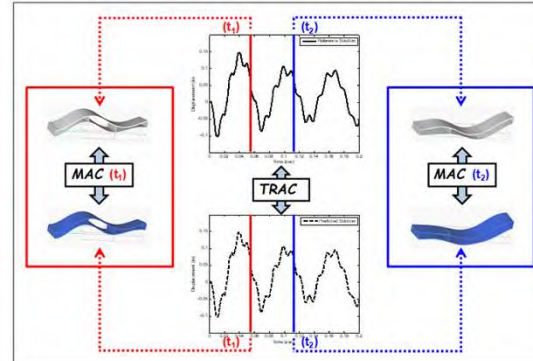
Table 3: Correlation of Upright FEA Model and Test Data along with MAC Matrix.

FEA Mode Number	Experimental Mode Number	FEA Frequencies (Hz)	Experimental Frequencies (Hz)	Frequency % Difference	MAC Value (%)	20 Mode POC Value
1	1	26.03	26.17	-0.54	99.8	1.071
2	2	70.70	67.34	4.75	98.5	0.973
3	3	77.69	75.89	2.32	99.3	1.033
4	4	108.80	104.48	3.97	97.1	1.073
5	5	158.01	161.55	-2.24	98.9	1.048
6	6	269.88	270.00	-0.04	98.9	1.019
7	7	304.40	299.93	1.47	96.5	1.024
8	8	350.45	354.60	-1.18	98.5	0.979
9	9	419.36	419.22	0.03	94.3	0.483
10	10	457.13	482.15	-5.47	92.7	0.836
11	12	541.29	547.84	-1.21	96.6	0.995
12	11	563.47	538.20	4.48	86.3	0.785
13	13	777.29	771.76	0.71	90.3	1.002
14	14	779.61	791.18	-1.48	97.7	0.944
15	15	862.76	871.90	-1.06	84.3	0.795



Test Response Assurance Criteria - TRAC

$$TRAC = \frac{[\{x_{n1}\}^T \{x_{n2}\}]^2}{[\{x_{n1}\}^T \{x_{n1}\}][\{x_{n2}\}^T \{x_{n2}\}]}$$



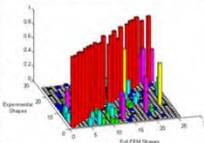
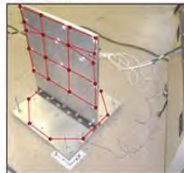
Pseudo Orthogonality Check - POC

$$POC = [U_a]^T [M_a][E_a]$$

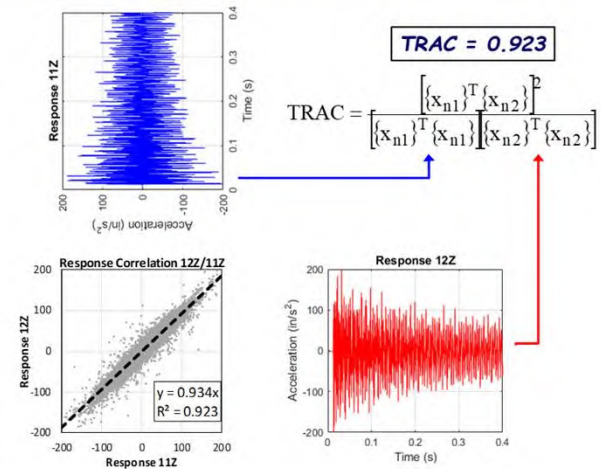
$$POC = [U_n]^T [M_n][E_n]$$

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Test Response Assurance Criteria - TRAC

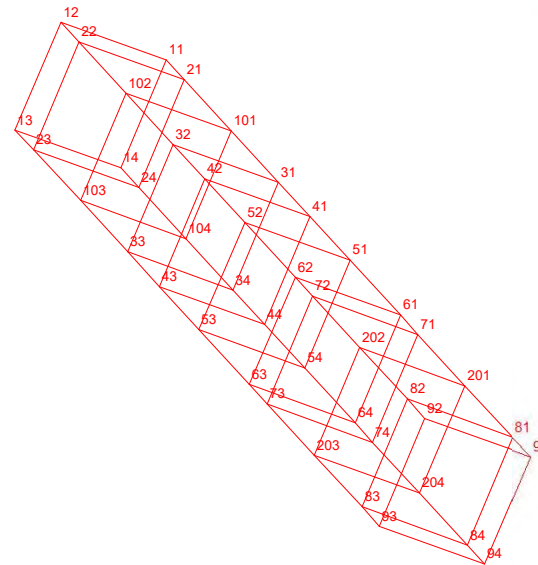
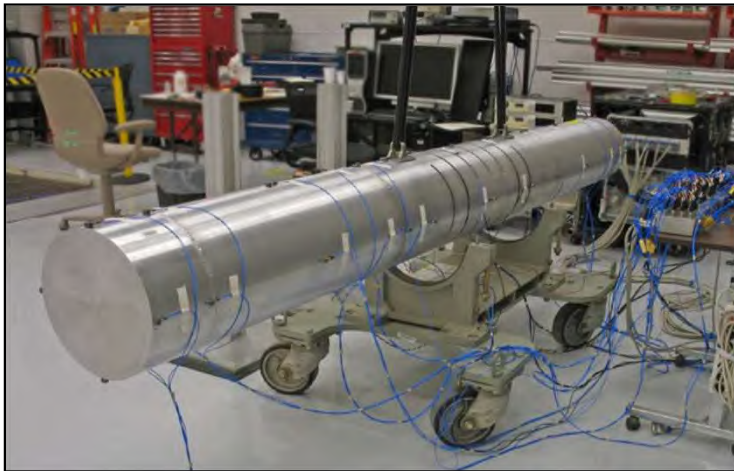




Structural Dynamics and Acoustic Systems Laboratory University of Massachusetts Lowell

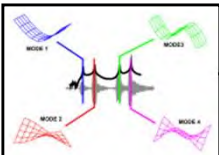
Updated Reduced Order Model Development for Forced Response Predictions

Sandia Monolithic Dynamic Mockup (MDM)

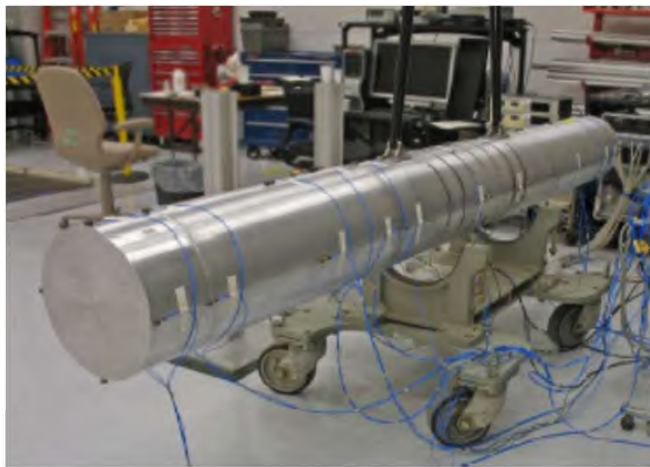


Structural Dynamics and Acoustic Systems Laboratory University of Massachusetts Lowell

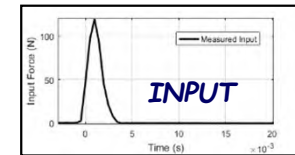
http://faculty.uml.edu/pavitabile/downloads/IMAC36_MDM_Response_101317_DRAFT.pdf



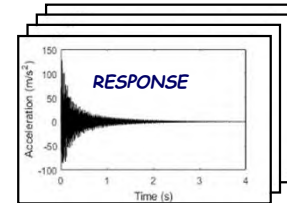
The MDM Model - Overall Process



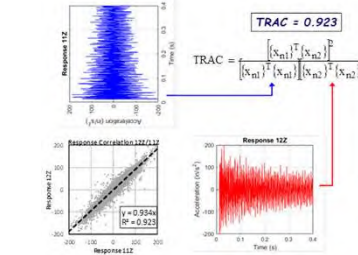
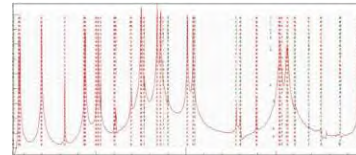
TEST MODEL



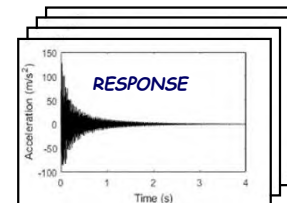
MEASURED RESPONSE



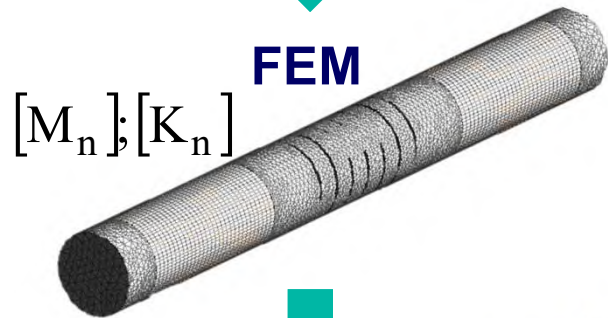
$$[\Omega^2]; [E_a]$$



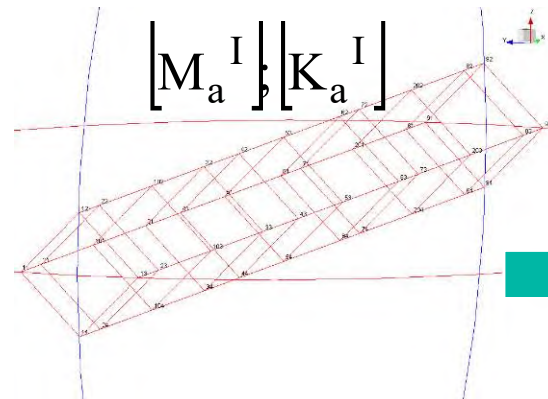
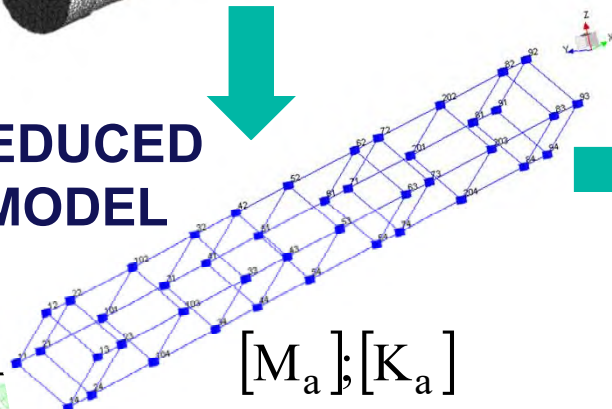
TRAC



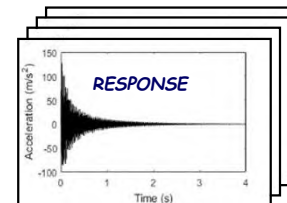
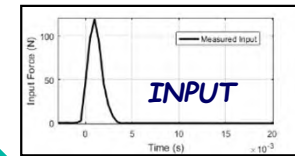
PREDICTED RESPONSE



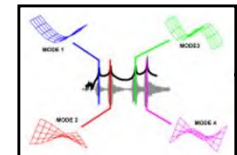
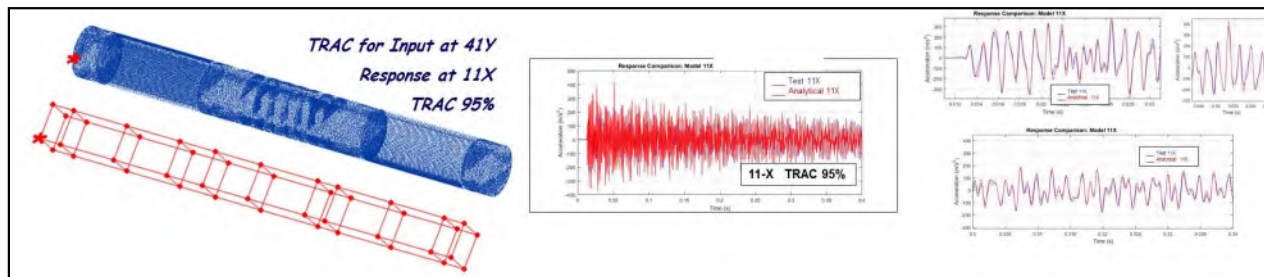
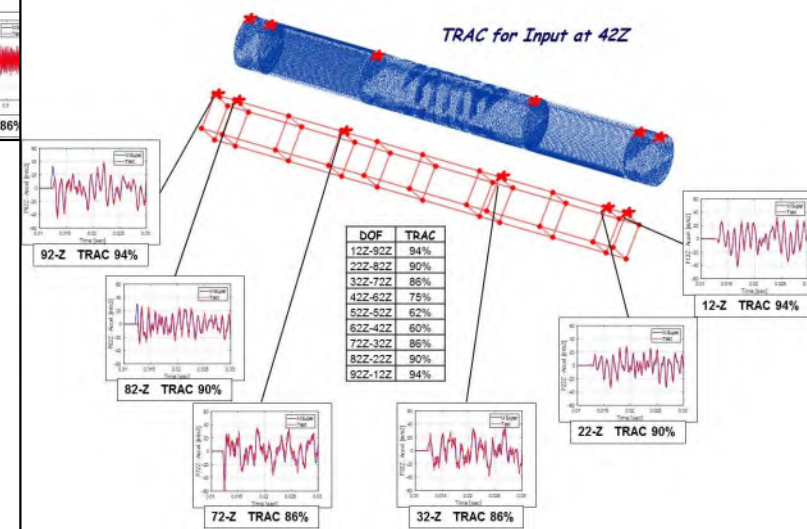
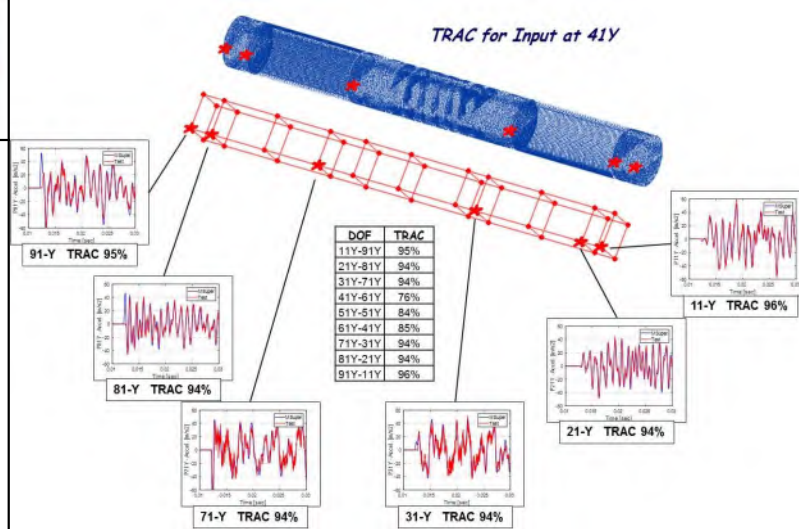
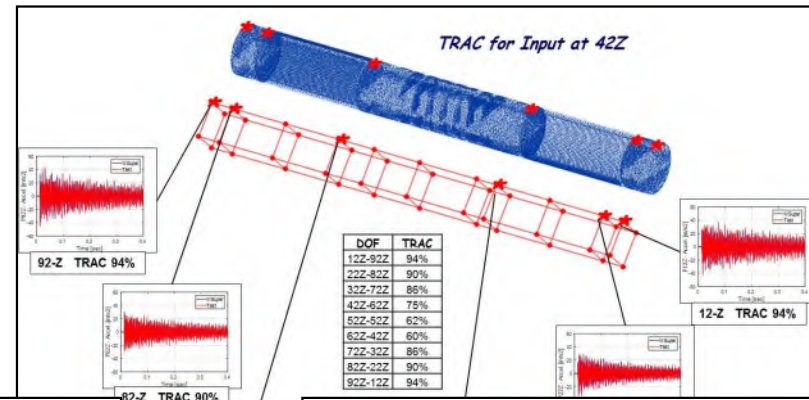
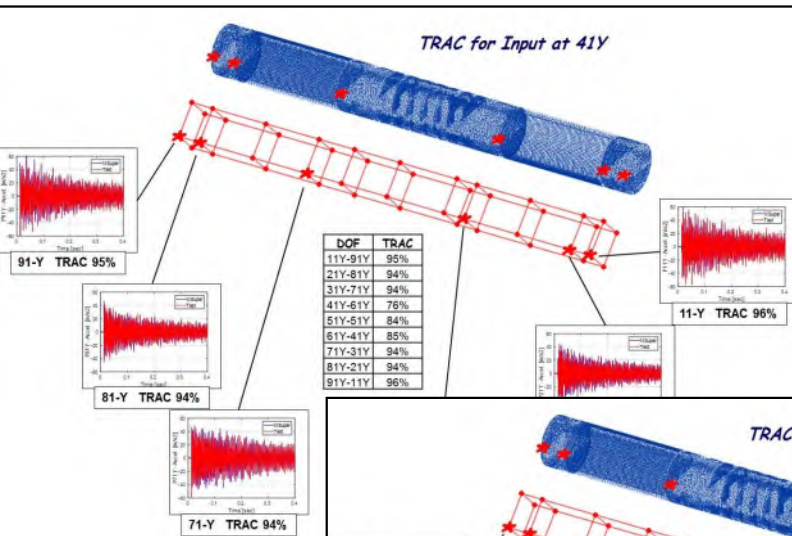
REDUCED MODEL



IMPROVED UPDATED REDUCED MODEL



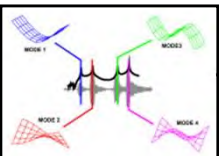
MDM Updated Reduced Model Forced Response Predictions



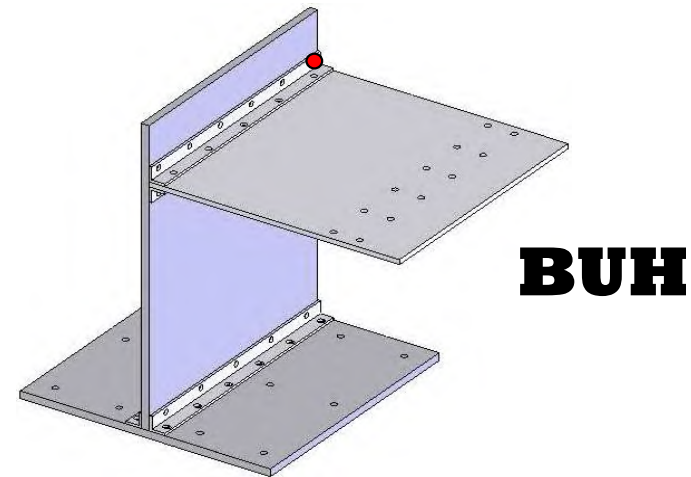
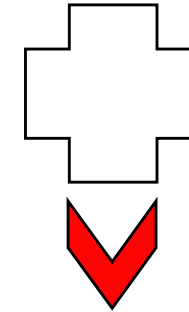
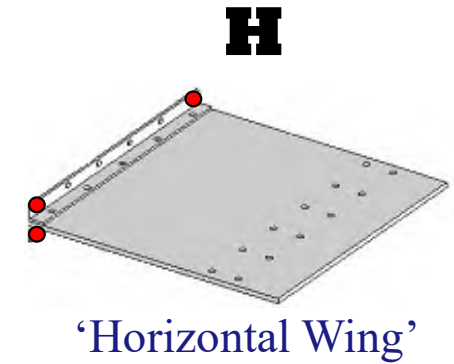
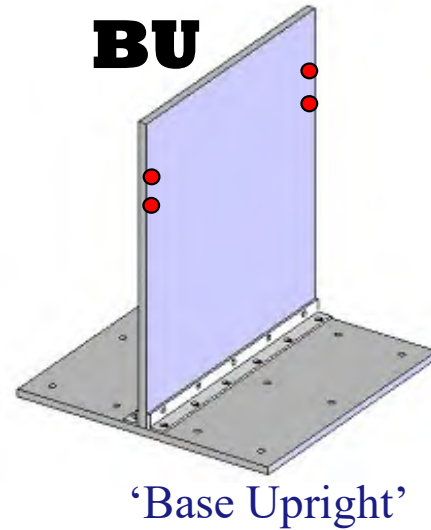
Reduced Order Models for Efficient Response

Redstone Apache Missile Firing Systems

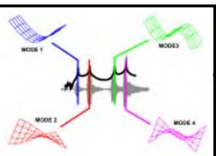
Van Zandt, Wirkkala, Butland, Nicgorski, Chipman, Avitabile



Test Analysis Correlation - Component/System



“Base Upright Horizontal” assembly



New Reduced-Order Test-Verified Approaches

Approach 1 - Complete System Model Reduction with Reduced Model Updating:

(Modal Based System Response - MBSR)

Approach 2 - Reduced Component Model Updating from Assembled System Measured Characteristics:

(Modal Based Component Response - MBCR)

Approach 3 - Reduced Component Model Updating from Impedance Developed Model for System Characteristics:

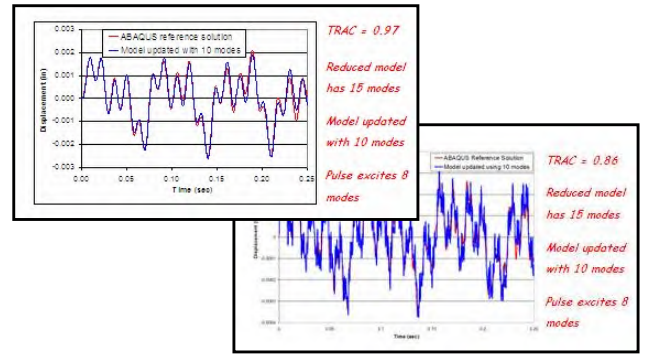
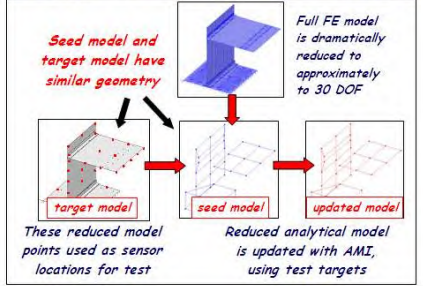
(Frequency Based Component Response - FBCR)

Approach 4 - Reduced Component Model Updating for Craig Bampton Constraint Model:

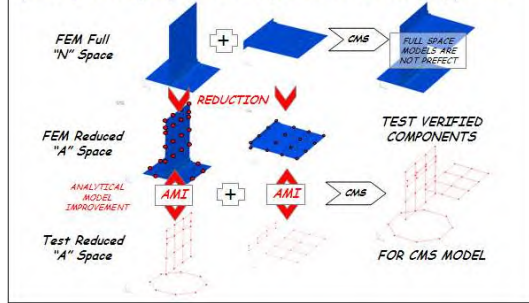
(Craig Bampton Component Response - CBCR)

New Reduced-Order Test-Verified Approaches

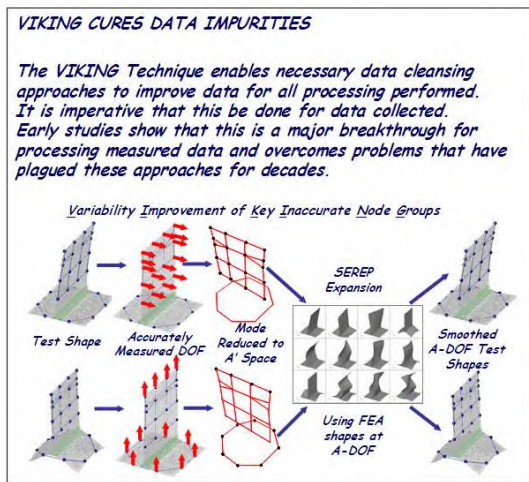
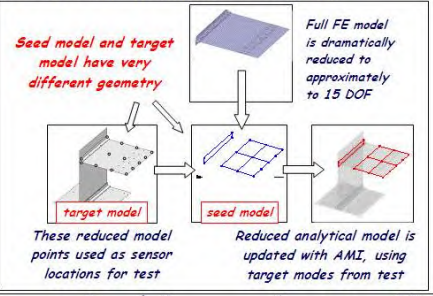
Modal Based System Response - MBSR
 Complete System Model Reduction with Direct Updating - Modeling technique usually employed in many instances and provides very good results. Requires full model representation



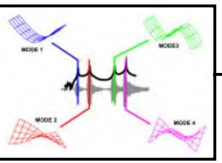
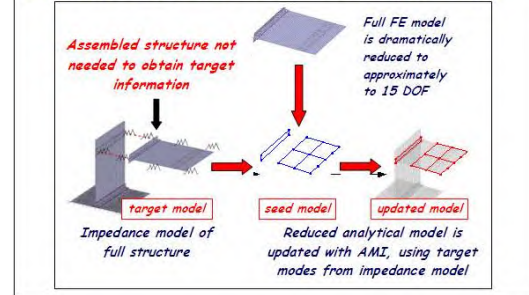
Craig Bampton Component Response - CBCR
 Test Verified, Reduced Order Component Models used to form Craig Bampton Component Mode Synthesis Models - Individual components are reduced and then updated using a direct updating technique prior to being used for CB-CMS approach



Modal Based Component Response - MBCR
 Reduced Component Model Updating from Assembled System Response - Advantage of technique is that only part of the structure needs to be modeled (such as component) but direct model updating is performed using test data from complete system assembly. This technique requires that the system targets be obtained from an assembled system



Frequency Based Component Response - FBCR
 Reduced Component Model Updating from Impedance Developed Model for System Characteristics - Similar to MBCR but the system targets are obtained from component information used to form a frequency based substructure. This implies that the components do not need to be assembled for system targets

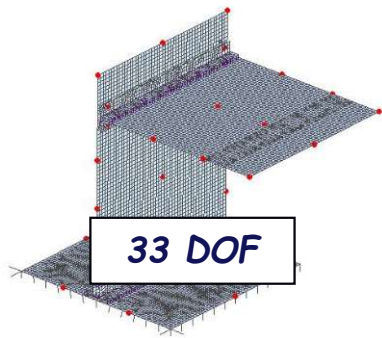




Efficient Reduced Models for Multibody Dynamic Response Analysis

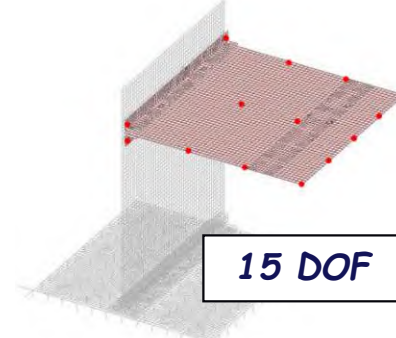


Dynamic Time Response Comparison to Full Model



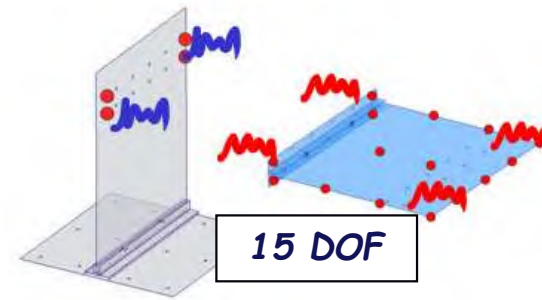
33 DOF

Reduced System Model



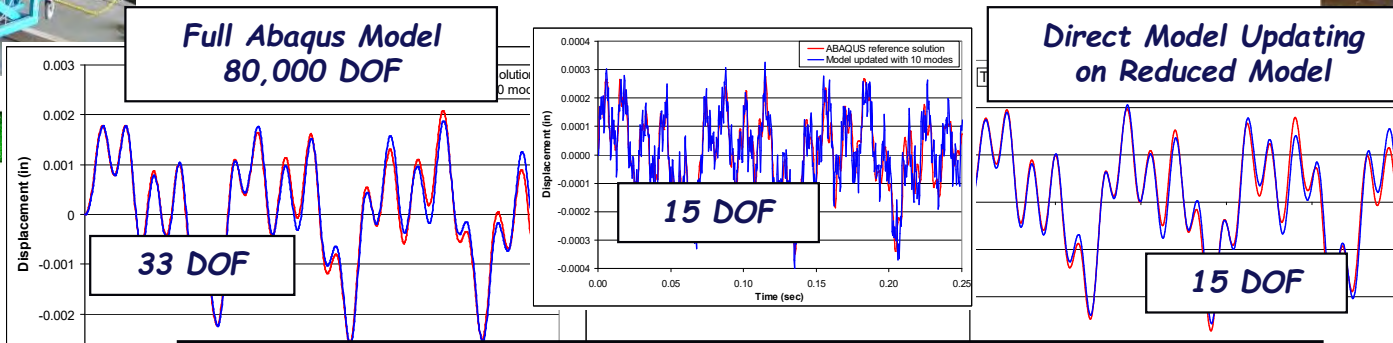
15 DOF

Reduced Component Model



15 DOF

Reduced Impedance Model



Full Abaqus Model
80,000 DOF

Direct Model Updating
on Reduced Model

33 DOF

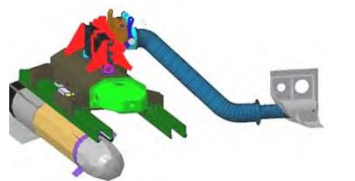
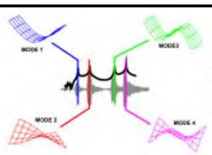
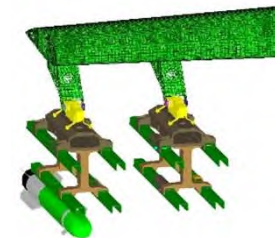
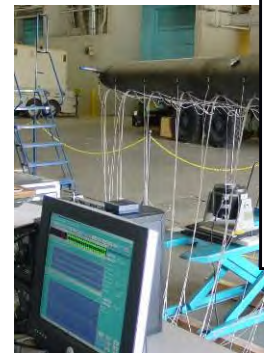
15 DOF

15 DOF

Direct Integration of the Models for Response

Full Abaqus (24 hr) vs 33 DOF (2 min)

Full Abaqus (24 hr) vs 15 DOF (1.5 min)

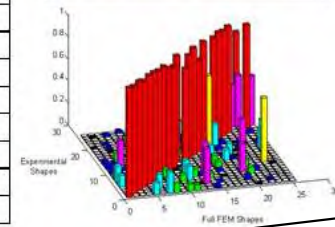
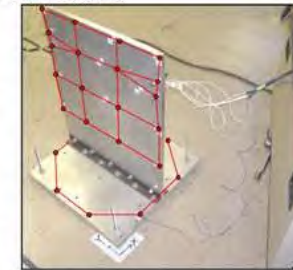


Test Analysis Correlation - Component/System



Table 3: Correlation of Upright FEA Model and Test Data along with MAC Matrix.

FEA Mode Number	Experimental Mode Number	FEA Frequencies (Hz)	Experimental Frequencies (Hz)	Frequency % Difference	MAC Value (%)	20 Mode POC Value
1	1	26.03	26.17	-0.54	99.8	1.071
2	2	70.70	67.34	4.75	98.5	0.973
3	3	77.69	75.89	2.32	99.3	1.033
4	4	108.80	104.48	3.97	97.1	1.073
5	5	158.01	161.55	-2.24	98.9	1.048
6	6	269.88	270.00	-0.04	98.9	1.019
7	7	304.40	299.93	1.47	96.5	1.024
8	8	350.45	354.60	-1.18	98.5	0.979
9	9	419.36	419.22	0.03	94.3	0.483
10	10	457.13	482.15	-5.47	92.7	0.836
11	12	541.29	547.84	-1.21	96.6	0.995
12	11	563.47	538.20	4.48	86.3	0.785
13	13	777.29	771.76	0.71	90.3	1.002
14	14	779.61	791.18	-1.48	97.7	0.944
15	15	862.76	871.90	-1.06	84.3	0.795



EVEN GOOD DATA NEEDS TO BE CLEANSSED !!!

Table 2: Correlation of Wing FEA Model and Test Data along with MAC Matrix.

FEA Flexible Mode Number	Experimental Flexible Mode Number	FEA Frequency (Hz)	Experimental Frequency (Hz)	Frequency % Difference	MAC Value (%)	15 Mode POC Value
1	1	87.87	90.63	-3.14	99.6	1.013
2	2	141.94	148.60	-4.69	99.2	1.026
3	3	219.27	228.83	-4.36	99.7	1.078
4	4	243.55	253.30	-4.00	99.5	1.048
5	5	411.43	427.25	-3.85	98.5	0.992
6	6	430.80	446.72	-3.70	97.7	1.056
7	7	506.55	528.47	-4.33	96.9	0.989
8	8	549.54	568.15	-3.39	99.0	0.958
9	9	681.00	673.10	1.16	93.8	1.022
10	10	749.30	783.14	-4.52	98.8	0.993

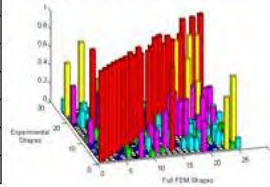
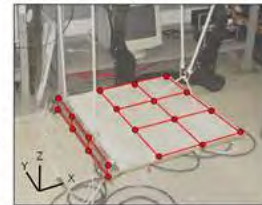
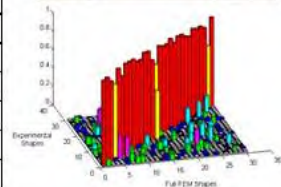
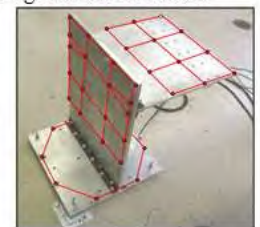


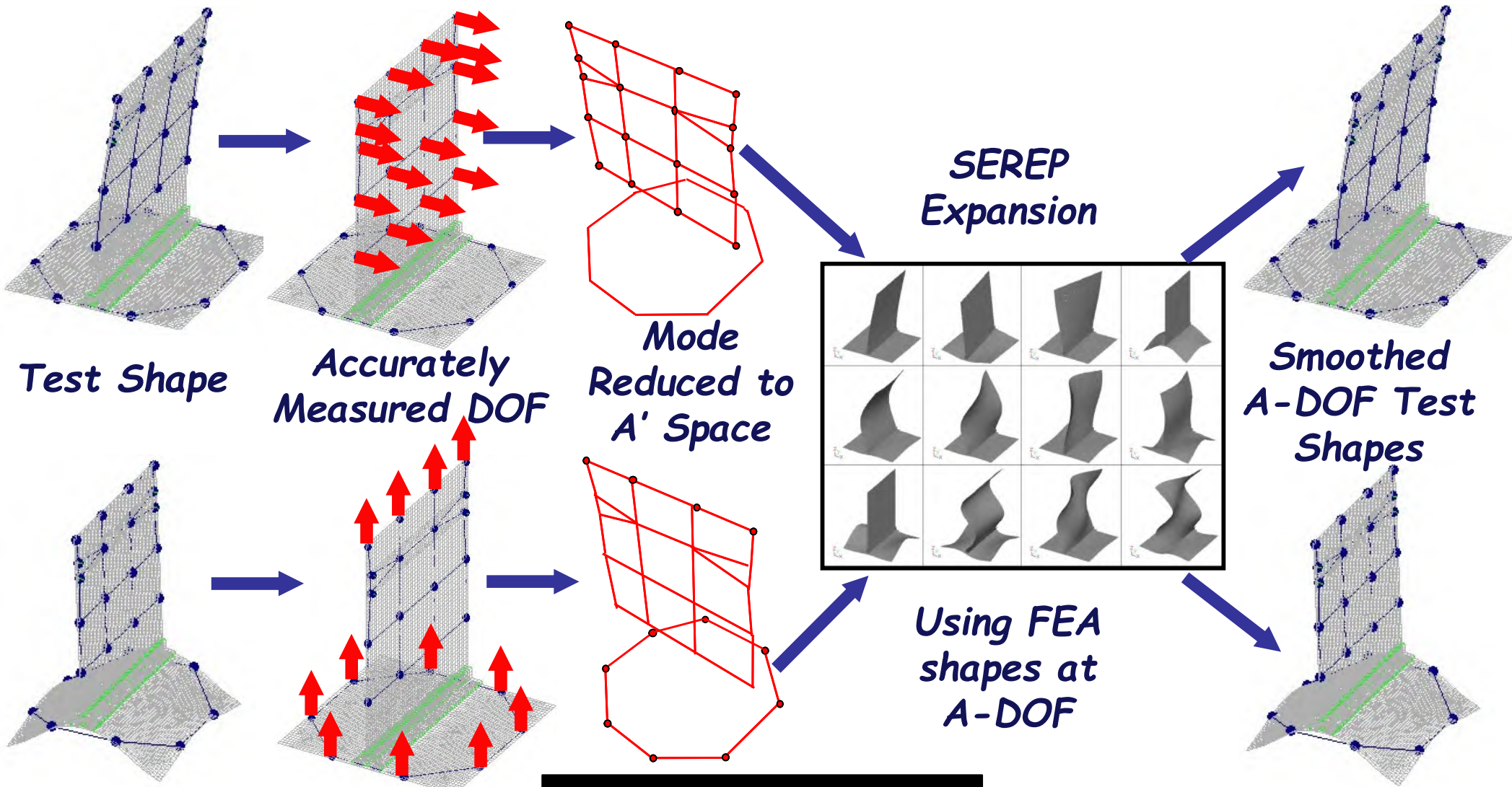
Table 4: Correlation of Assembly Test-Data and FEA Model along with MAC Matrix.

Mode Number	(Fig. A 4) FEA Freq. (Hz)	Experimental Frequencies (Hz)	Frequency % Difference	MAC Value	20 Mode POC Value
1	13.06	13.29	-1.76	98.8	0.993
2	25.2	26.03	-3.29	99.1	1.055
3	28.55	28.57	-0.07	99.1	1.075
4	53.41	52.36	1.97	98.6	1.019
5	70.25	72.28	-2.89	99.4	0.983
6	96.12	93.5	2.73	99.1	1.033
7	141.63	145.81	-2.95	99.1	1.052
8	185.54	191.72	-3.33	98.6	1.010
9	218.57	226.33	-3.55	97.3	0.998
10	231.27	237.02	-2.49	93.4	1.015

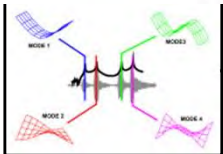


VIKING Mode Shape Smoothing Process

Variability Improvement of Key Inaccurate Node Groups

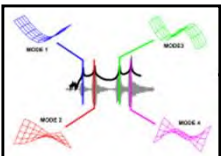
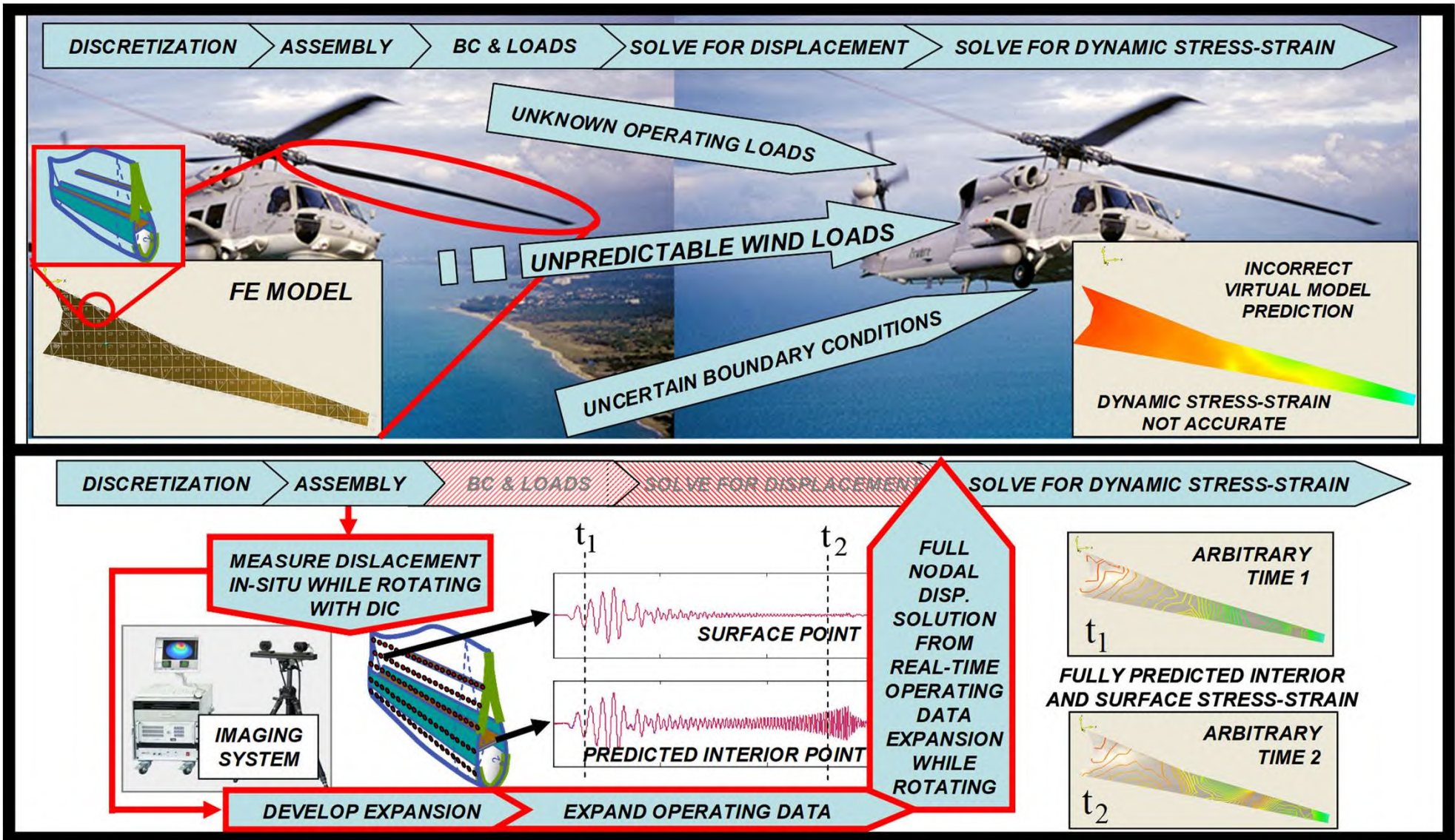


Expansion on Steroids



Full Field Strain Expansion from Sparsely Measured Data (and forces unknown)

Conventional Approach vs. Alternate Approach



Dynamic Response - Fundamental Change in Approach

Develop FEA model as usual

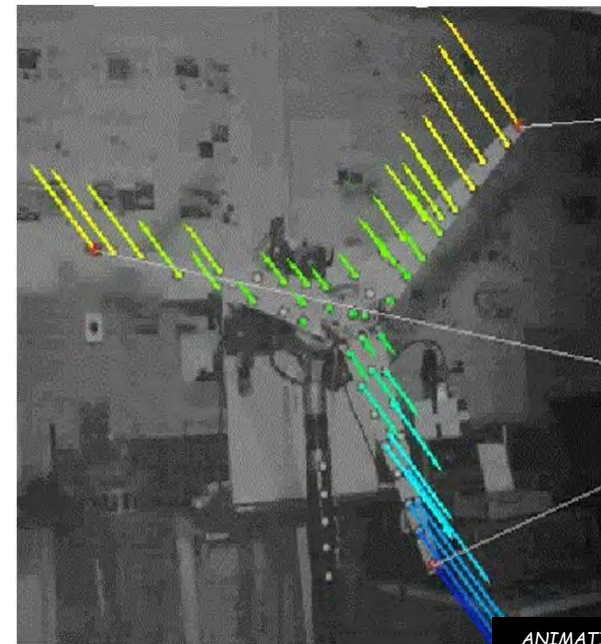
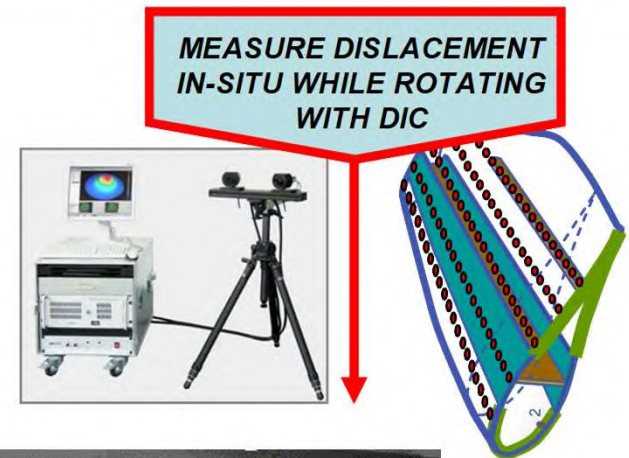
Measure response using full field approaches

- Pontos for discrete points
- Aramis for surface strain

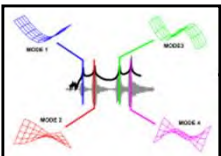
Advantage

No assumption as to load or boundary conditions

Actual displacement directly obtained



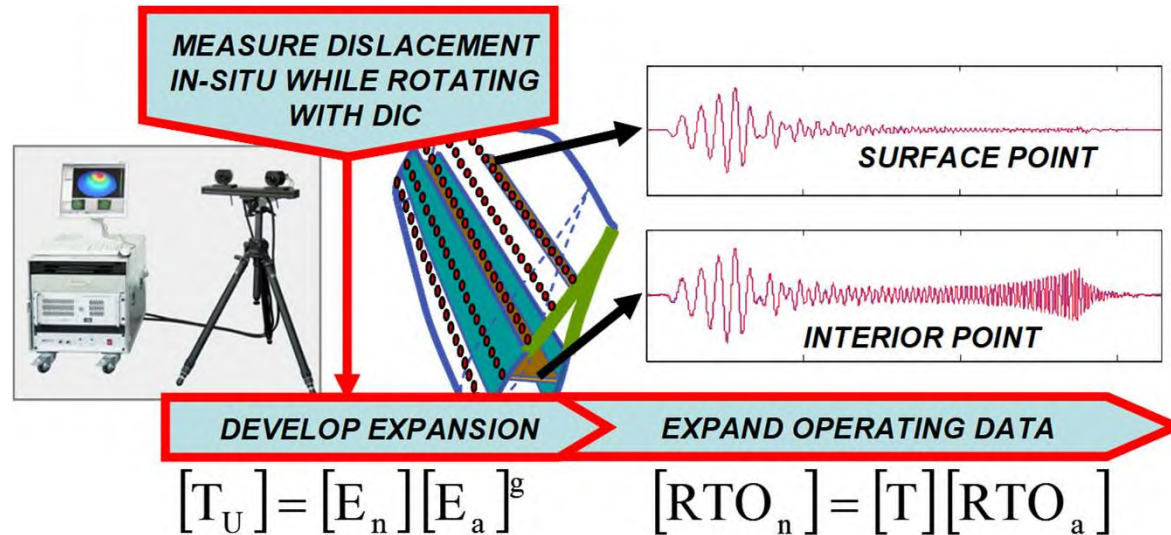
ANIMATION



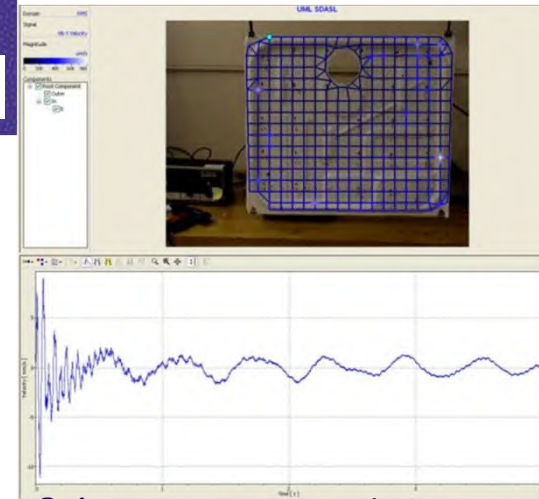
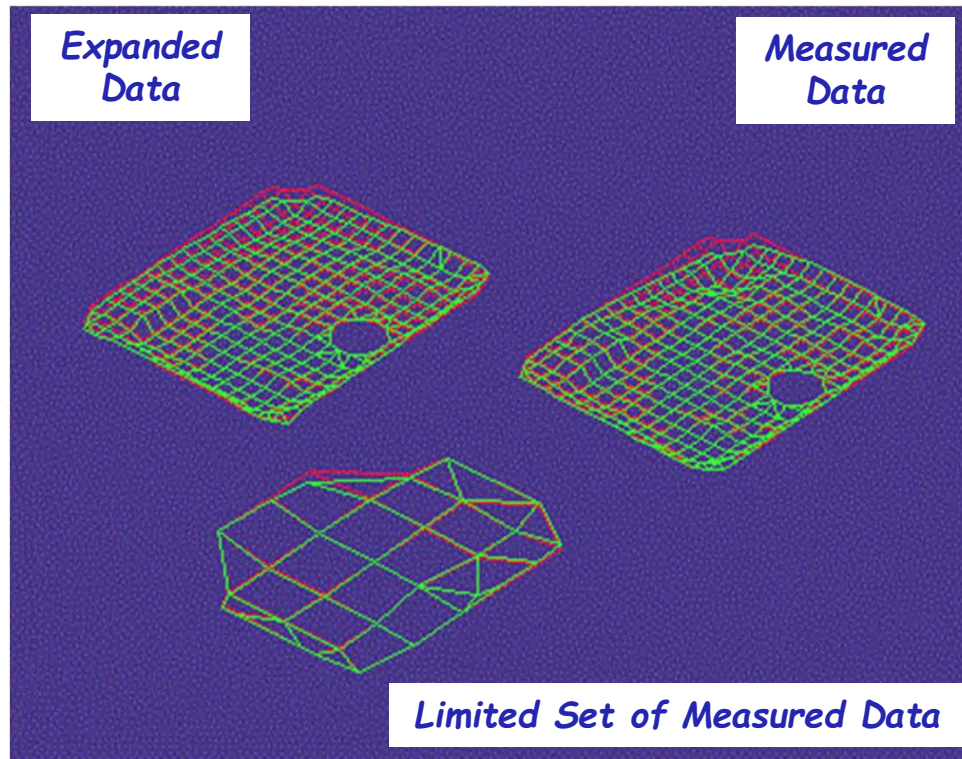
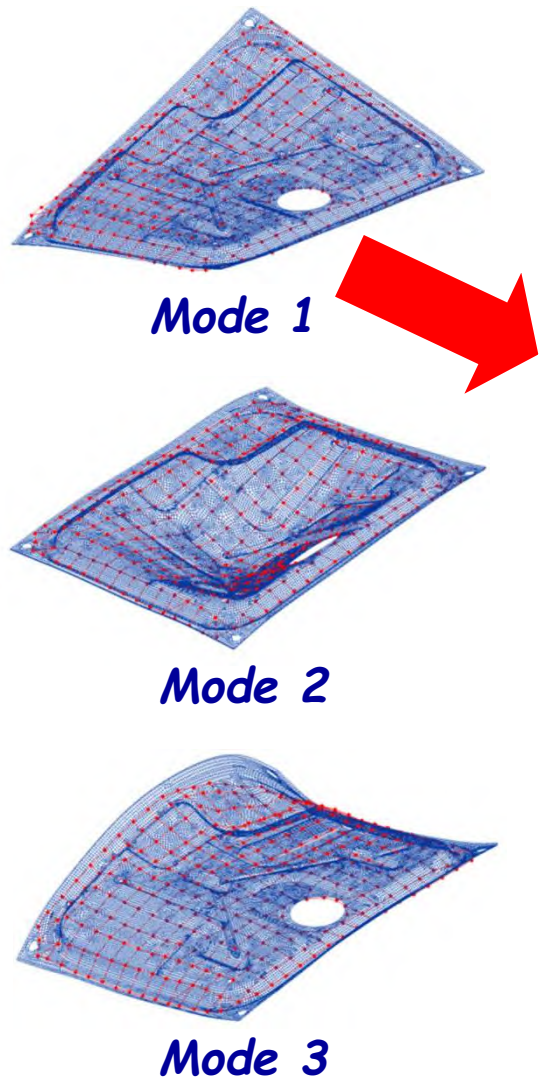
Dynamic Response - Fundamental Change in Approach

Operating (Real-time) displacements are expanded to the full set of analytical degrees of freedom in the finite element model using orthogonal shape based expansion functions.

Provides full field displacement solution



Real-time data expansion - Redstone Arsenal



'a' set

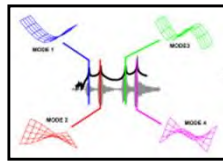
Limited Measurement Points

$$[T_U] = [E_n][E_a]^g$$

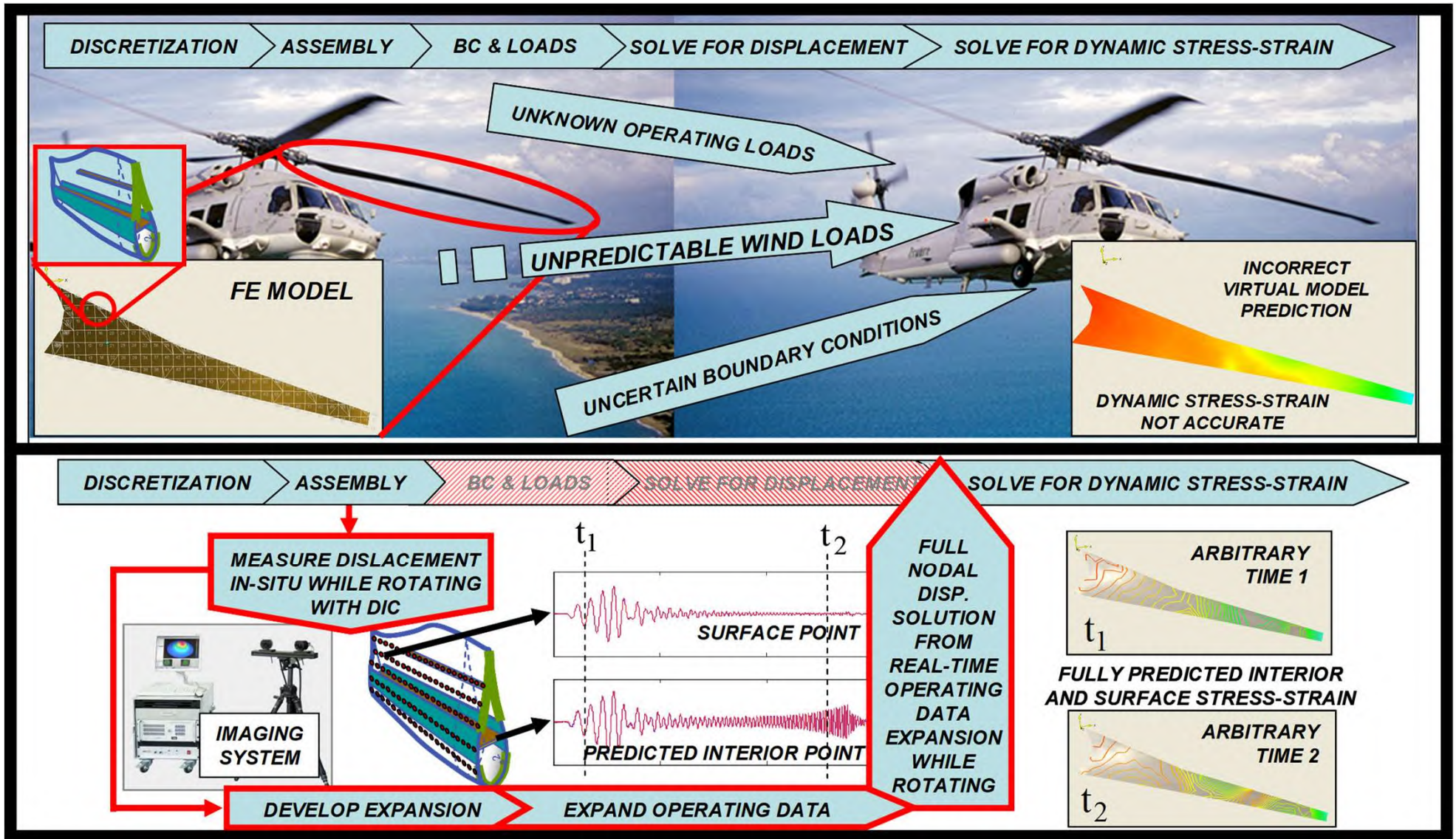
$$[RTO_n] = [T][RTO_a]$$

Expansion is Key Here

ANIMATION

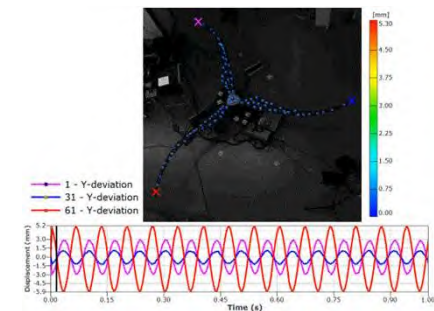
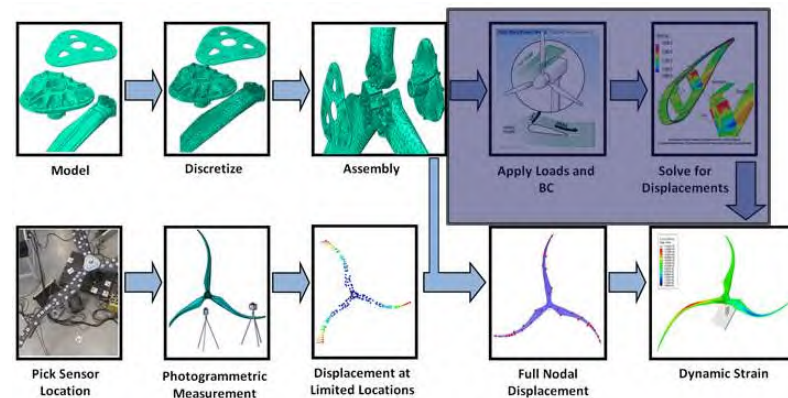
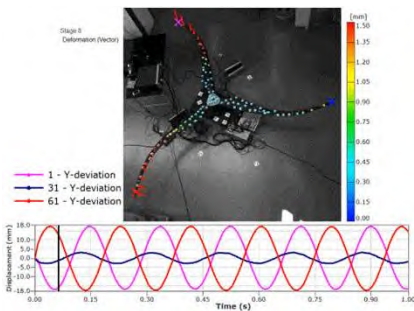
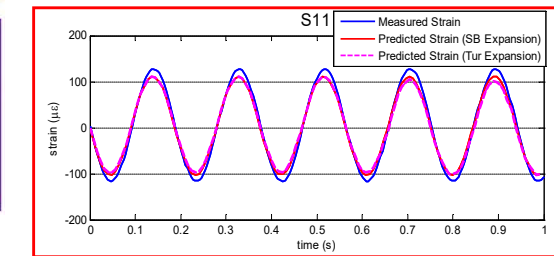
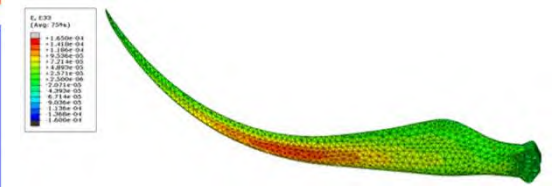
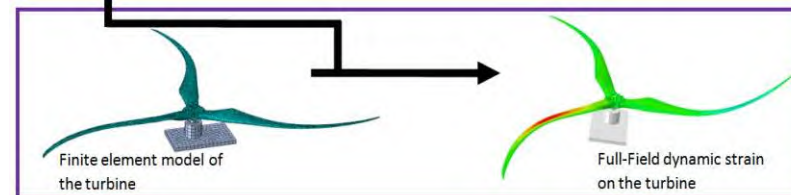
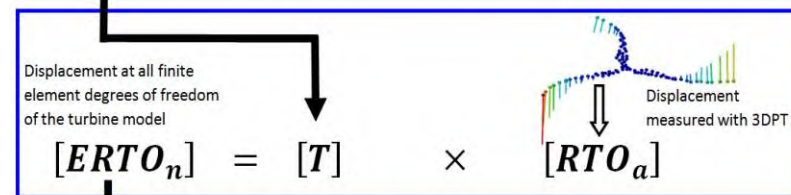
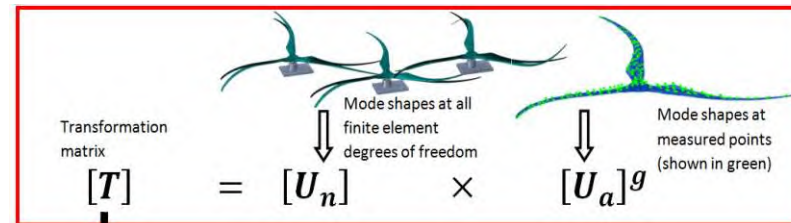
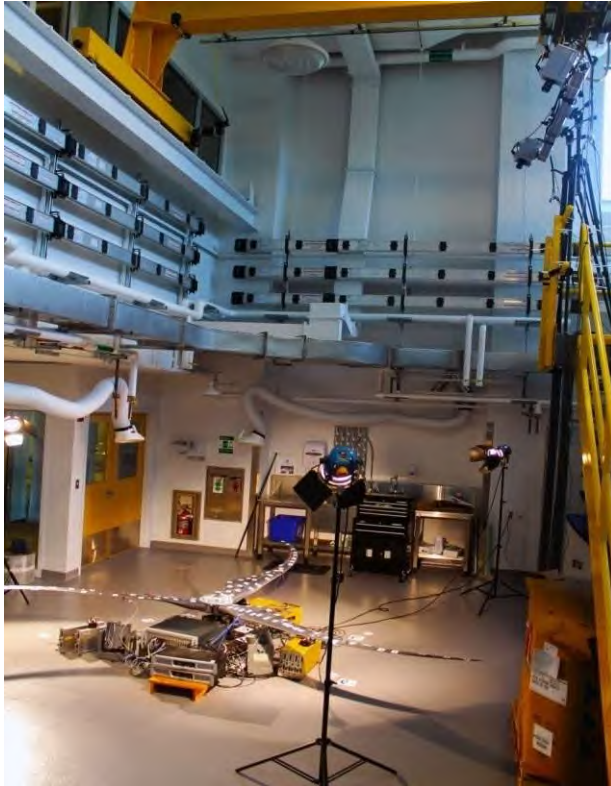


Conventional Approach vs. Alternate Approach

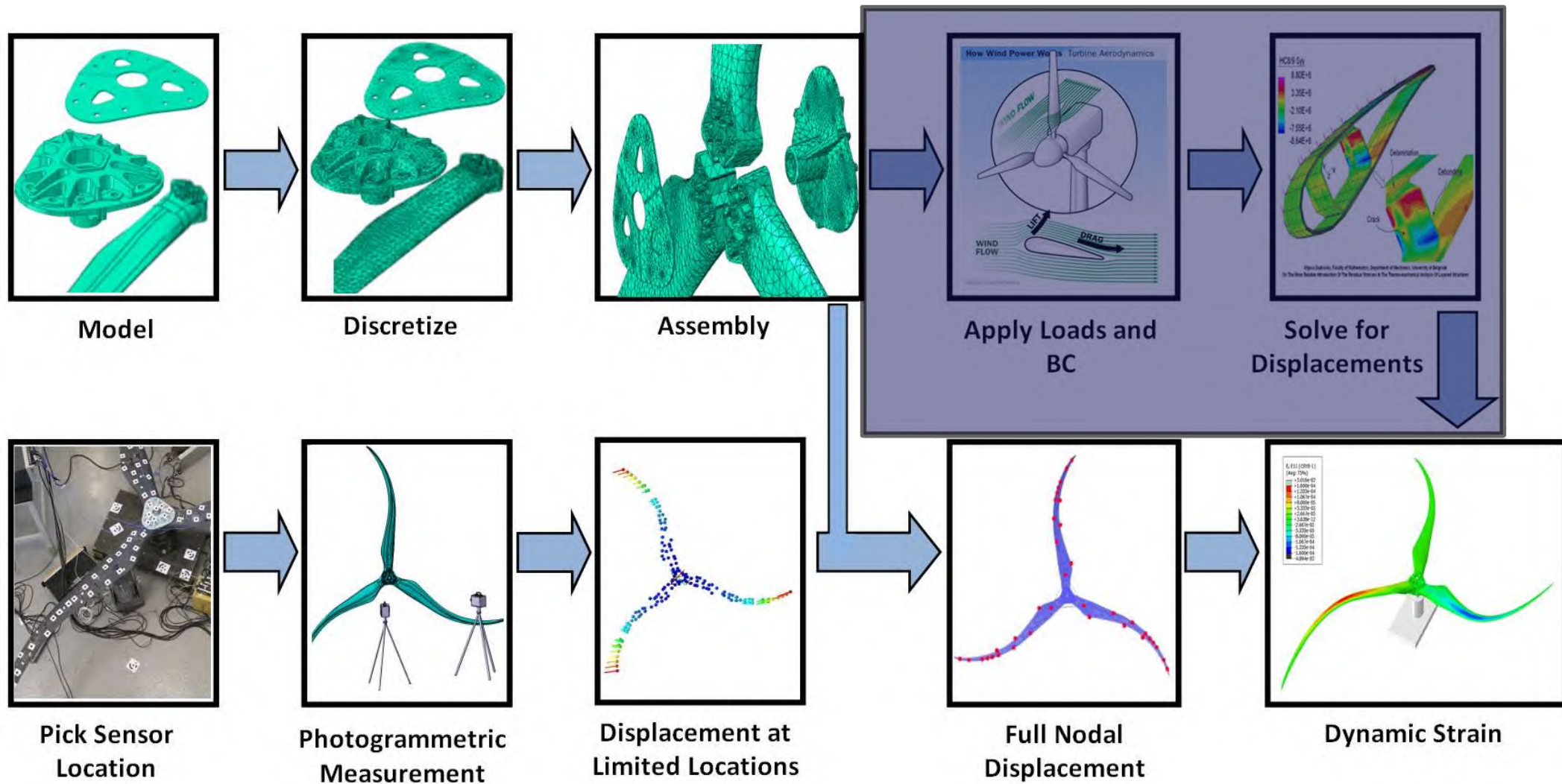


Real Time Dynamic Strain from Limited Measurements

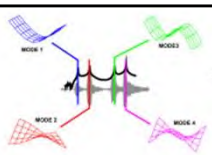
Javad Baqersad, Peter Avitabile



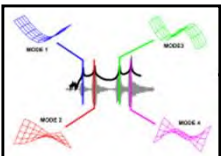
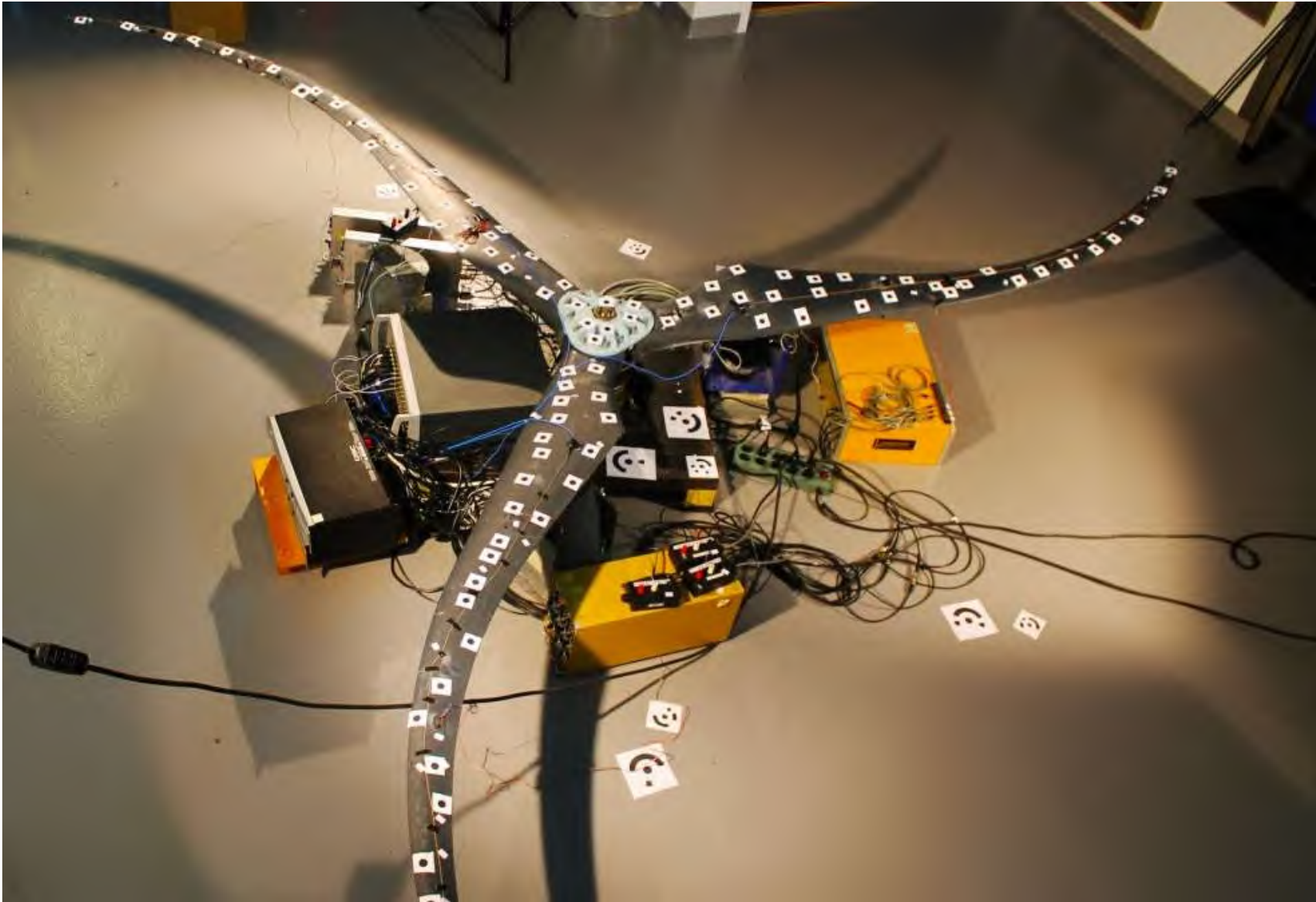
Conventional Approach vs. Alternate Approach



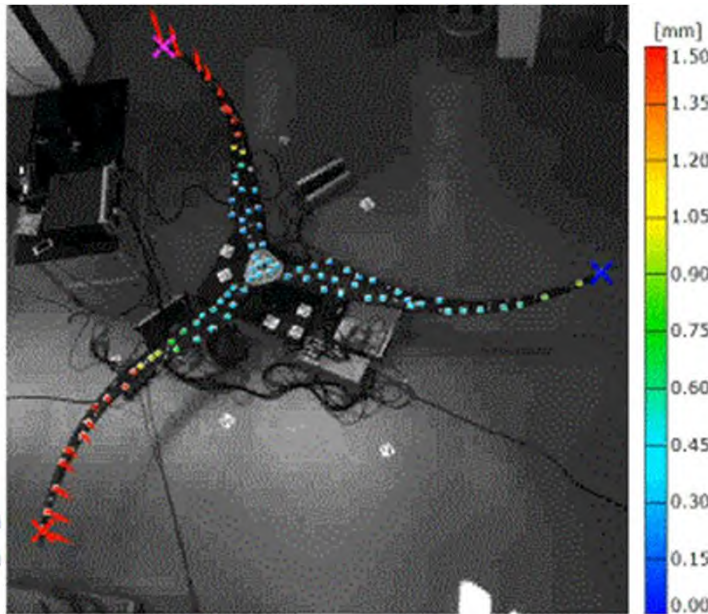
New Approach for Full Field Strain !!!



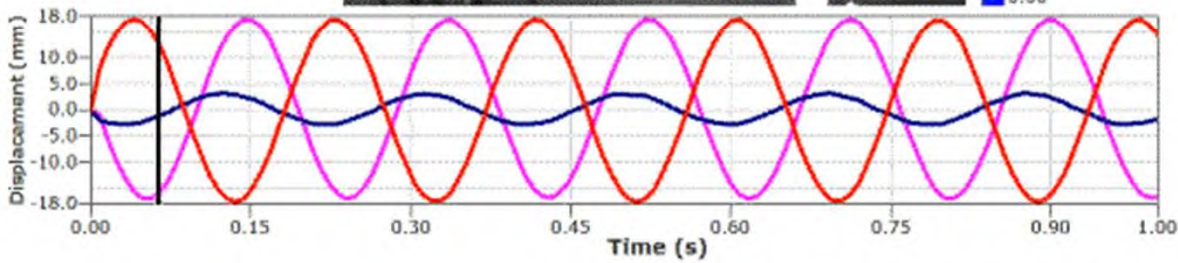
Three Bladed System - Turbine Targets



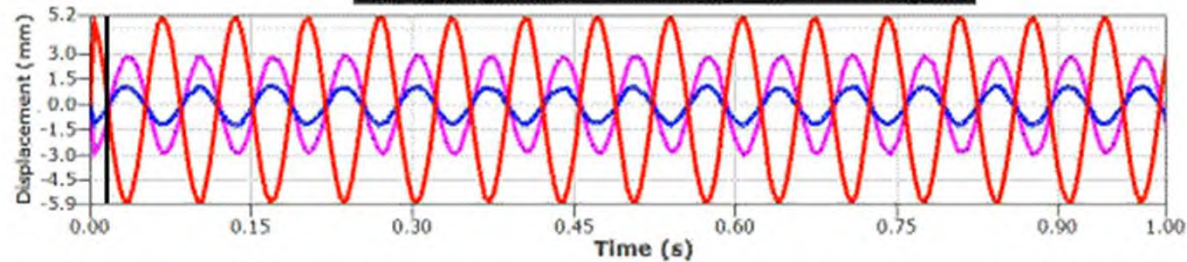
Stage 8
Deformation (Vector)



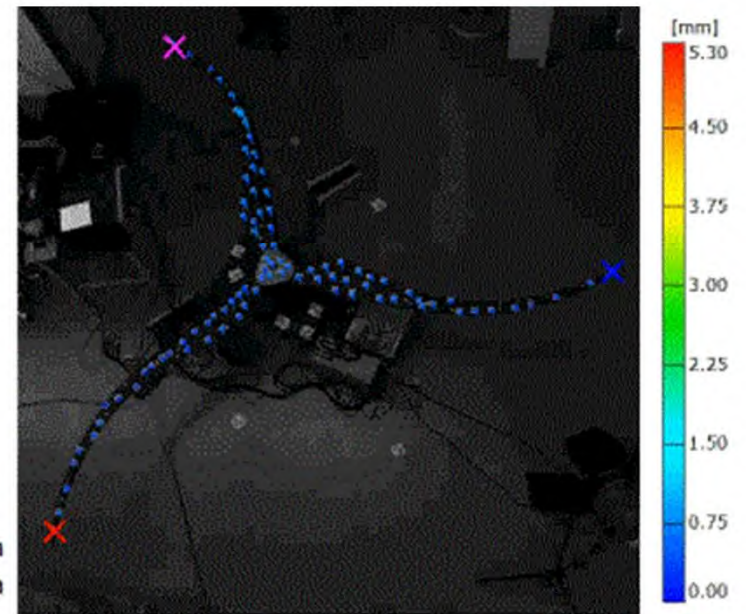
- 1 - Y-deviation
- 31 - Y-deviation
- 61 - Y-deviation



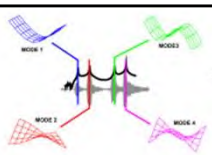
- 1 - Y-deviation
- 31 - Y-deviation
- 61 - Y-deviation



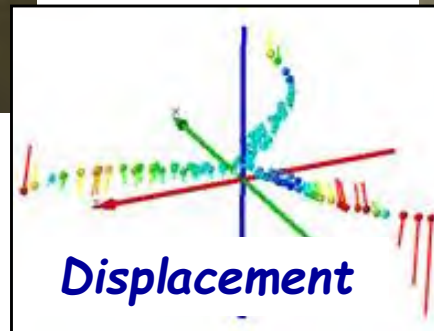
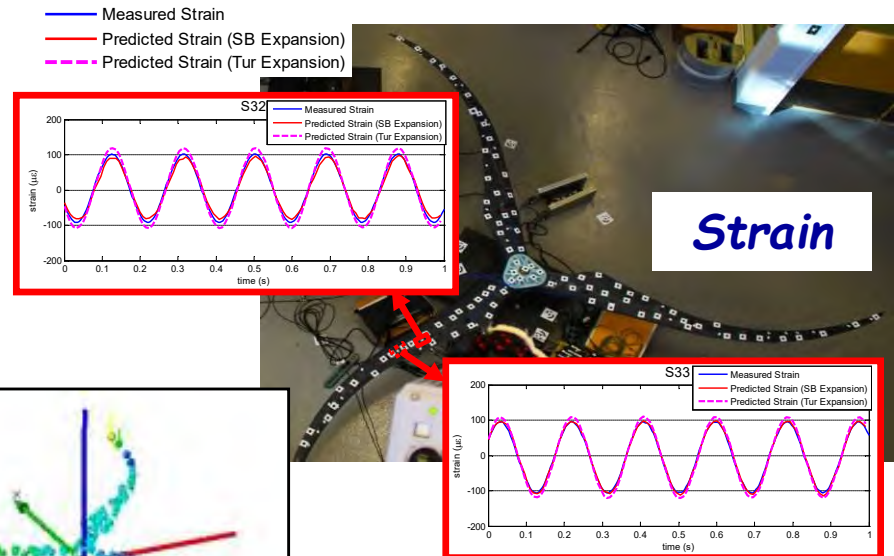
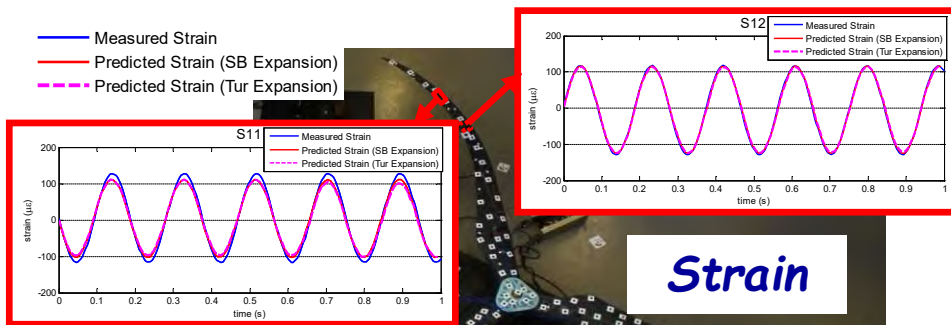
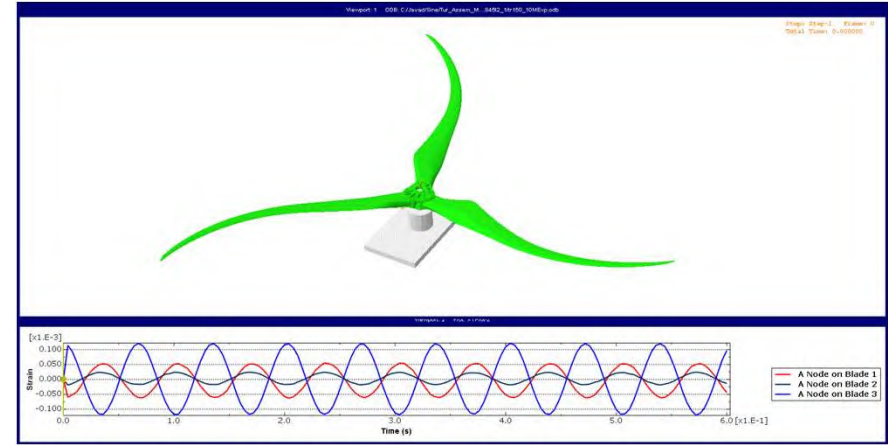
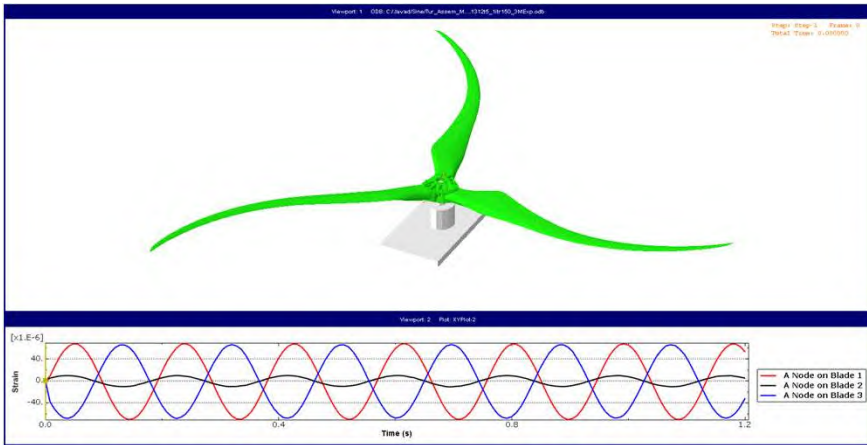
Measured Displacement at Limited Points



ANIMATION



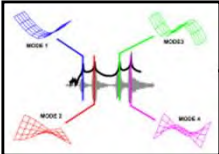
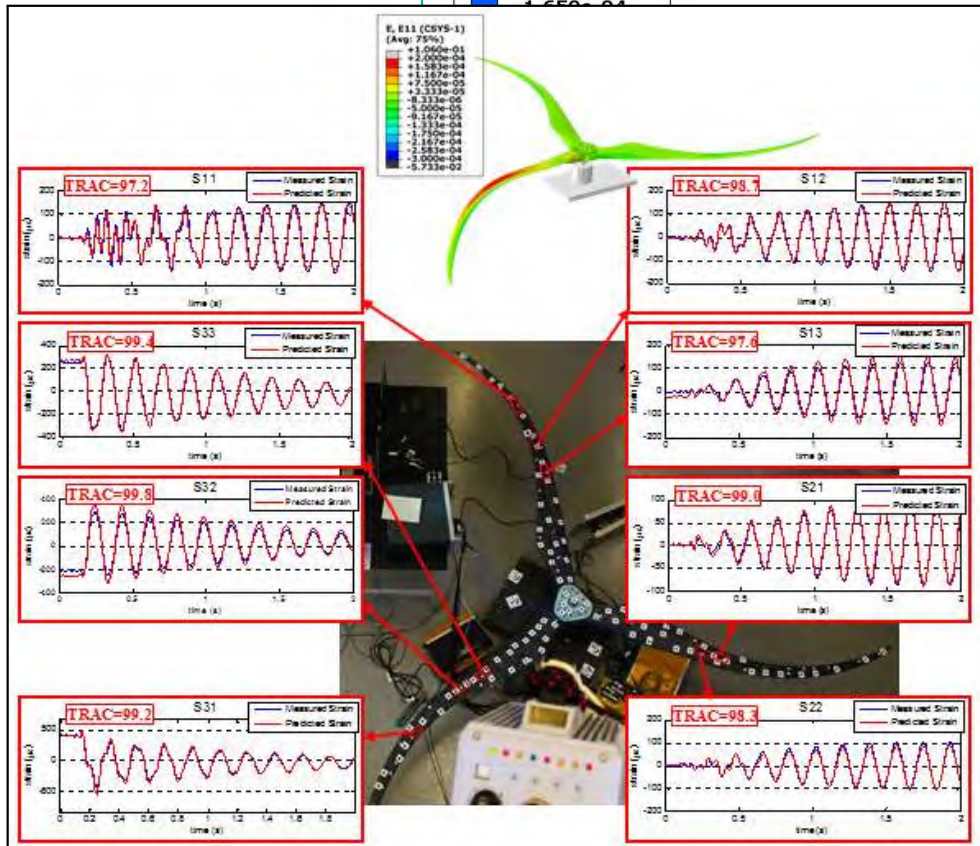
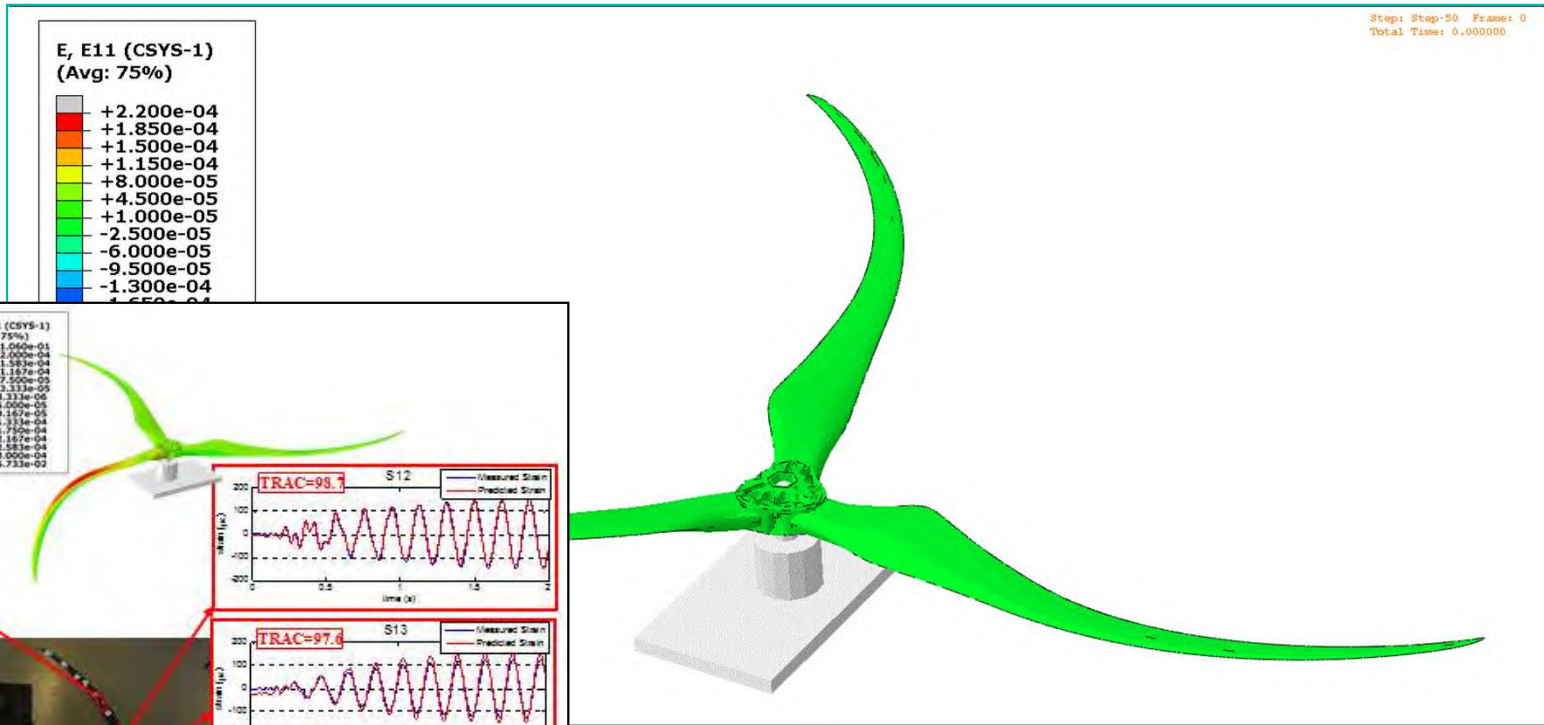
Three Bladed System - Expansion Process



ANIMATION

Turbine Blade - Pluck Test

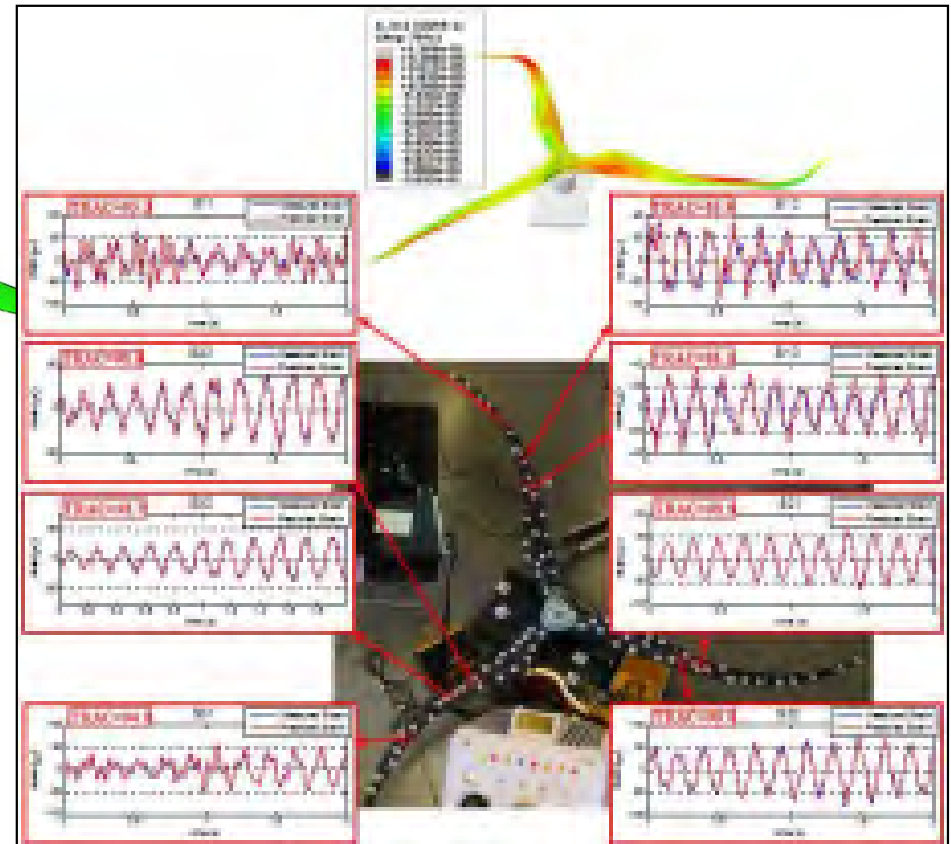
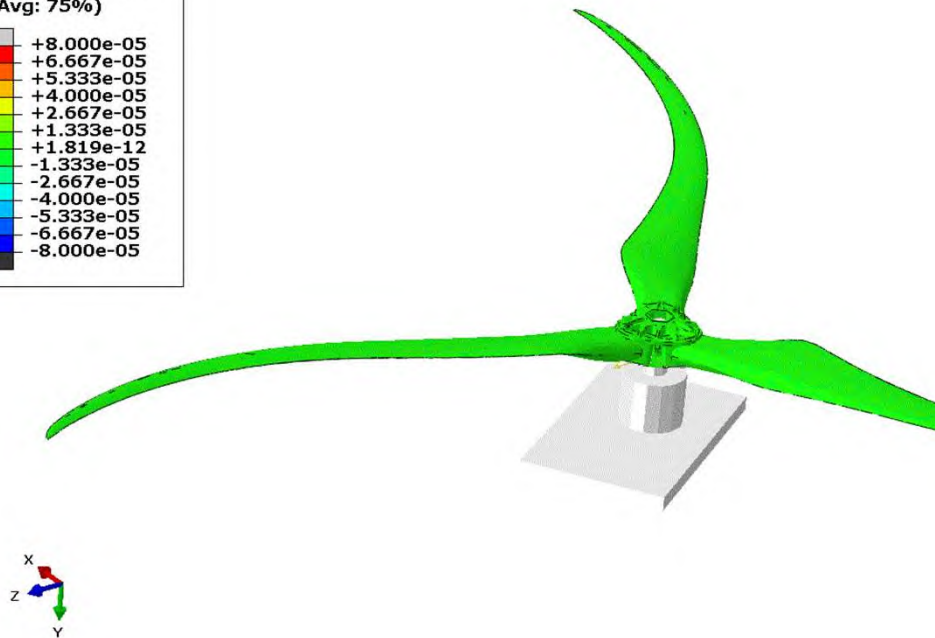
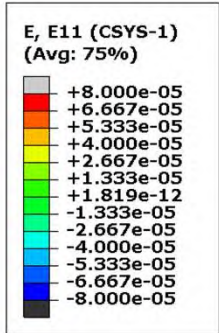
Pluck Test



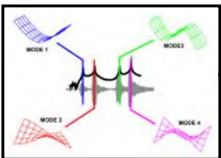
Turbine Blade - Bird Strikes

Step: Step-151 Frame: 0
Total Time: 0.000000

Random Impacts



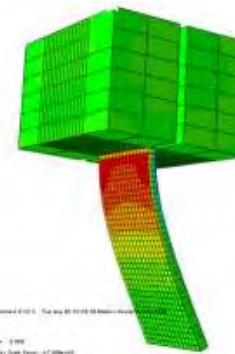
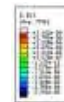
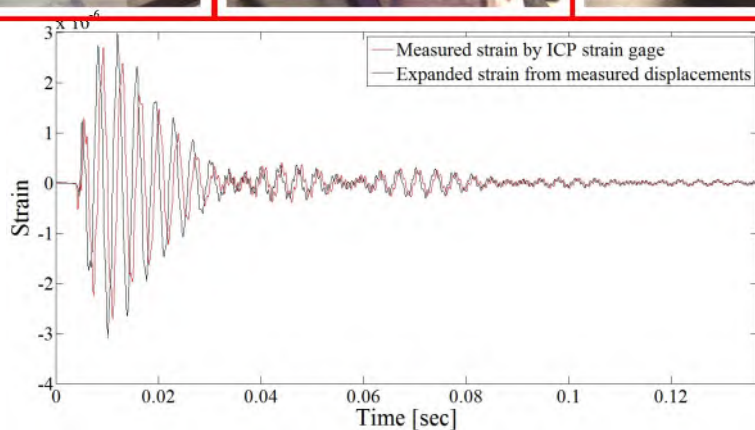
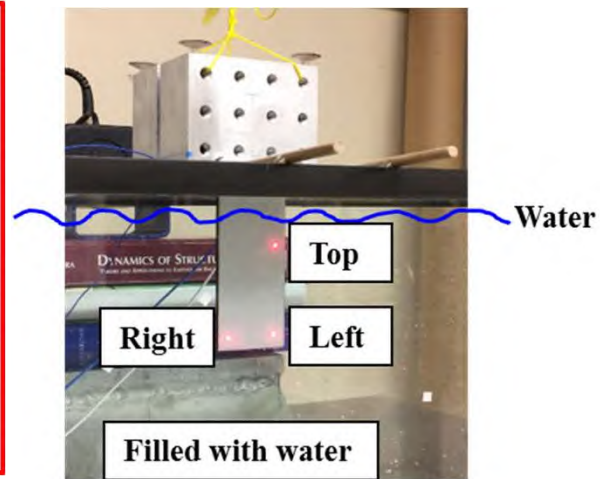
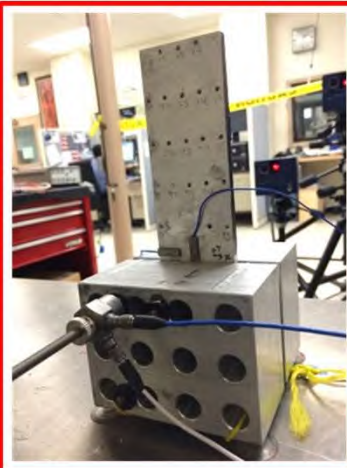
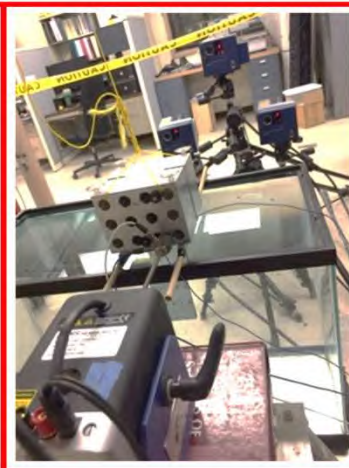
ANIMATION





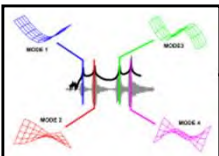
Structural Dynamics and Acoustic Systems Laboratory University of Massachusetts Lowell

Underwater Dynamic Response at Limited Points Expanded to Full-Field Displacement Response



Yuanchang Chen, Peter Avitabile

http://faculty.uml.edu/pavitabile/downloads/IMAC35_Underwater_Expansion_Yuan_102016.pdf

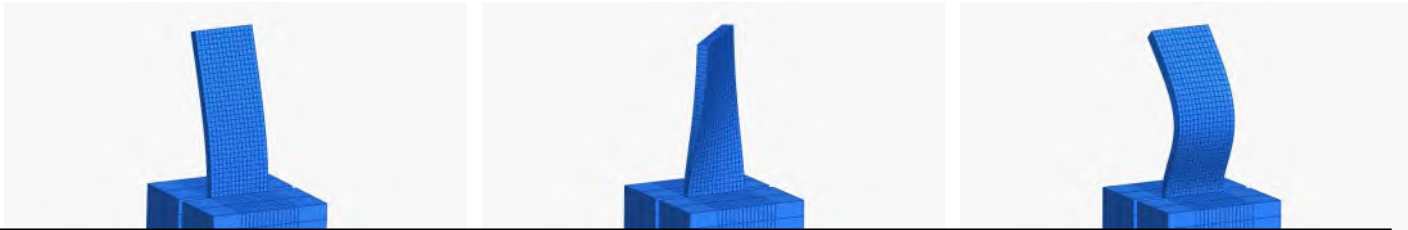


Structure Water - expansion results

Then the displacement at three laser points is converted to the full field using the following equations

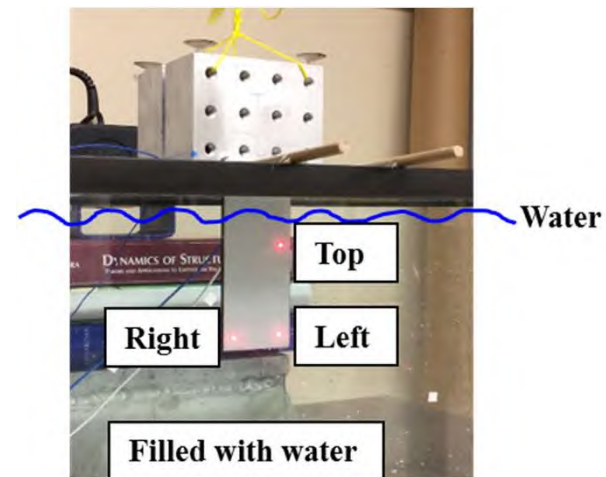
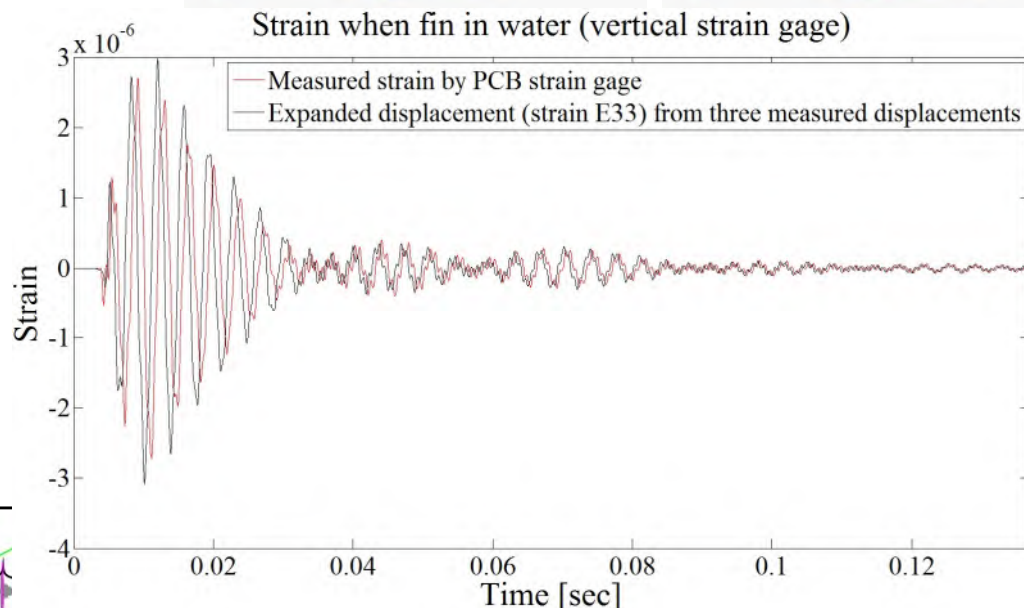
$$\begin{Bmatrix} \mathbf{X}_n \\ \mathbf{X}_d \end{Bmatrix} = \begin{Bmatrix} \mathbf{X}_a \\ \mathbf{X}_a \end{Bmatrix} = [\mathbf{T}]\{\mathbf{X}_a\} \quad [\mathbf{T}_U] = [\mathbf{U}_n][\mathbf{U}_a]^g$$

$[\mathbf{U}_n]$ is composed of the first three flexible modes of the FE model, as follows



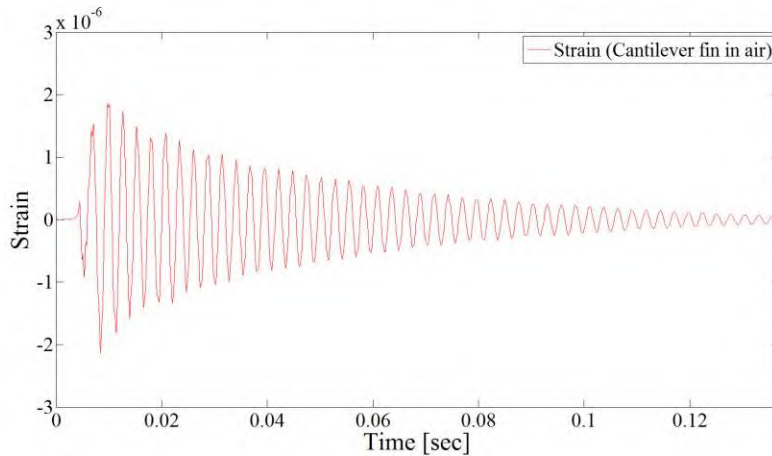
Results	
1..Mode 1, 22.95521 Hz	
2..Mode 2, 23.58787 Hz	
3..Mode 3, 33.17689 Hz	
4..Mode 4, 55.90902 Hz	
5..Mode 5, 56.6845 Hz	
6..Mode 6, 57.63521 Hz	
7..Mode 7, 374.7504 Hz	
8..Mode 8, 1585.737 Hz	
9..Mode 9, 2011.39 Hz	
10..Mode 10, 2580.46 Hz	
11..Mode 11, 4989.329 Hz	
12..Mode 12, 5442.574 Hz	
13..Mode 13, 7987.654 Hz	
14..Mode 14, 8942.454 Hz	
15..Mode 15, 9133.132 Hz	

Then the full field displacement is imported into Abaqus to get the strain

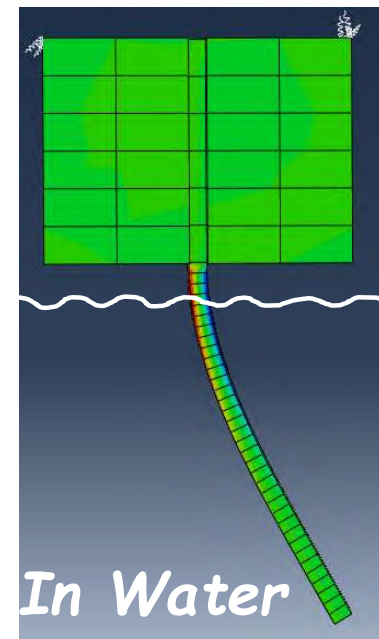
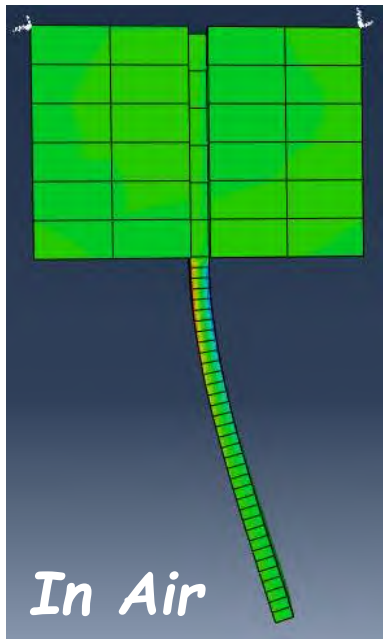
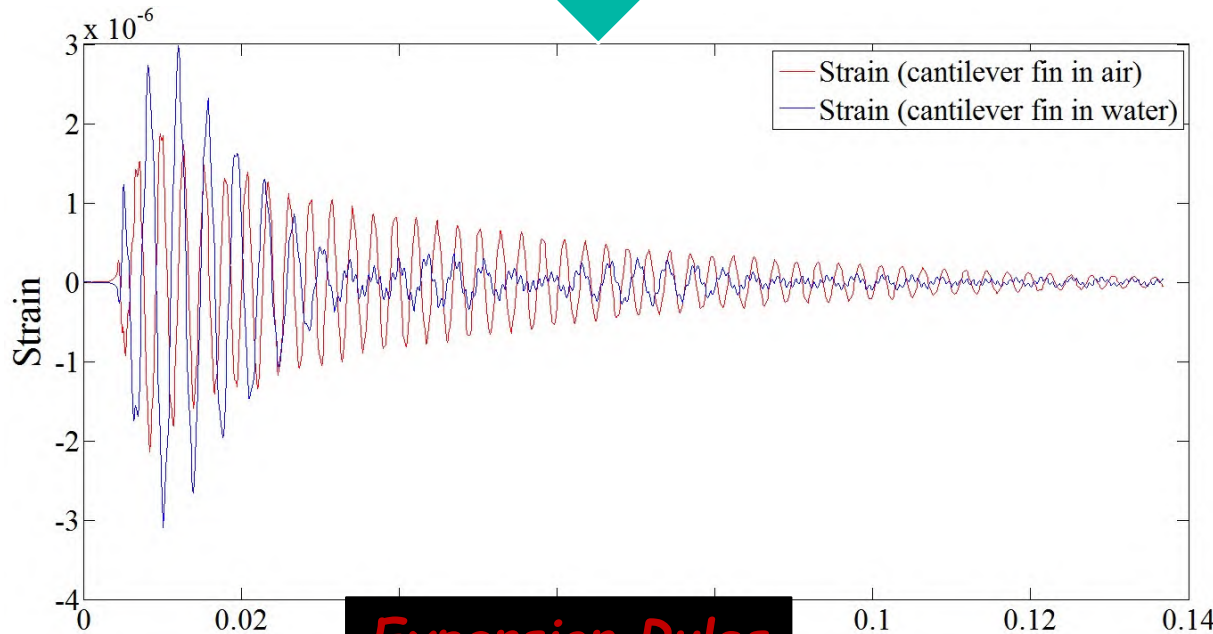
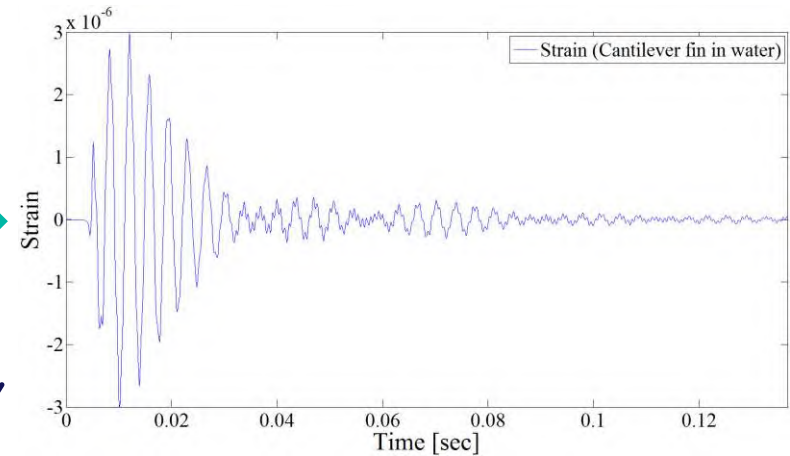


Comparison between in air and in water

Strain in Air



Strain in Water



Expansion Rules

Nonlinear Response using Linear Components with Nonlinear Connections

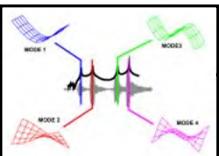
Lou Thibault, Tim Marinone, Julie Harvie, Sergio Obando, Peter Avitabile

*Linear Modal Substructuring with Nonlinear Connections, SEM Handbook of Experimental Structural Dynamics,
DOI:10.1007/978-1-4614-4547-0_34, July 2022*

*Integrating Test and FEA for Enhanced
Structural Dynamic Modeling Applications*

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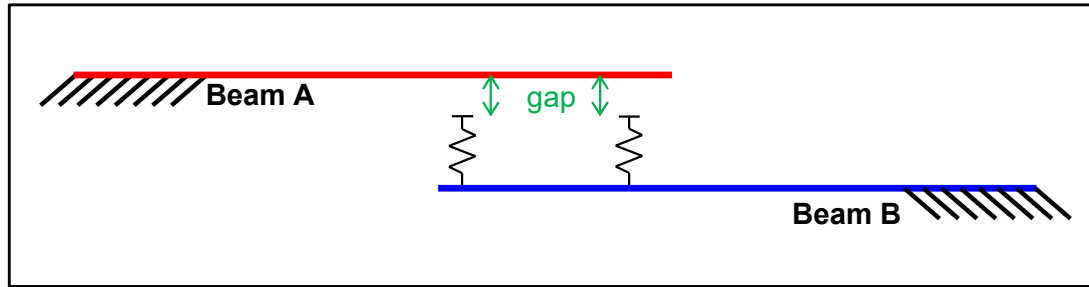
*Peter Avitabile - Mechanical Engineering
Structural Dynamics And Acoustic Systems Lab*



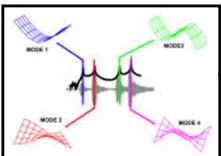
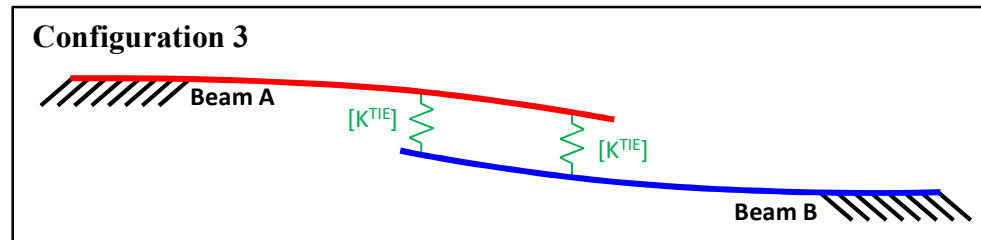
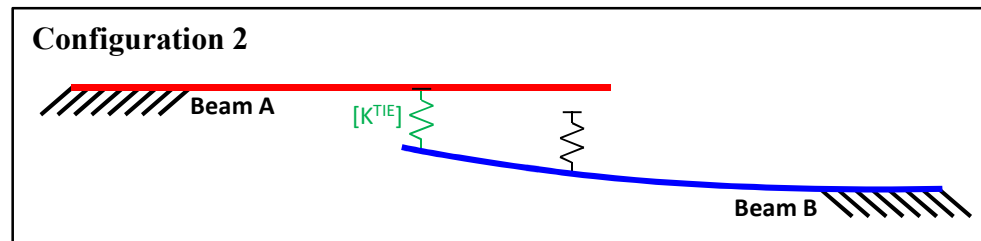
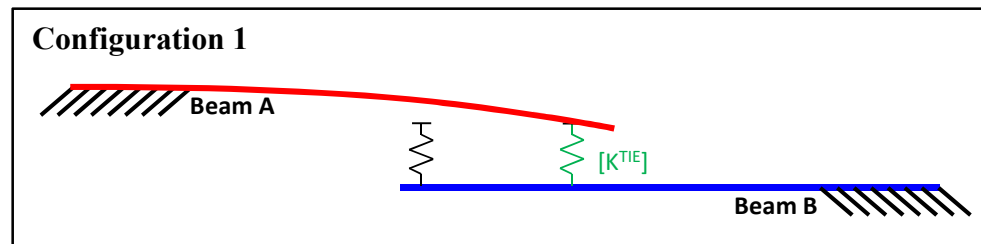
System Modeling

Uncoupled system

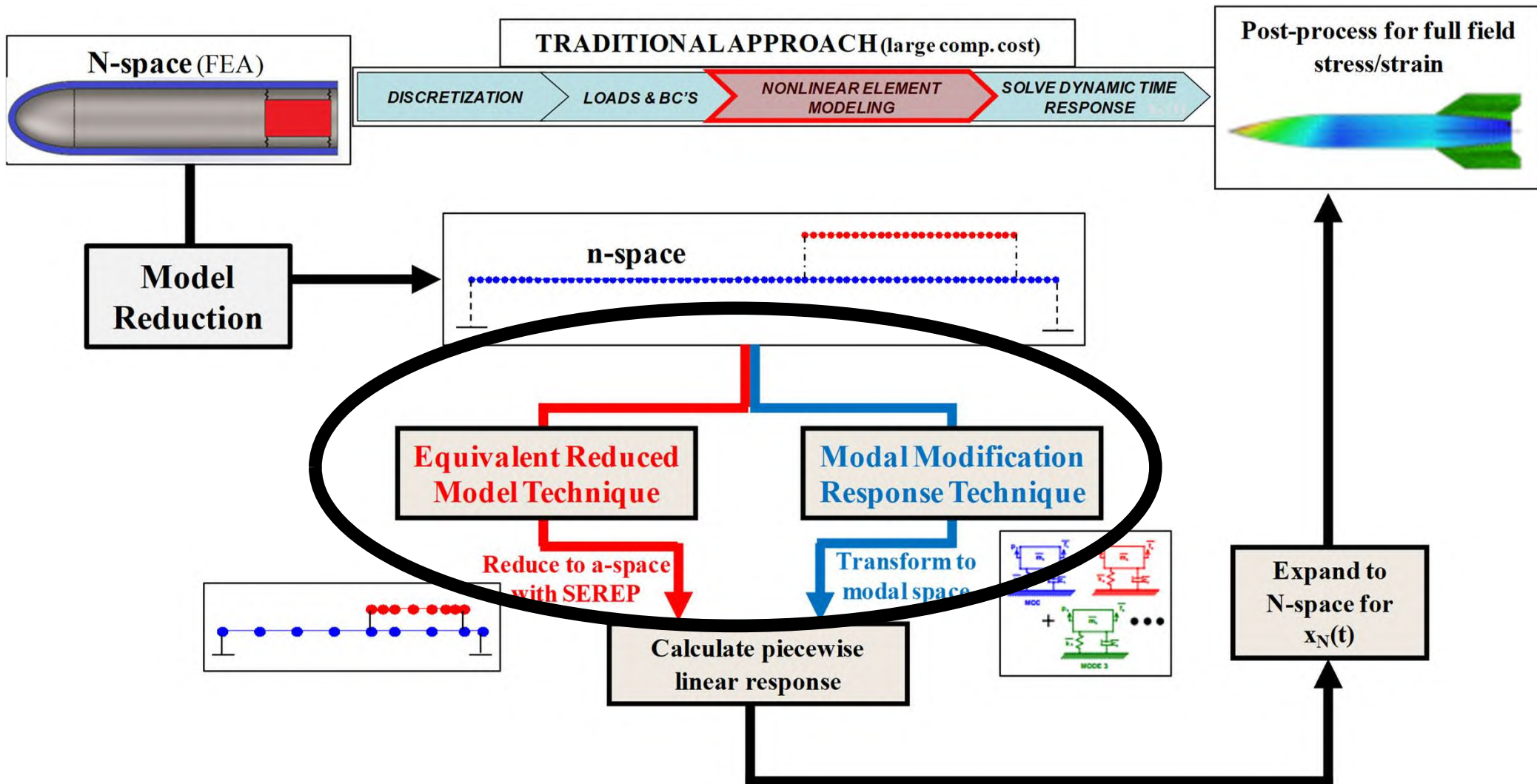
- Discrete coupling terms must be introduced to generate possible configurations



- Structural Dynamic Modifications (SDM) and Component Mode Synthesis (CMS)
- Physical System Modeling
- Mode Contribution Matrix

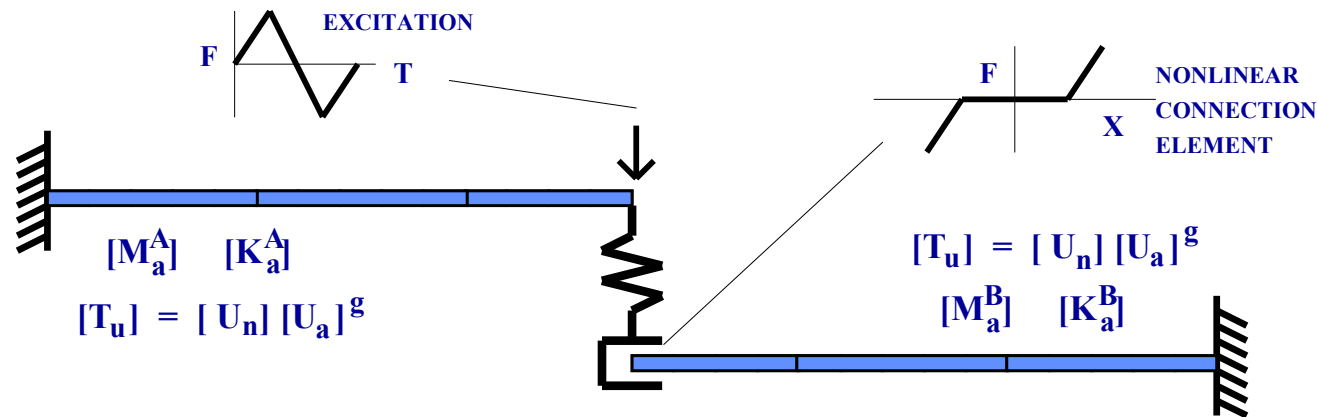


A Piecewise Linear Approach to Nonlinear Response



Lou Thibault, Tim Marinone, Julie Harvie, Sergio Obando, Peter Avitabile

Equivalent Reduced Order Model Technique



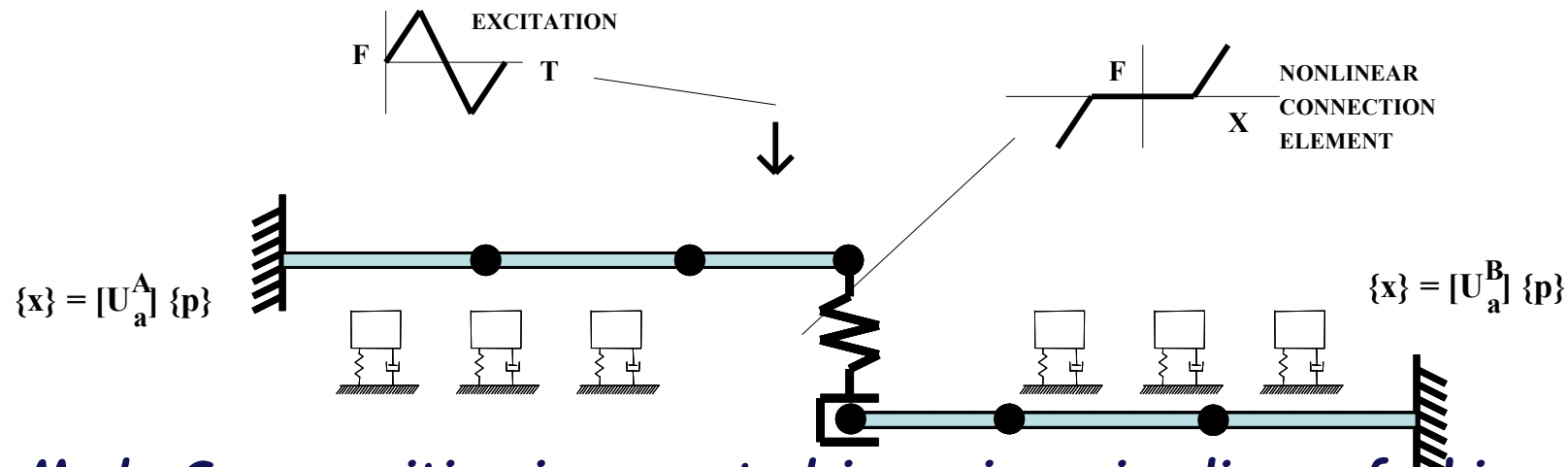
Direct Integration of the Reduced Physical Equations of Motion can be performed for a reduced component model representation. Numerical integration proceeds as typically performed but with drastically reduced model size. The technique greatly improves the efficiency of the process and with optimization can integrate test.

$$\begin{bmatrix} [M_a^A] \\ [M_a^B] \end{bmatrix} \begin{Bmatrix} \ddot{x}_a^A \\ \ddot{x}_a^B \end{Bmatrix} + \begin{bmatrix} [K_a^A] \\ [K_a^B] + [K_{TIE}] \end{bmatrix} \begin{Bmatrix} x_a^A \\ x_a^B \end{Bmatrix} = \begin{Bmatrix} f_a^A \\ f_a^B \end{Bmatrix}$$

$$[M_a] = [U_a]^g T [U_a]^g$$

$$[K_a] = [U_a]^g T [\Omega^2] [U_a]^g$$

Modal Modification Response Technique



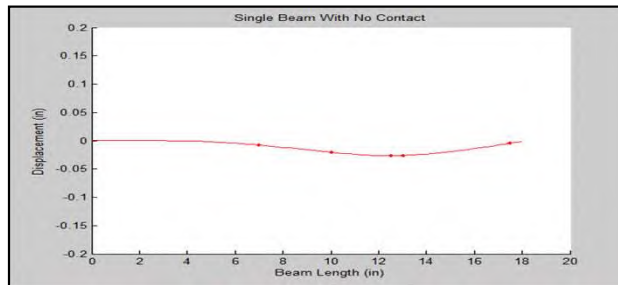
Mode Superposition is executed in a piecewise linear fashion depending on the "state" of the nonlinear connection element. Once the linear state changes, a structural modification is performed to update the characteristics of the system along with updated initial conditions to proceed on with the numerical integration. Current state of connection elements are monitored at each integration step

$$\begin{bmatrix} \begin{bmatrix} \mathbf{I}_m^A \\ \mathbf{I}_m^B \end{bmatrix} \begin{Bmatrix} \ddot{\mathbf{p}}^A \\ \ddot{\mathbf{p}}^B \end{Bmatrix} + \begin{bmatrix} \begin{bmatrix} \Omega_m^2{}^A \\ \Omega_m^2{}^B \end{bmatrix} \\ \mathbf{U}^T [\mathbf{K}_{TIE}] \mathbf{U} \end{bmatrix} \begin{Bmatrix} \mathbf{p}^A \\ \mathbf{p}^B \end{Bmatrix} = \begin{Bmatrix} \mathbf{U}^A{}^T \mathbf{f}^A \\ \mathbf{U}^B{}^T \mathbf{f}_a^B \end{Bmatrix}$$

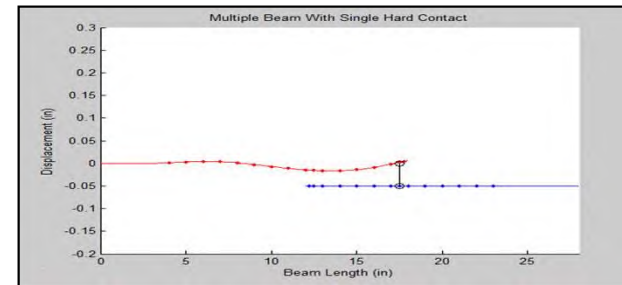
$$\{\mathbf{p}\}_j = [\mathbf{U}]_j^g \{\mathbf{x}\}^{(i-1)} \quad \{\dot{\mathbf{p}}\}_j = [\mathbf{U}]_j^g \{\dot{\mathbf{x}}\}^{(i-1)}$$

Linear and Nonlinear Response - Highly Reduced Order Models

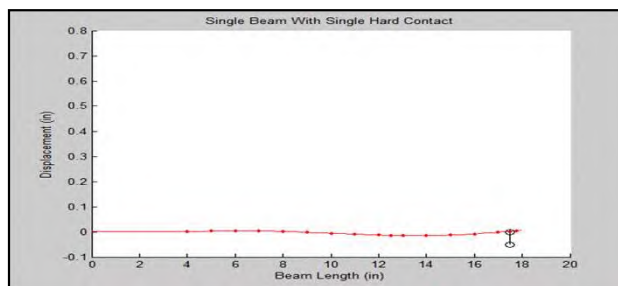
Single Beam Linear Response



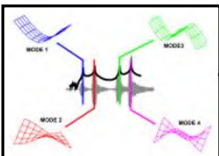
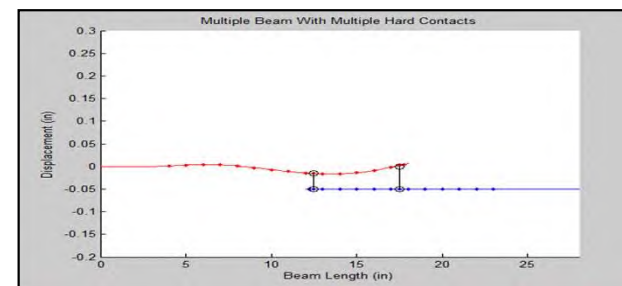
Two Beam Single Contact



Single Beam Single Contact

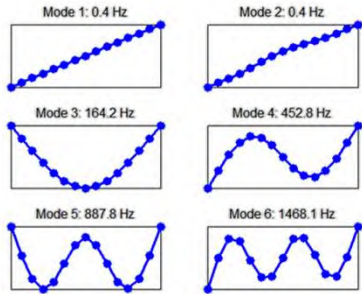


Two Beam Multiple Contact



Previous Work - Expansion of Coupled Mode Shapes

Component A



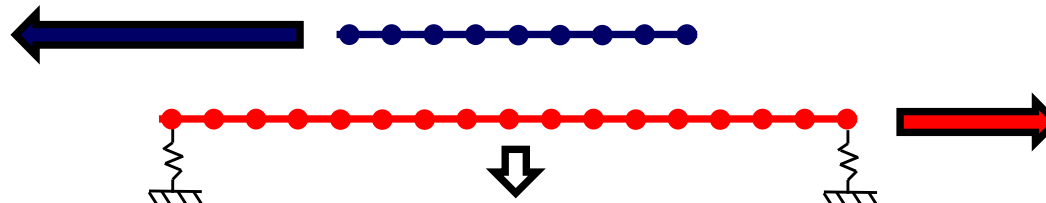
Tu_A



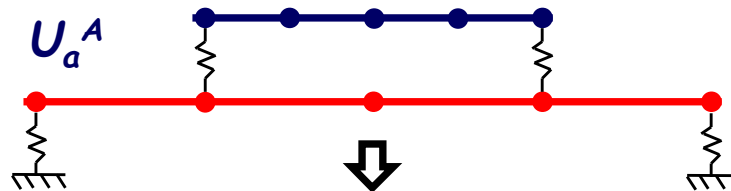
$$U_n^A = Tu^A Ua^A$$

Expansion Matrix

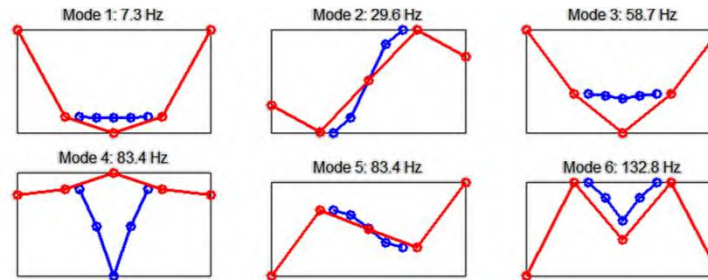
Full Space Component Models



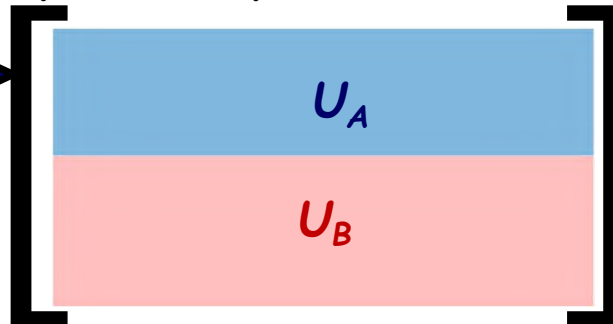
Reduced Order System Model



Reduced Order System Modes

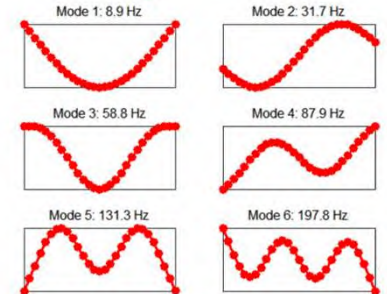


Expanded System Modes



$$\begin{bmatrix} U_n \end{bmatrix} =$$

Component B

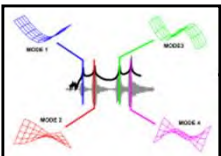


Tu_B



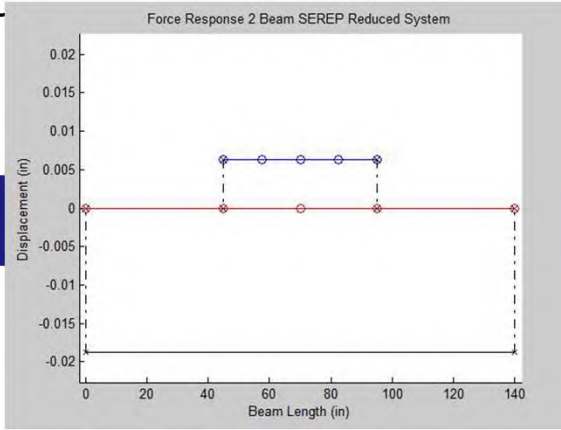
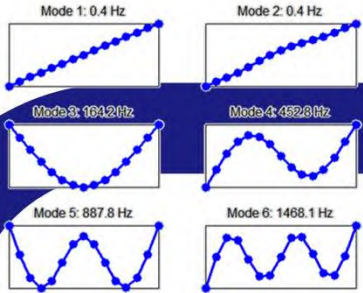
$$U_n^B = Tu^B Ua^B$$

Expansion Matrix

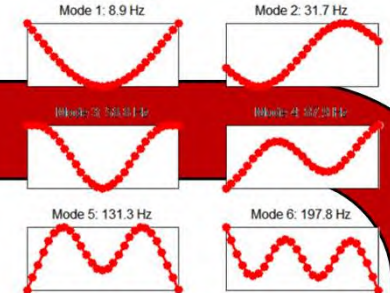


Previous Work - Expansion Methodology

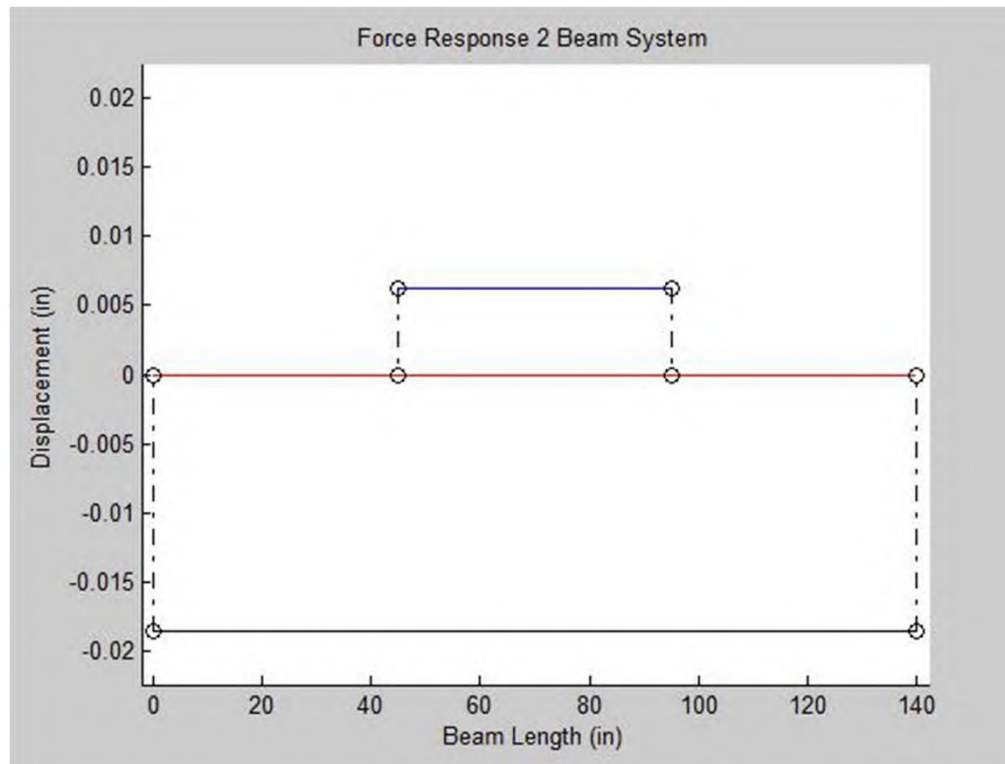
Component A



Component B



Reduced Order System Model Response



Tu_A



$$U_n^A = Tu^A Ua^A$$

Expansion Matrix

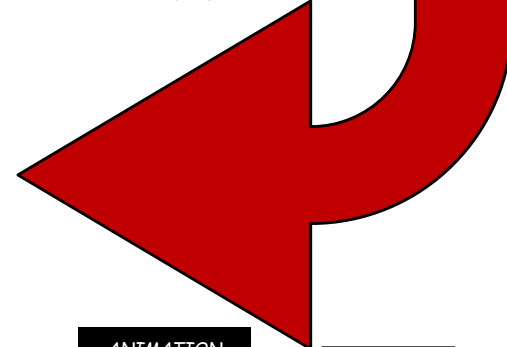


Tu_B

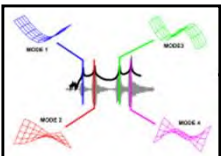


$$U_n^B = Tu^B Ua^B$$

Expansion Matrix

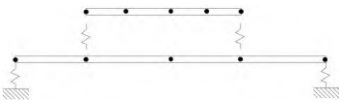


ANIMATION

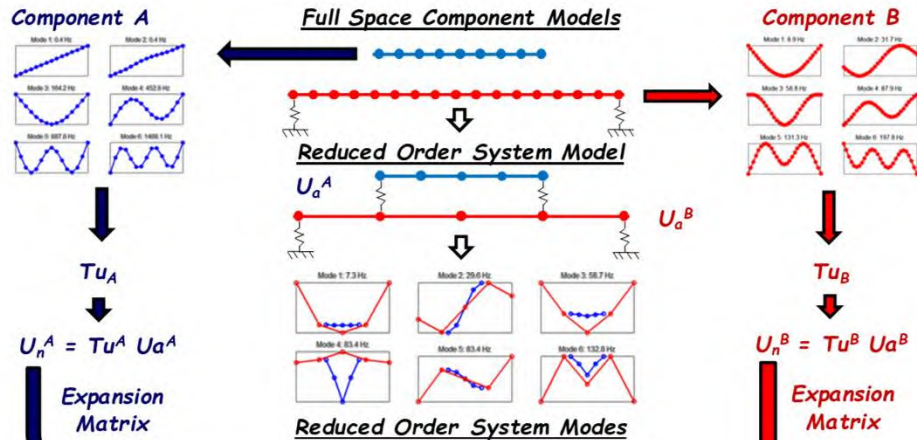


Expansion of System Modes from Component Info

Linear

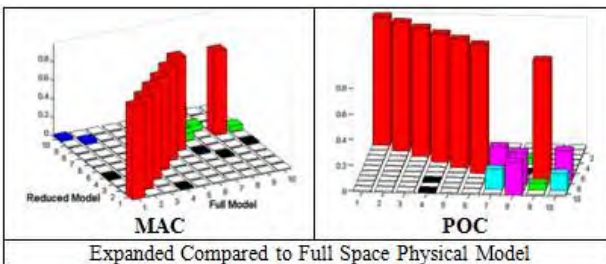
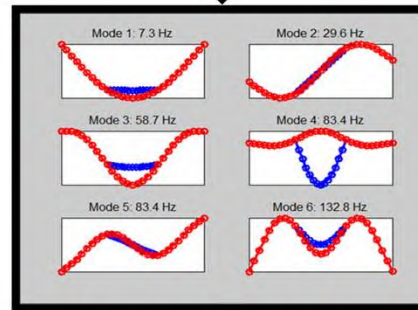
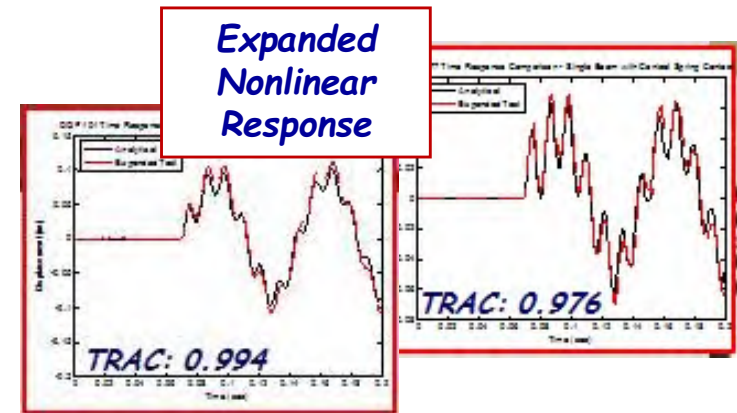
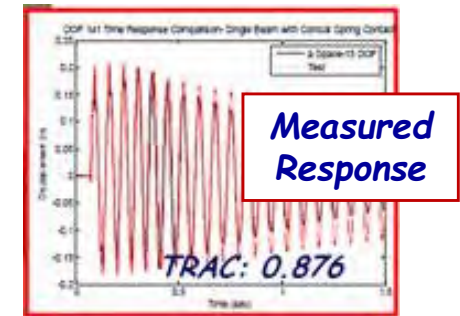
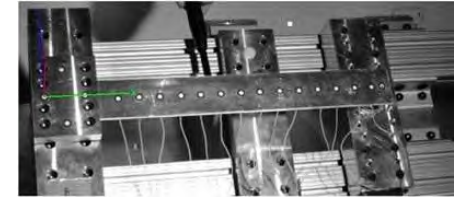


Mode	Full Space System Model Comparison				
	Frequency (Hz)			MAC	POC
	Full Physical 'n' space	Expanded model from 'a' space	% Diff		
1	7.3	7.2	0.01	1.000	1.000
2	29.5	29.5	0.02	1.000	1.000
3	58.7	58.7	0.01	1.000	1.000
4	81.7	83.36	2.00	0.997	0.998
5	83.3	83.37	0.10	1.000	0.999
6	132.8	132.8	0.00	1.000	0.999
7	191.8	316.4	65.0	0.073	0.253
8	273.8	680.8	148	0.064	0.252
9	306.2	4461.2	1356	0.002	0.058
10	398.7	4977.3	1148	0.000	0.000



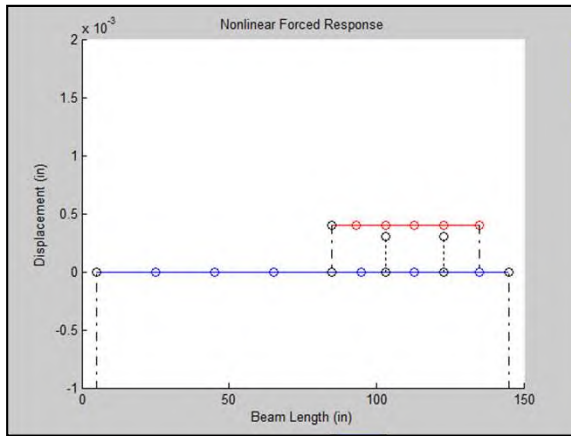
$$[U_n] = \begin{bmatrix} U_A \\ U_B \end{bmatrix}$$

Nonlinear

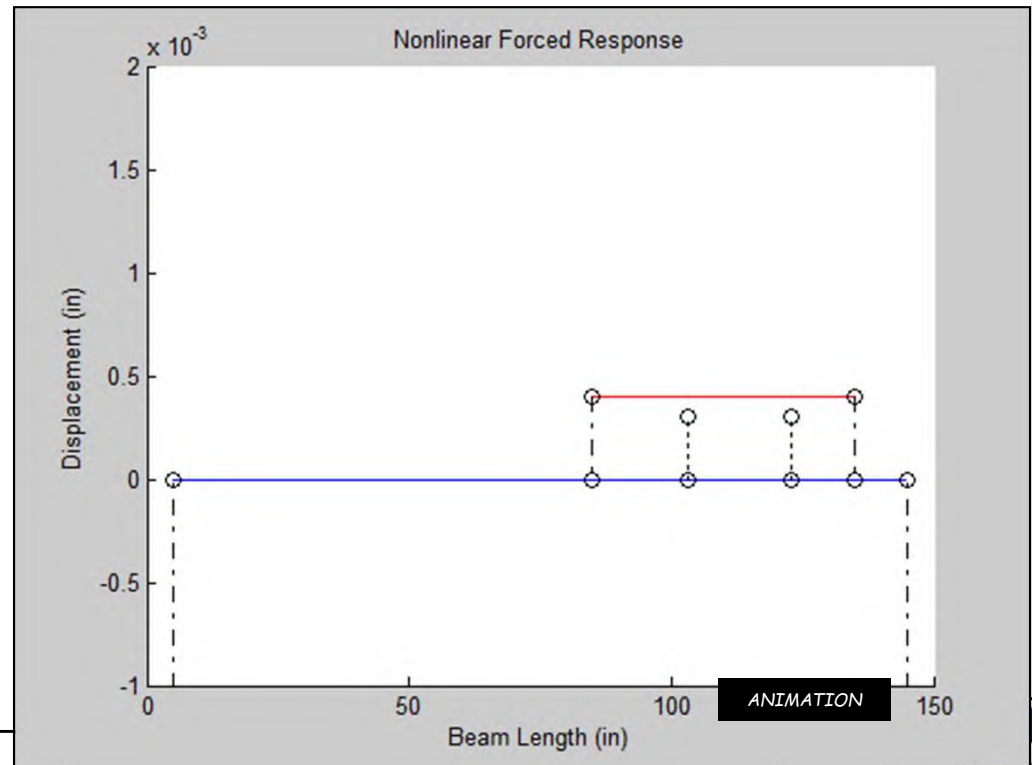


Nonlinear Response - Highly Reduced Order Models & Expansion

**Reduced Order
Nonlinear Response**



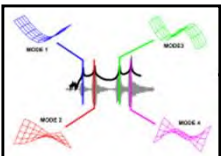
**Expanded
Full Field
Nonlinear Response**



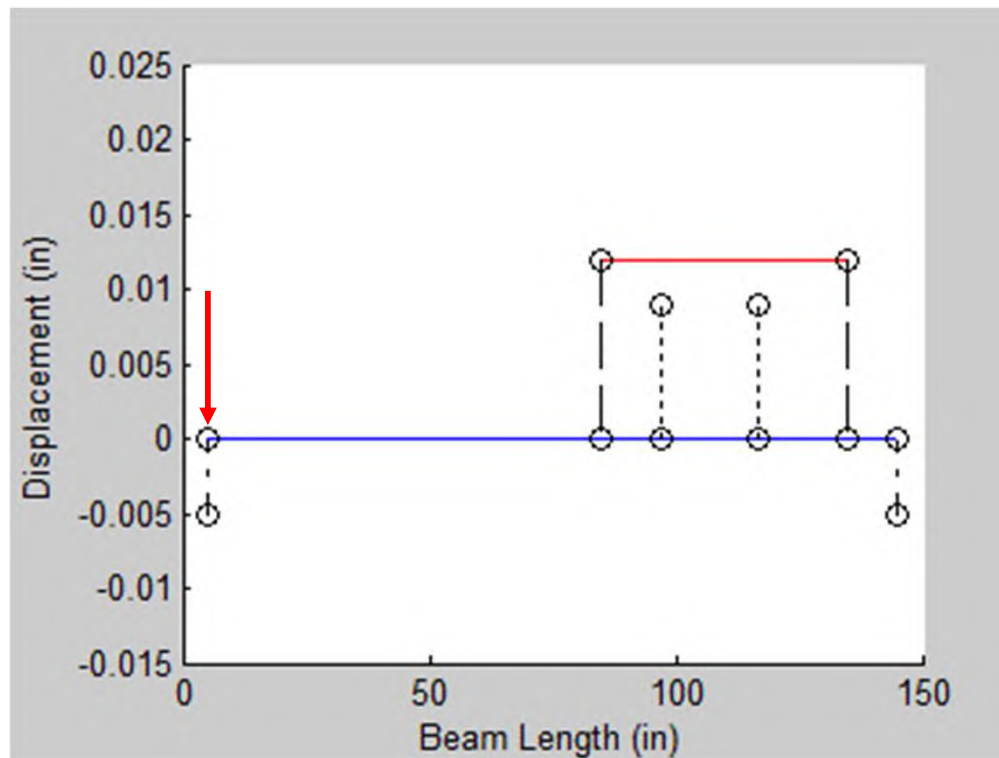
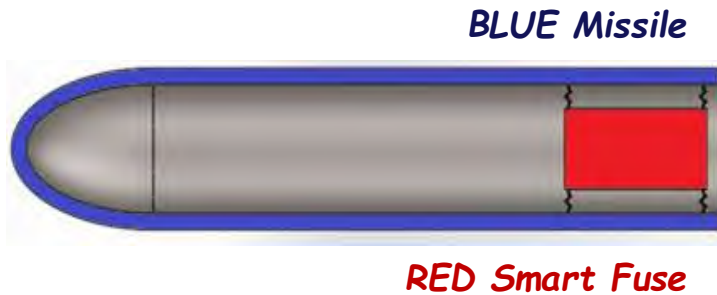
Expansion is Key Here

*Integrating Test and FEA for Enhanced
Structural Dynamic Modeling Applications*

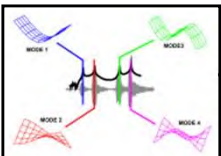
Structural Dynamics And Acoustic Systems Lab



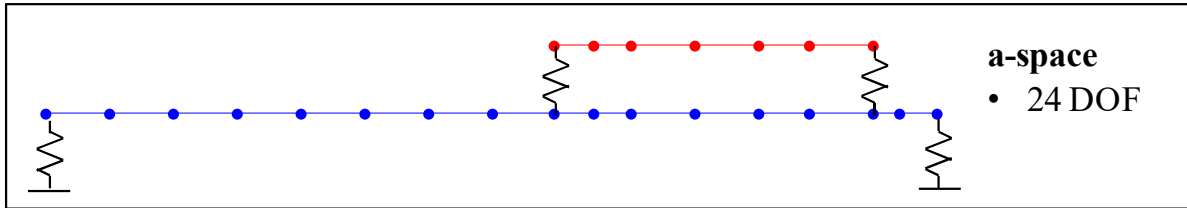
Typical Response Scenario



- Majority of response dominated by linear system response
- Points in time where discrete nonlinearities exist becomes an issue
- Large finite element models become computationally prohibitive to run more than a few scenarios

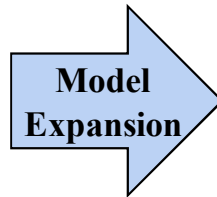
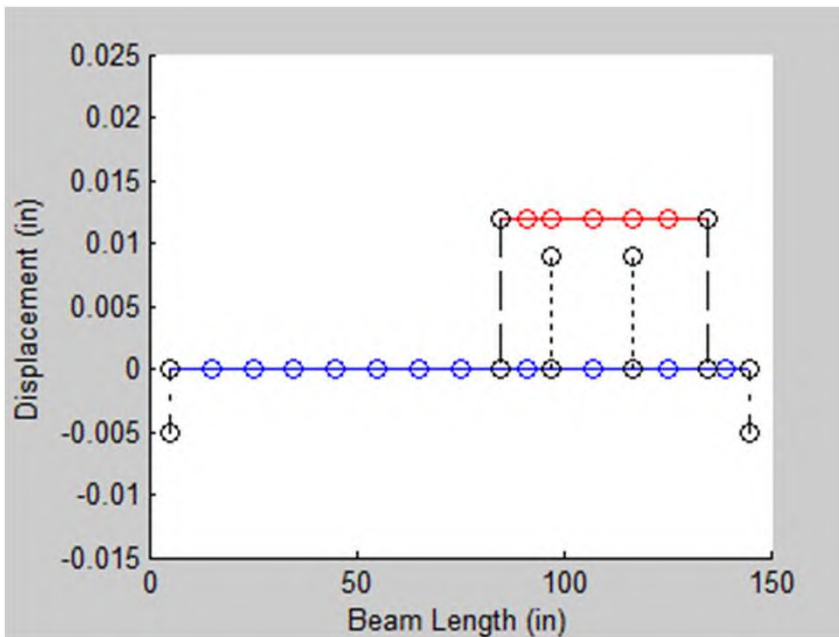


Case 2 - Model reduced to "a" space

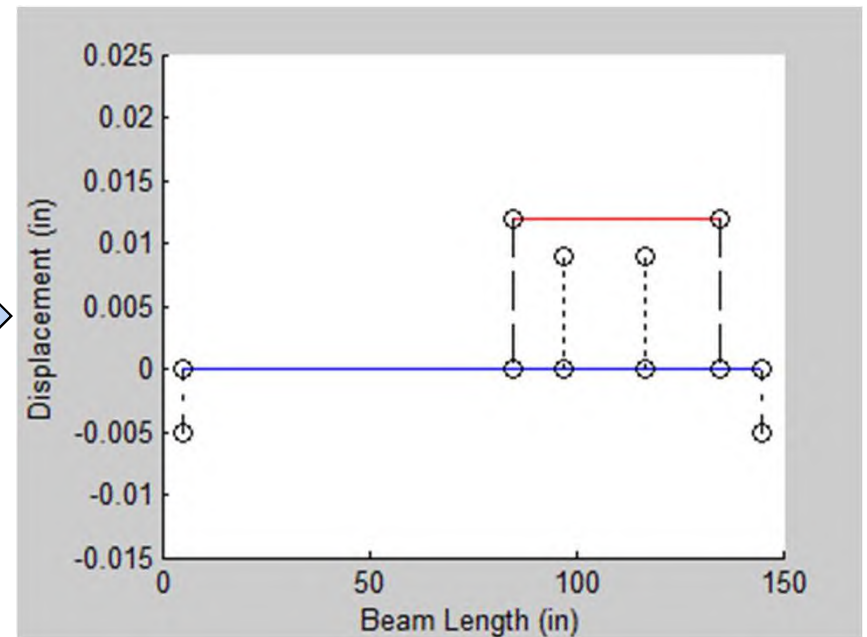


Response calculated using ERMT with a-space model

ERMT Response at a-space



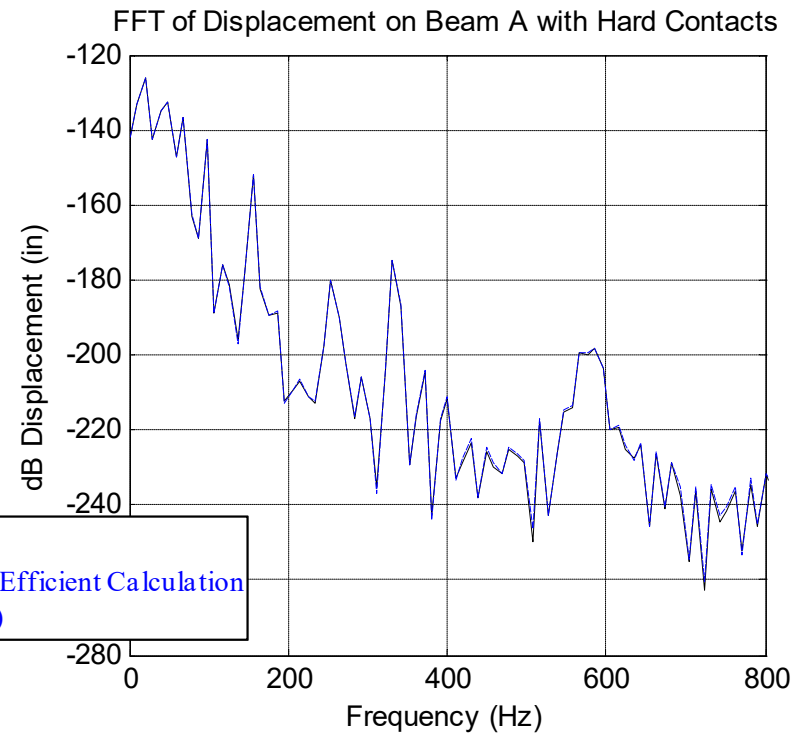
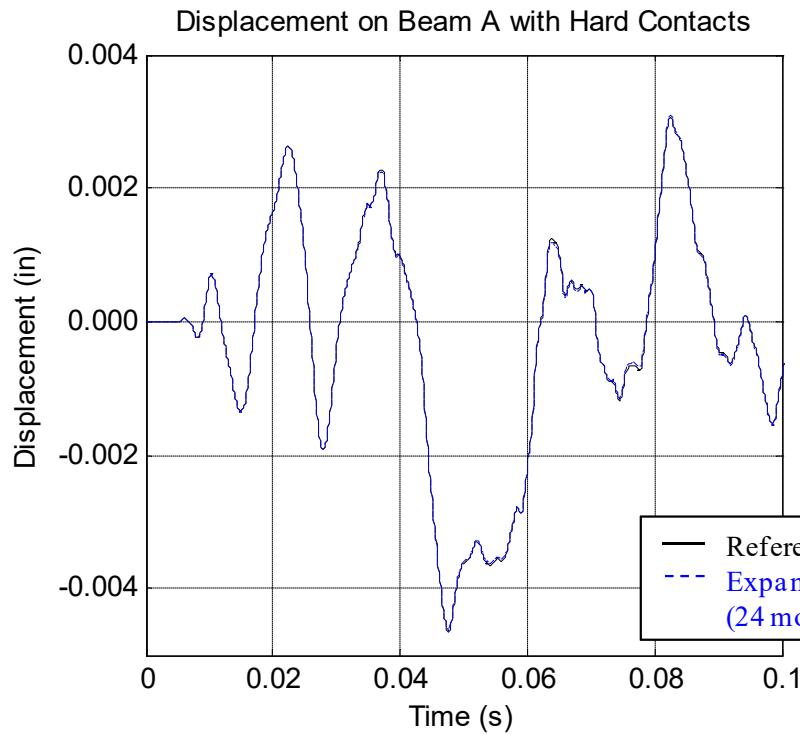
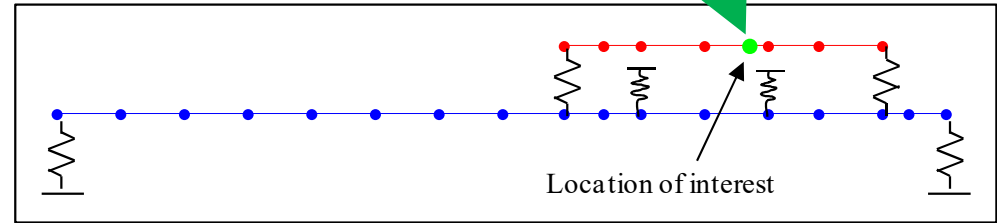
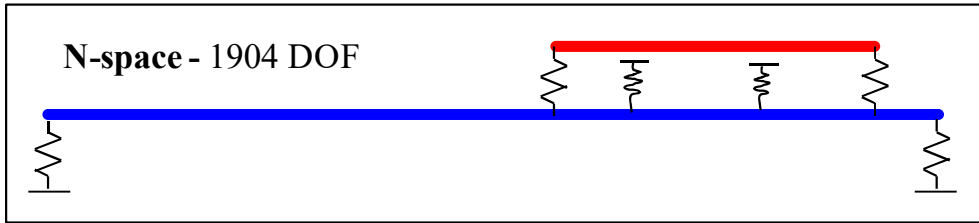
Expanded ERMT Response at N-space



Expansion is Key Here

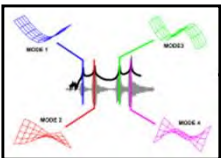
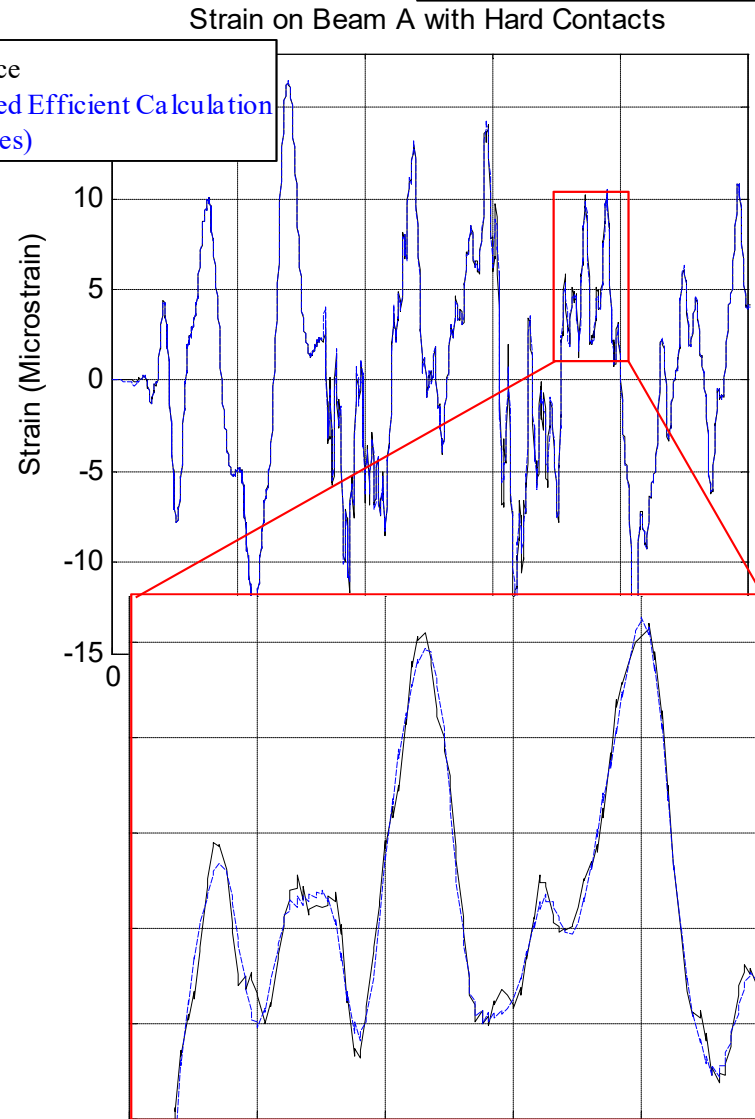
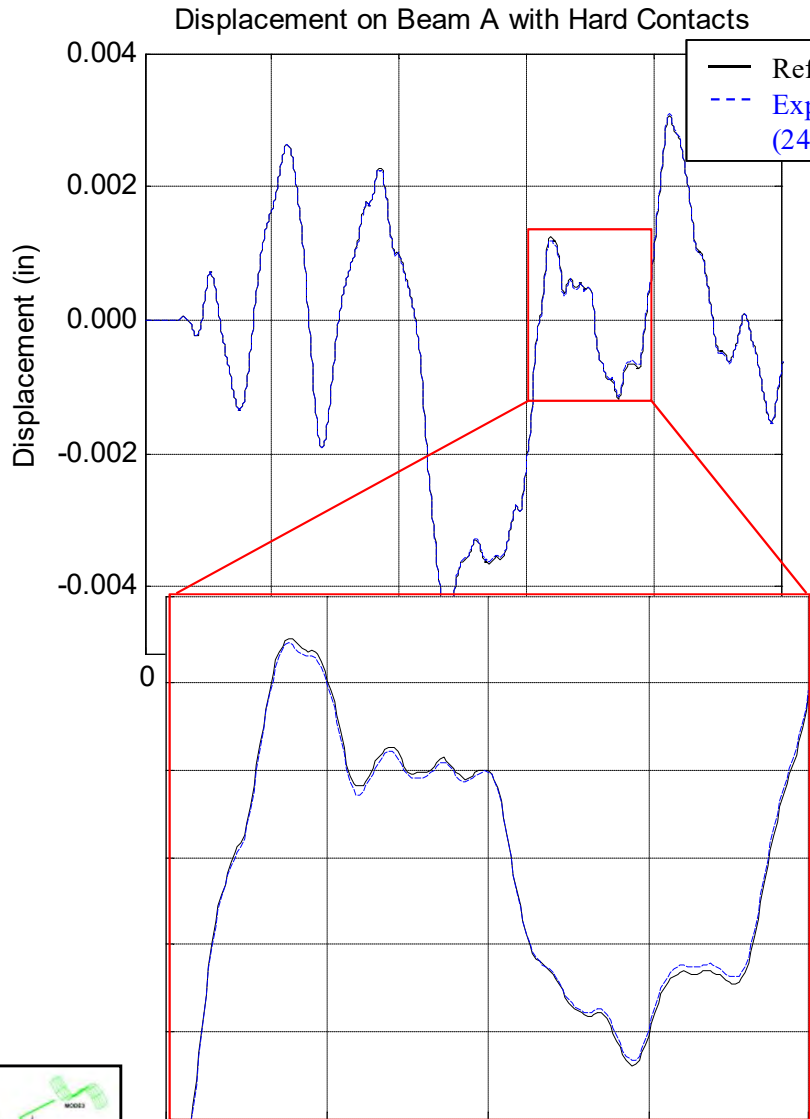
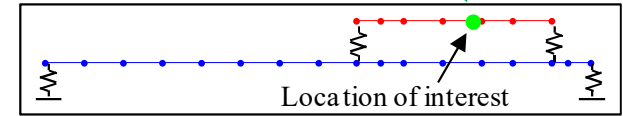
ANIMATION

Case 2 Displacement Results

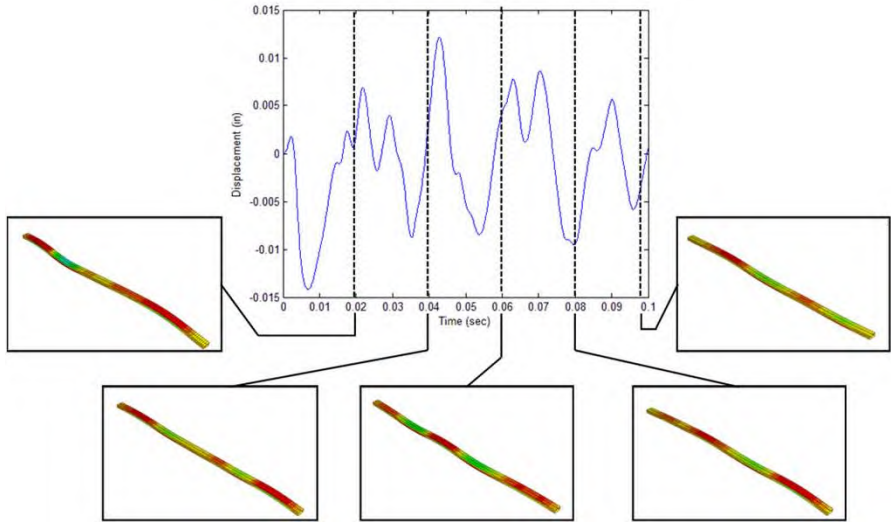
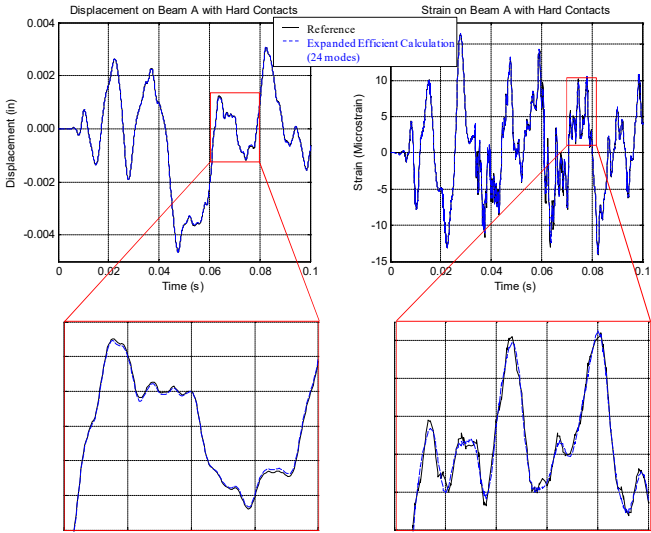
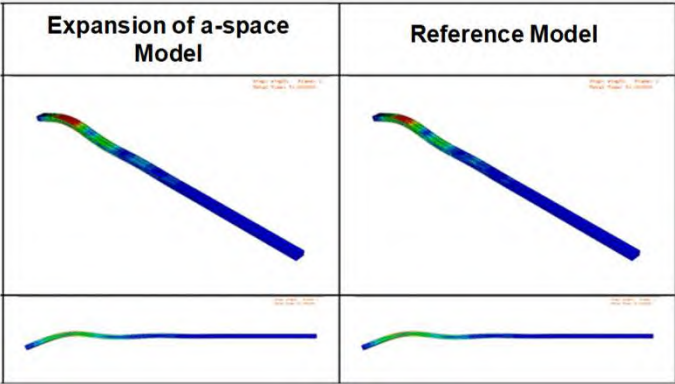
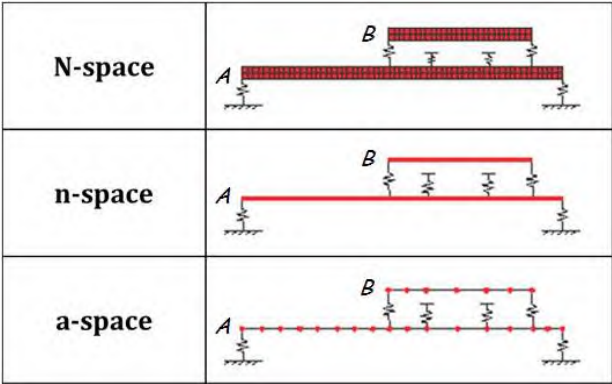


Model	# of DOF	Solution Time (sec)	Average MAC	Average TRAC
Full Space	1904	740.18	0.9998	0.9999
Reduced	24	0.28		

Case 2 Strain Results



Nonlinear Response - Expansion to Full 3D Models



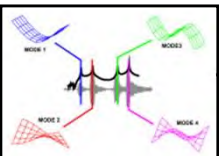
Full Field Damage Detection

Excerpts from 35 International Modal Analysis Conference papers

*Integrating Test and FEA for Enhanced
Structural Dynamic Modeling Applications*

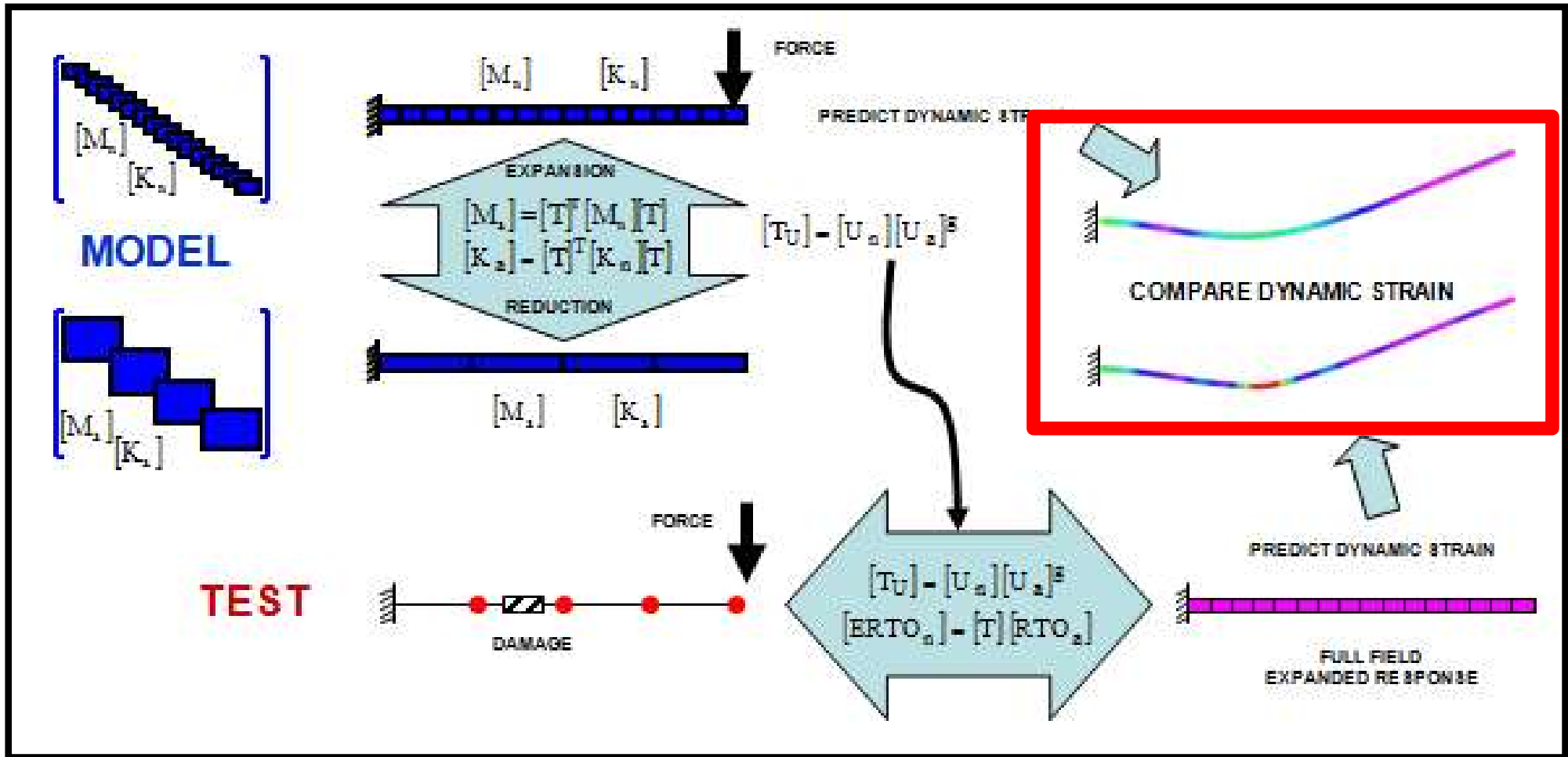
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*Peter Avitabile - Mechanical Engineering
Structural Dynamics And Acoustic Systems Lab*



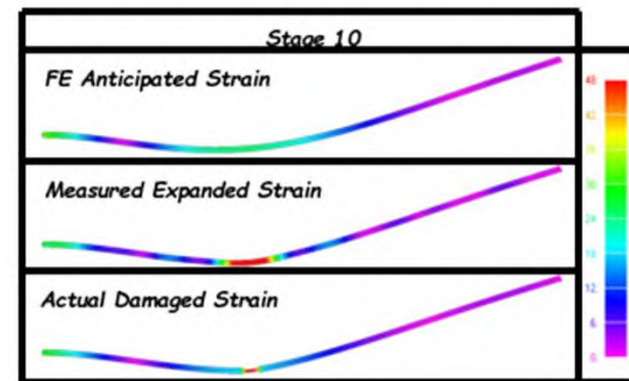
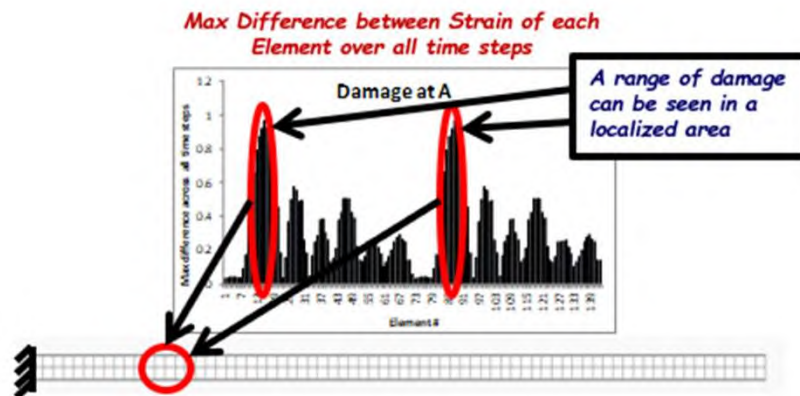
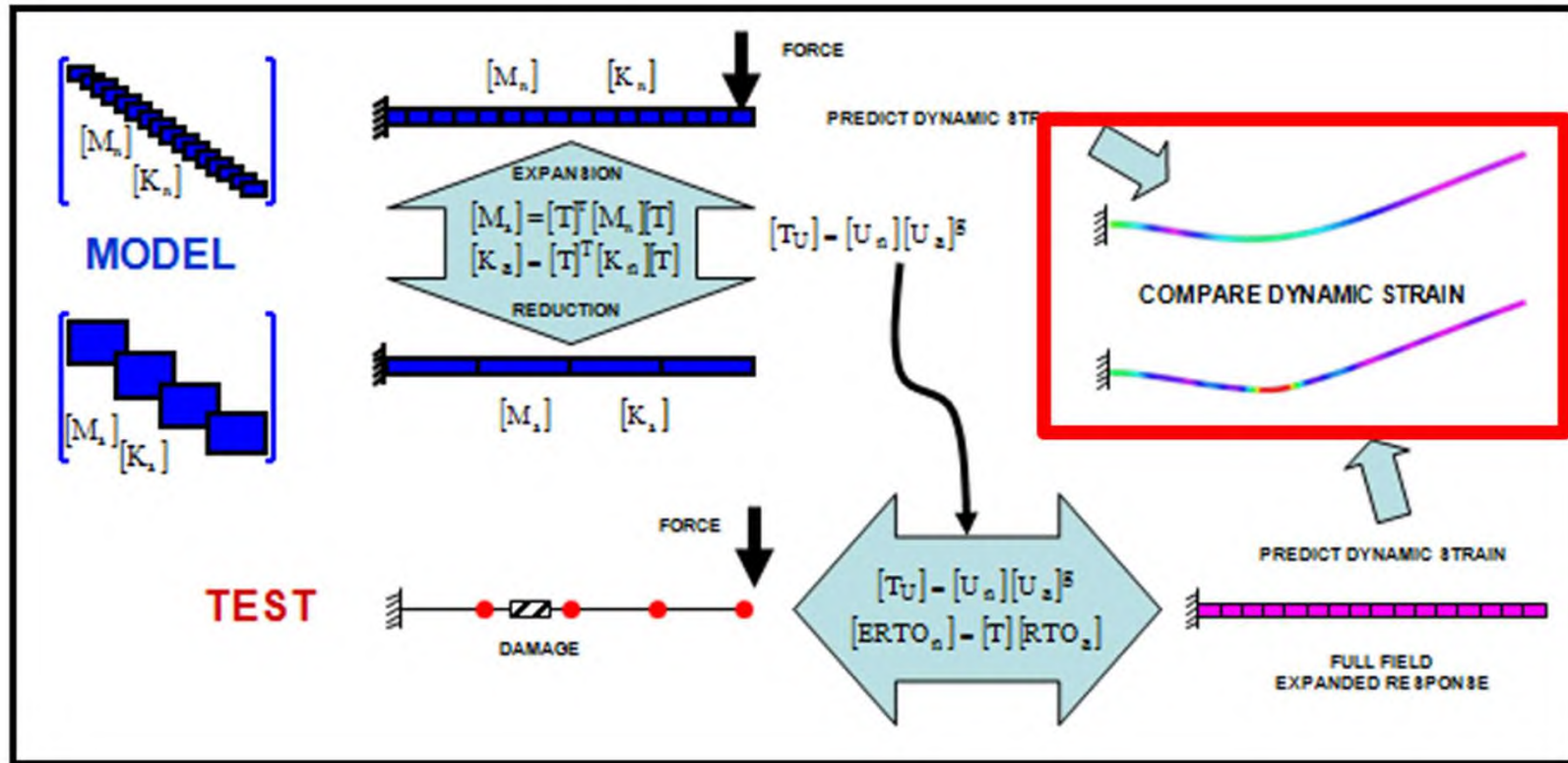
Damage Location

Eric Harvey, Justin Rudduck, Brett Daniels, Peter Avitabile



Expansion is Key Here

Current Work - Damage Detection - Overview

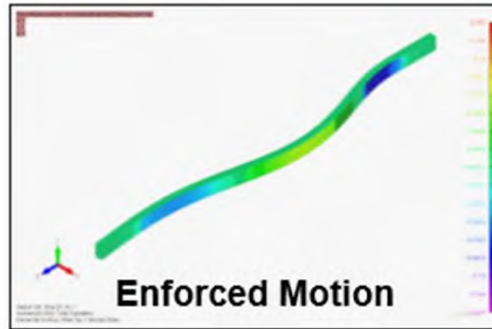


Displacement and Strain Approaches

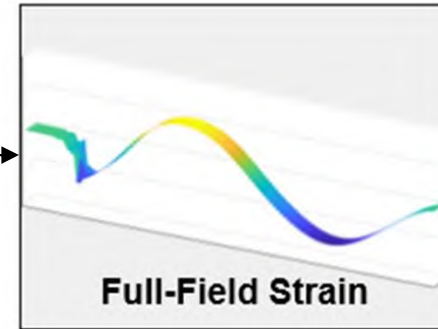
Displacement Expansion Enforced Motion

Expand to full-field displacement

$$\begin{bmatrix} X_n \end{bmatrix} = \begin{bmatrix} T_u \end{bmatrix} \begin{bmatrix} X_a \end{bmatrix}$$



Extract full-field strain



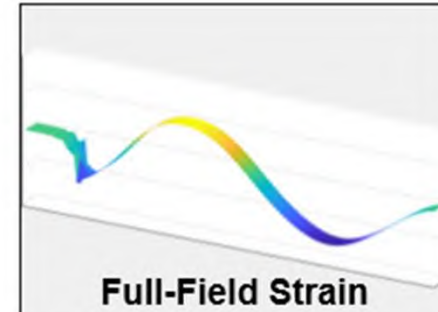
Direct Strain Expansion



Direct Expansion

$$\begin{bmatrix} \epsilon_n \end{bmatrix} = \begin{bmatrix} S_n \end{bmatrix} \begin{bmatrix} U_a \end{bmatrix}^g \begin{bmatrix} X_a \end{bmatrix}$$

from displacement to strain

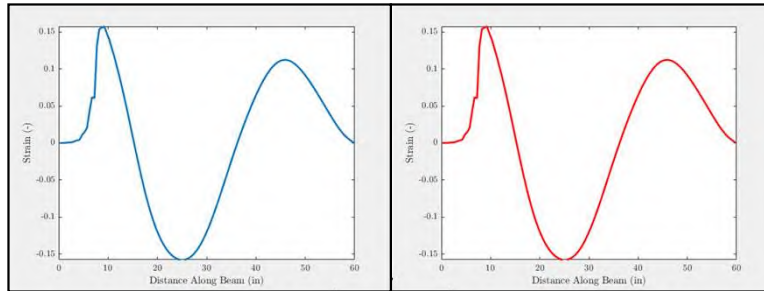


Damage Identification Index

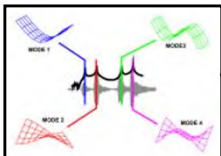
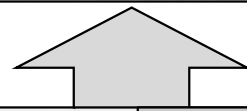
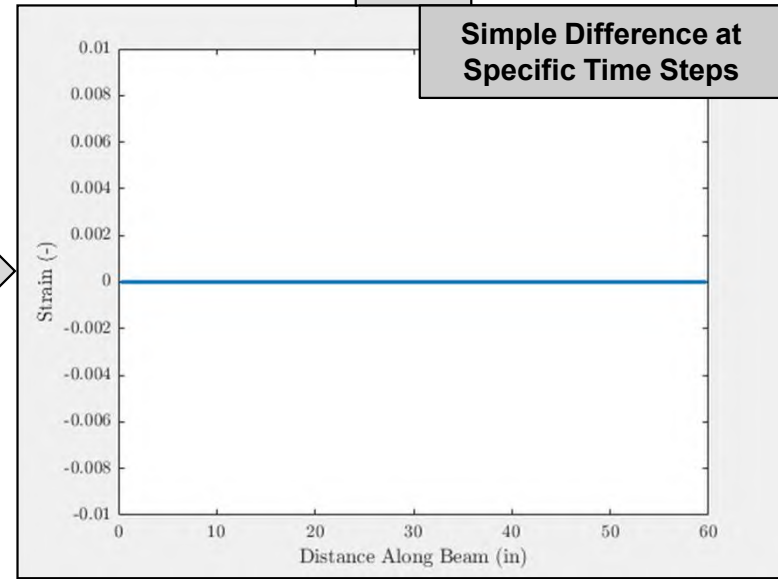
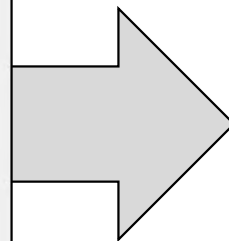
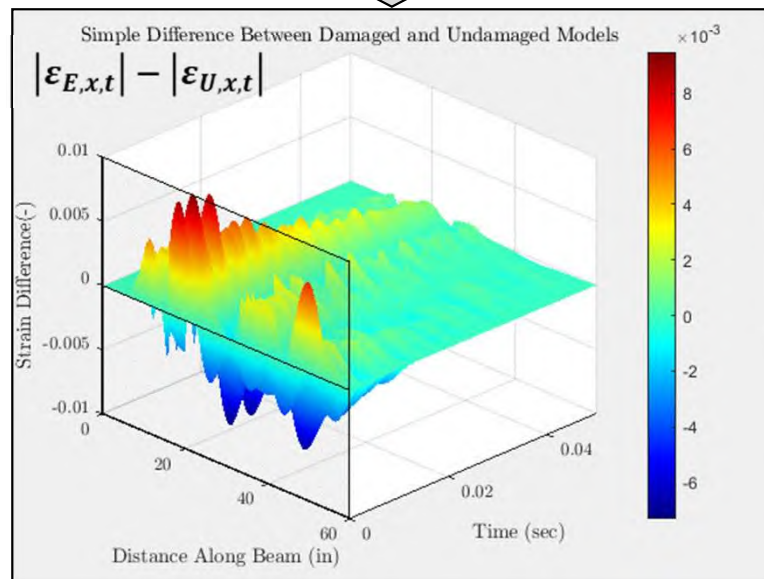
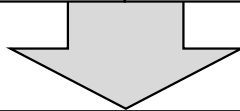
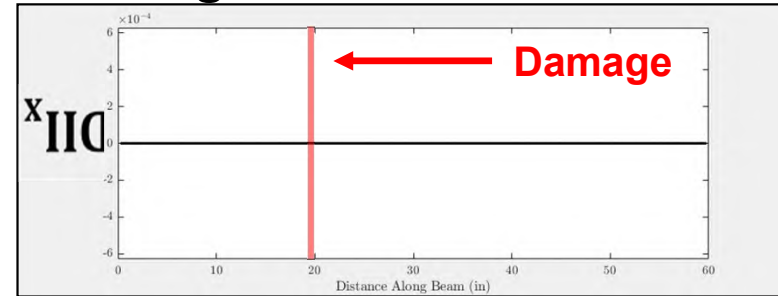


Undamaged

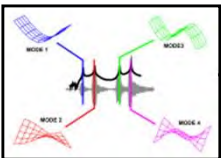
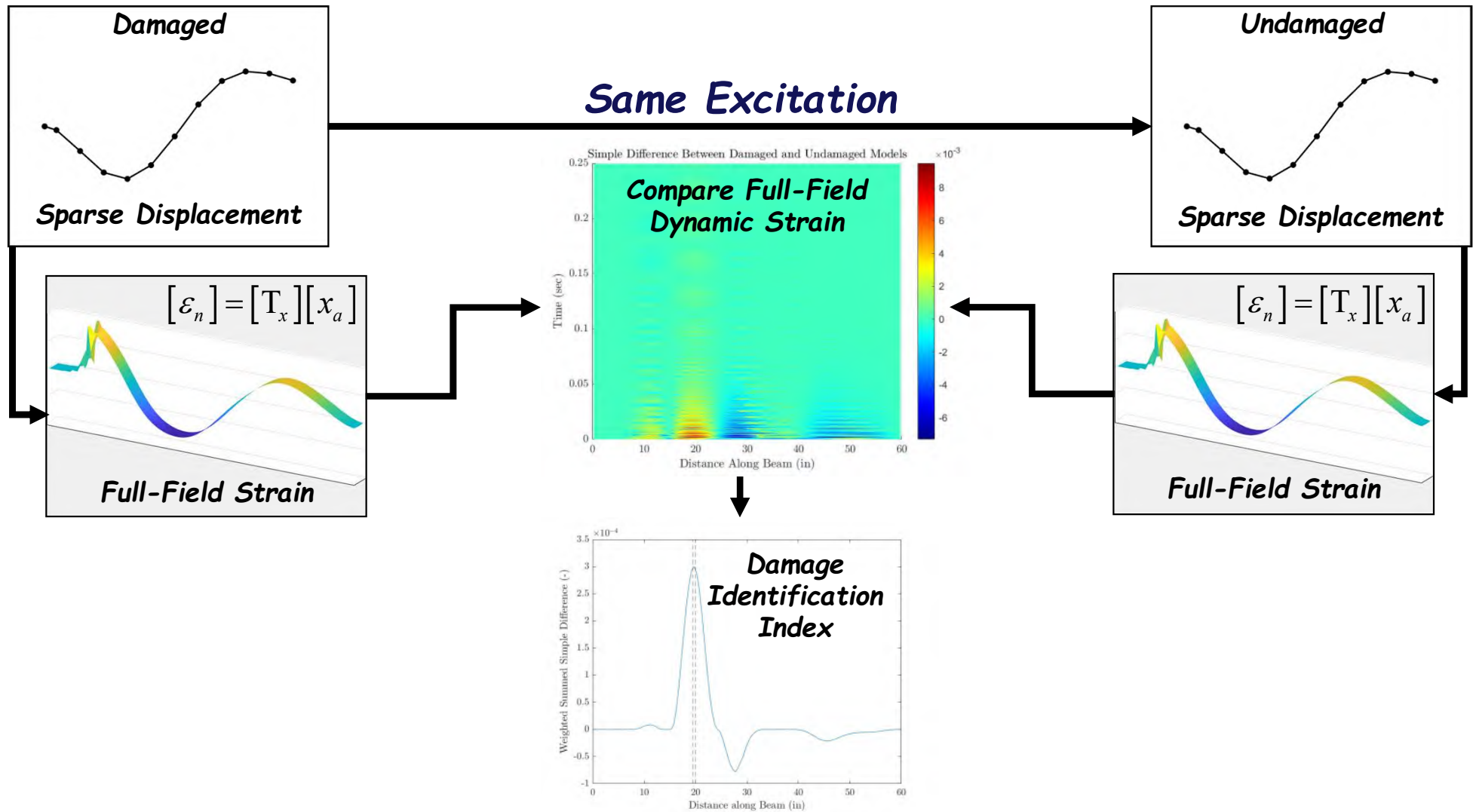
Damaged



Damage Identification Index



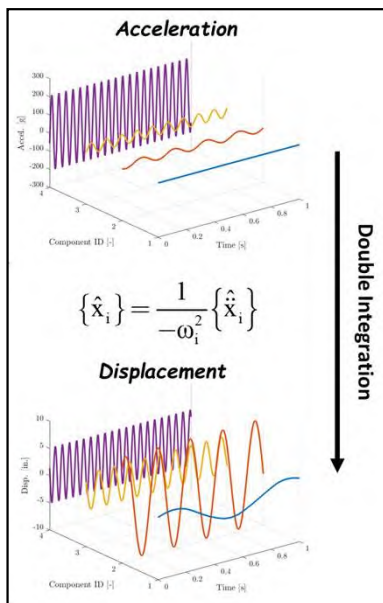
Damage Identification Index



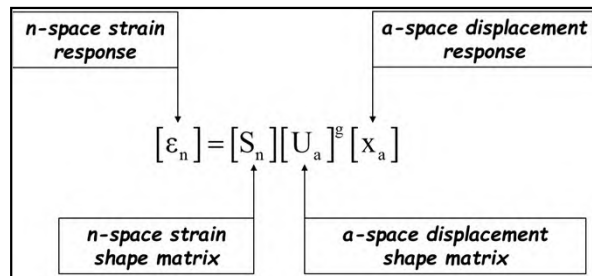
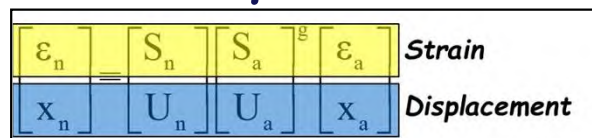
Summary of Approaches

But how does everything fit together?

Spectral Integration

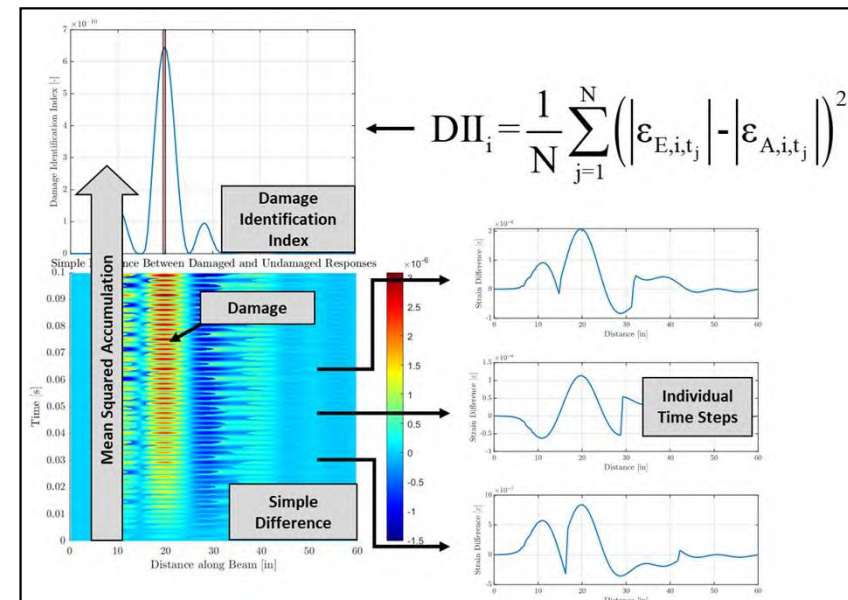


Mixed-domain Expansion



Displacement-Strain Expansion

Damage Identification Index



Force Reconstruction (beyond measured points)

"Reconstruction of Nonlinear Contact Forces Beyond Limited Measurement Locations Using an SVD Modal Filtering Approach", P.Logan, P.Avitabile, J.Dodson, *Experimental Techniques* (2020) accepted EXTE-D-19-00148R1, March 2020

"Reconstruction of External Forces Beyond Measured Points Using a Modal Filtering Decomposition Approach", P Logan, P Avitabile, J Dodson, *Experimental Techniques* (2019), accepted EXTE-D-18-00246R3

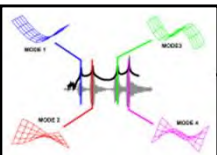
Seriously Relies on Expansion

Excerpts from 35 International Modal Analysis Conference papers

*Integrating Test and FEA for Enhanced
Structural Dynamic Modeling Applications*

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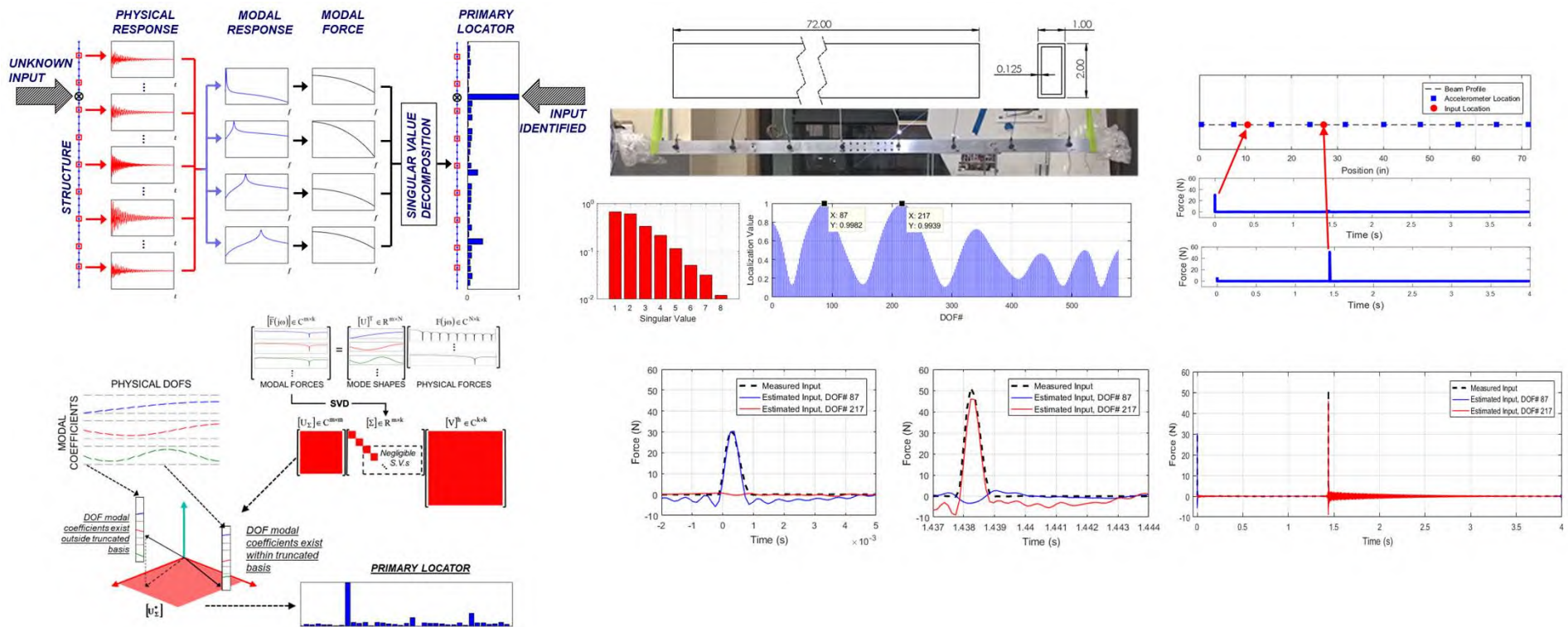
*Peter Avitabile - Mechanical Engineering
Structural Dynamics And Acoustic Systems Lab*



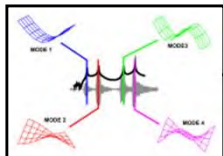


Structural Dynamics and Acoustic Systems Laboratory University of Massachusetts Lowell

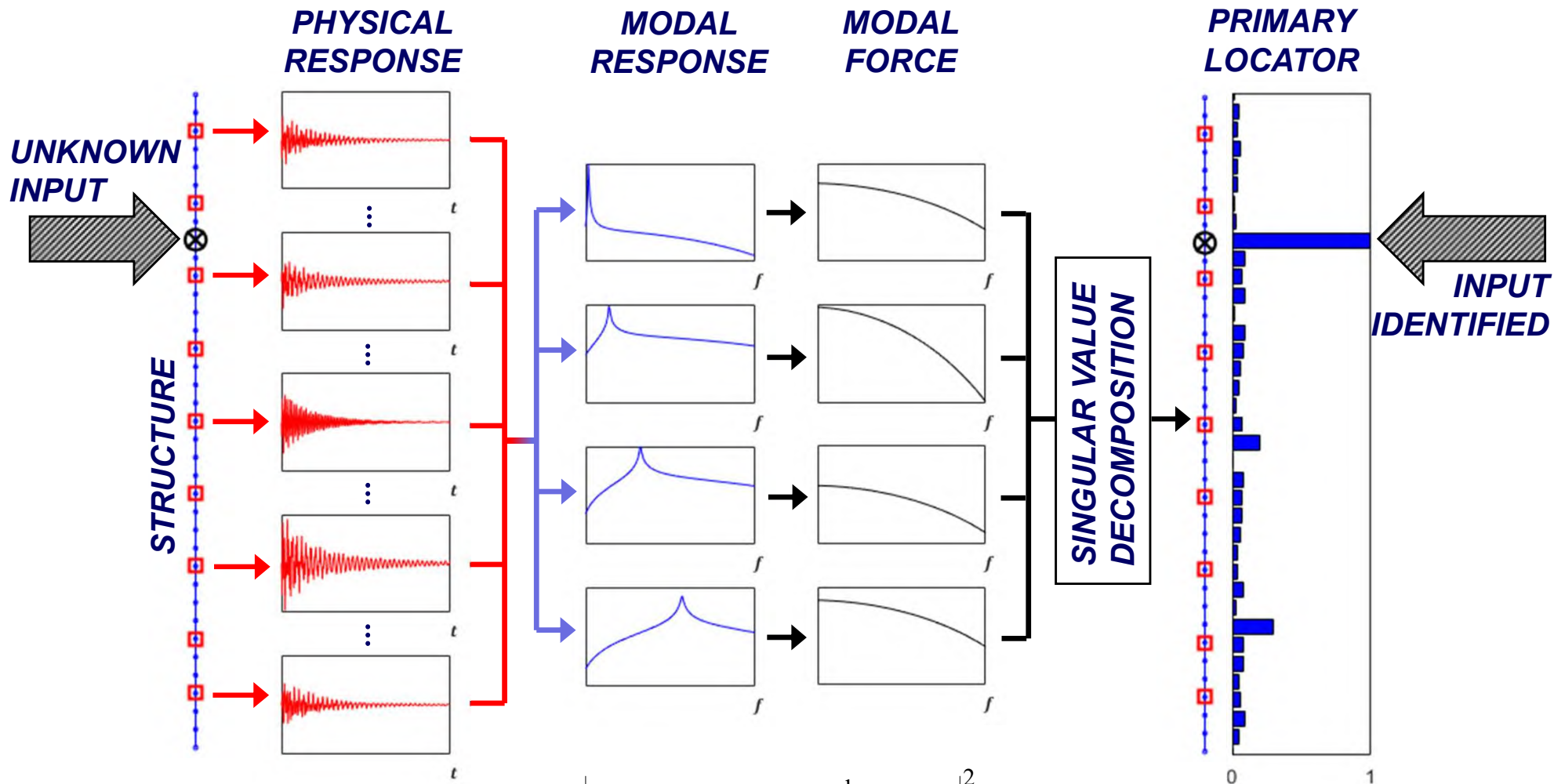
Force Reconstruction Beyond Measured Points



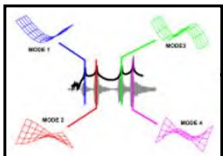
Patrick Logan, Peter Avitabile



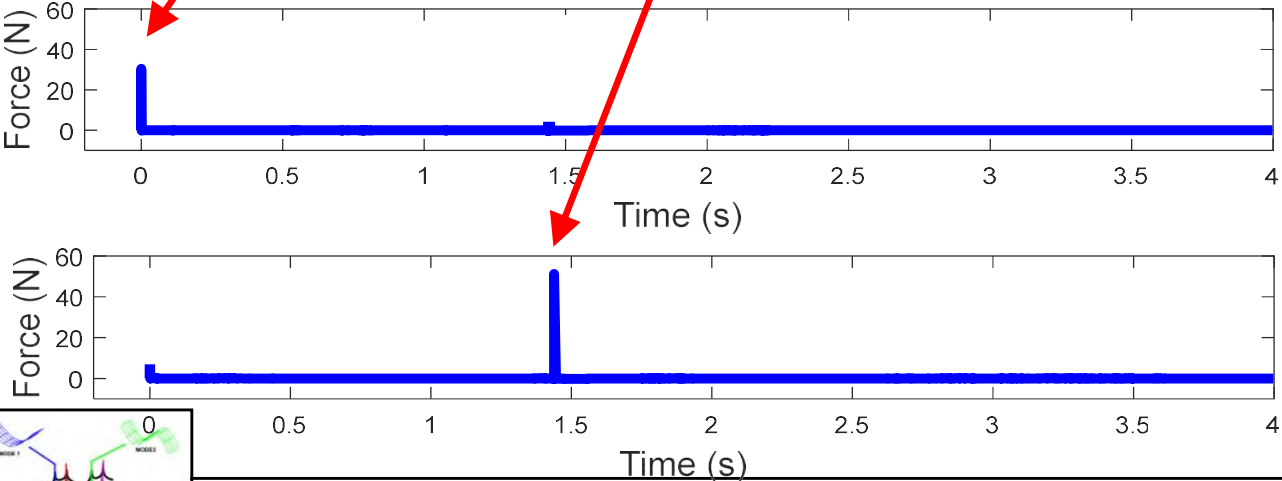
Source Identification from Modal Forces



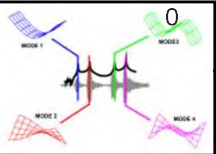
$$PL_i = \frac{\left| \left([U_{\Sigma}^* [U_{\Sigma}^*]^{\dagger} [U_{i,m}]^{\top} \right)^h [U_{i,m}]^{\top} \right|^2}{\left(\left([U_{\Sigma}^* [U_{\Sigma}^*]^{\dagger} [U_{i,m}]^{\top} \right)^h [U_{\Sigma}^* [U_{\Sigma}^*]^{\dagger} [U_{i,m}]^{\top} \right) ([U_{i,m}] [U_{i,m}]^{\top}) \right)}$$



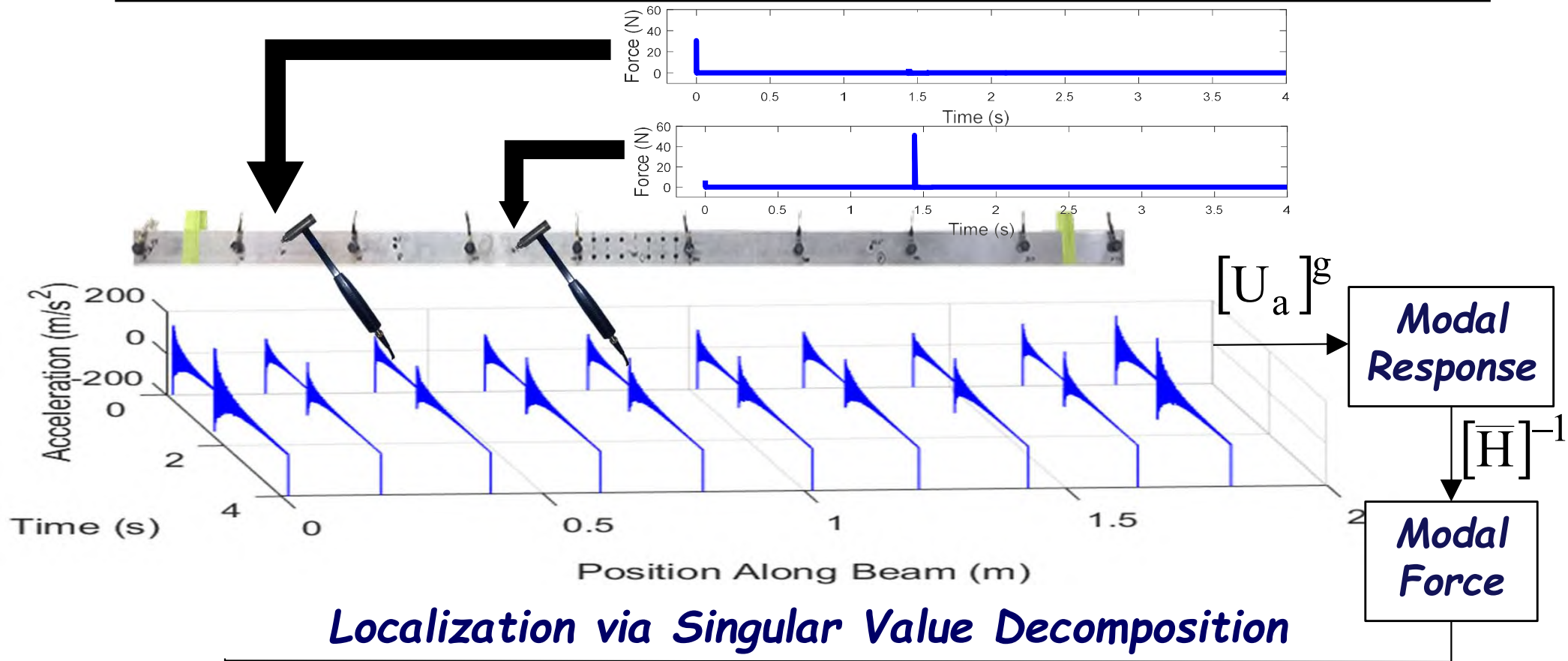
Input Estimation with a Free-Free Beam



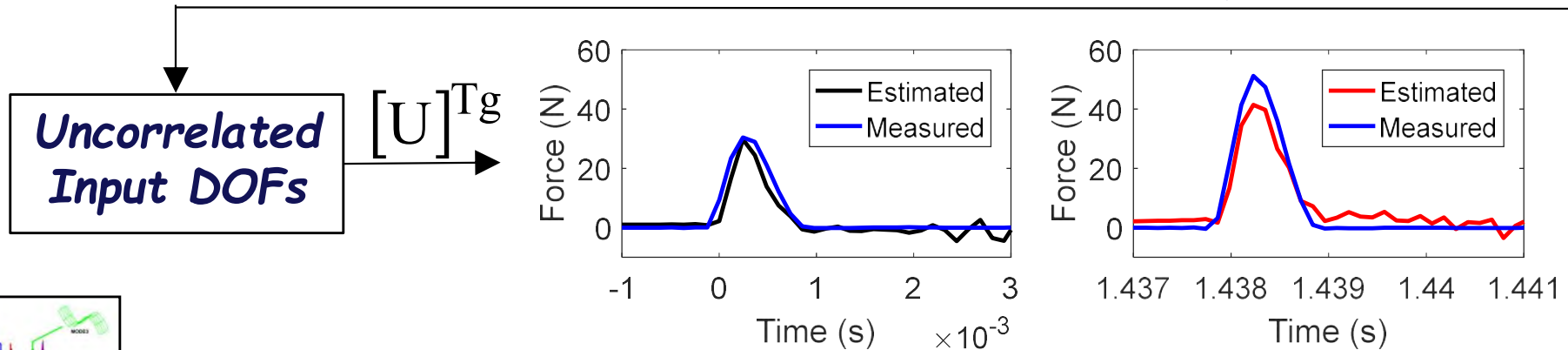
A free-free beam is subject to time staggered impulses at different locations.



Application of SVD Localization to a Free-Free Beam

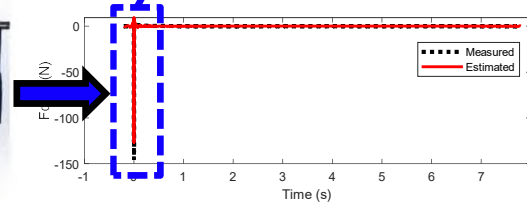
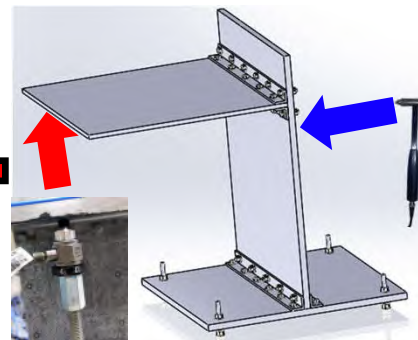
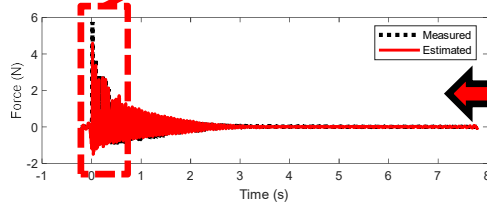
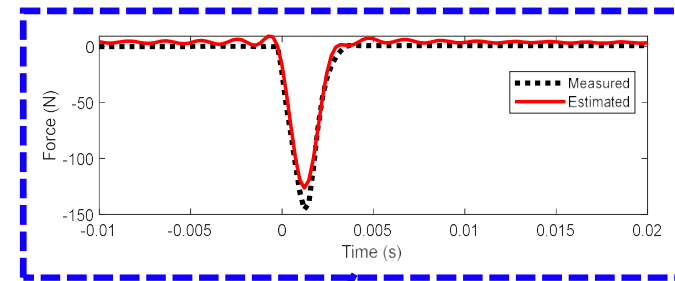
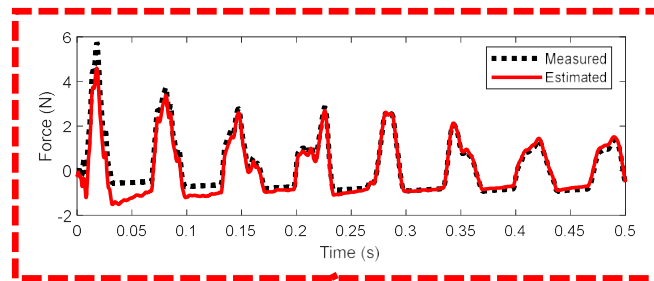


Localization via Singular Value Decomposition



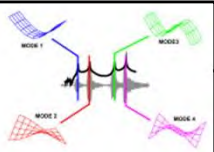
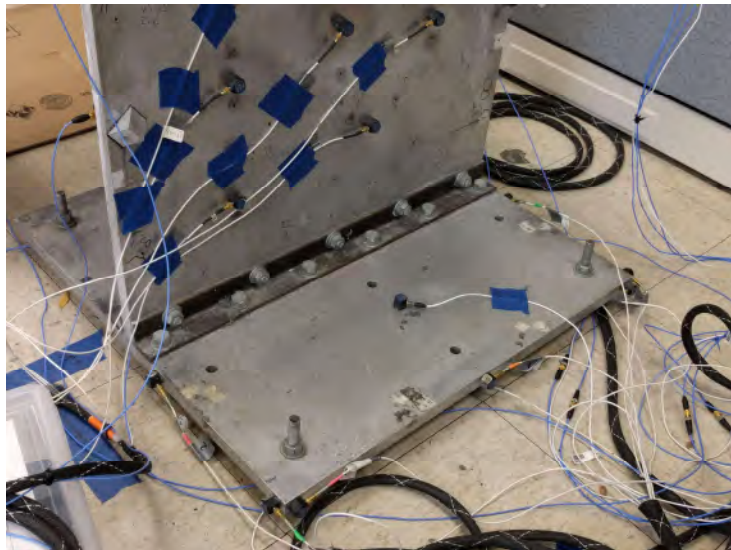
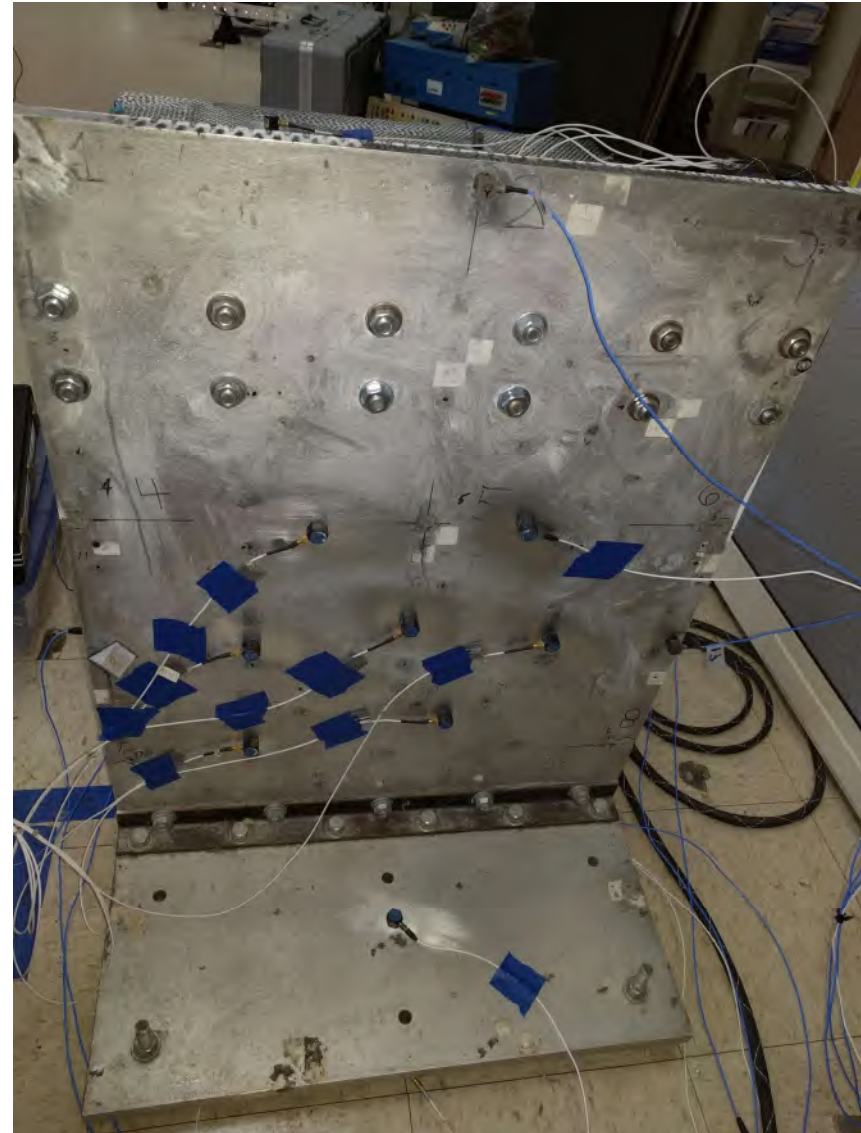
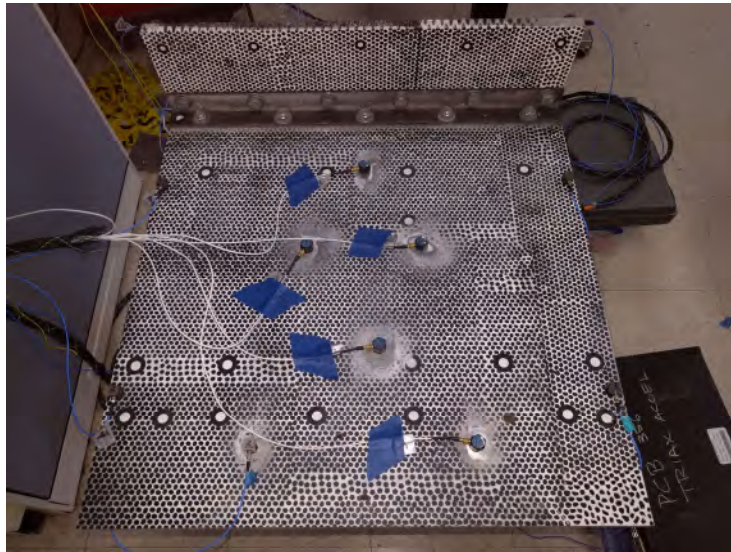


Validation of Force Reconstruction for Linear And Nonlinear System Response



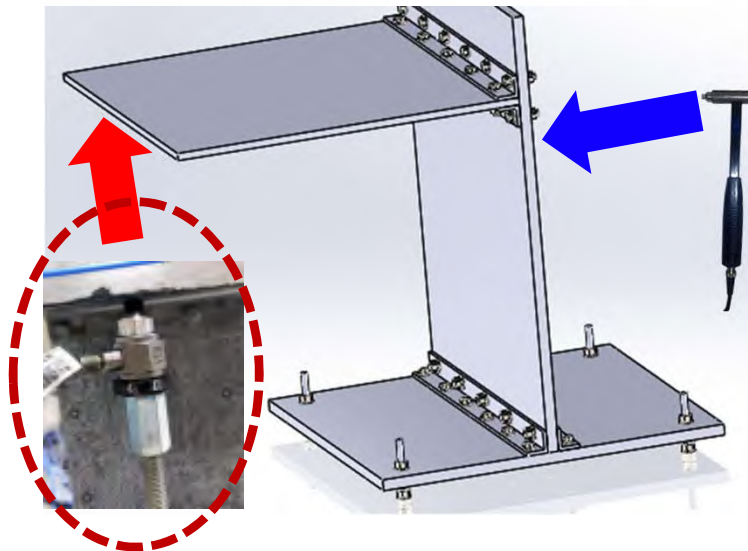
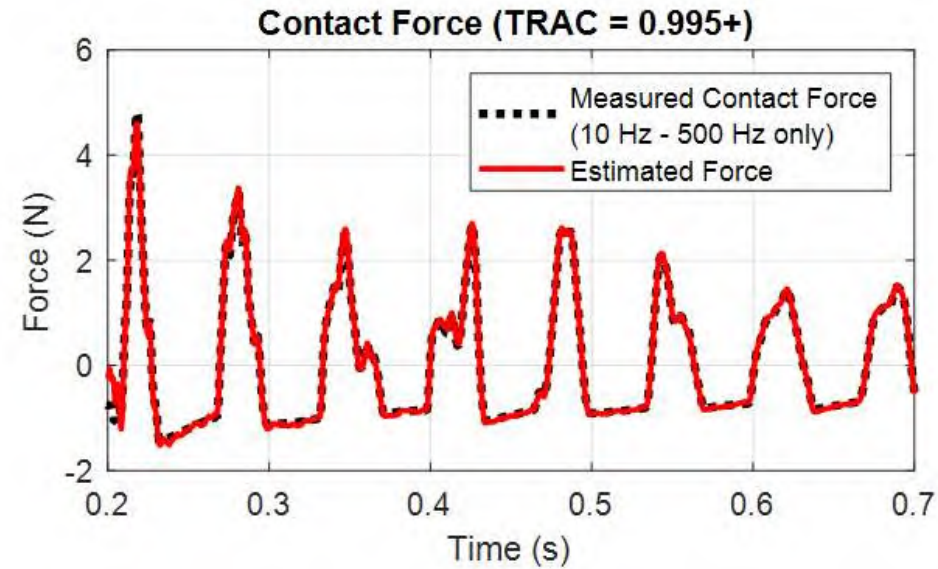
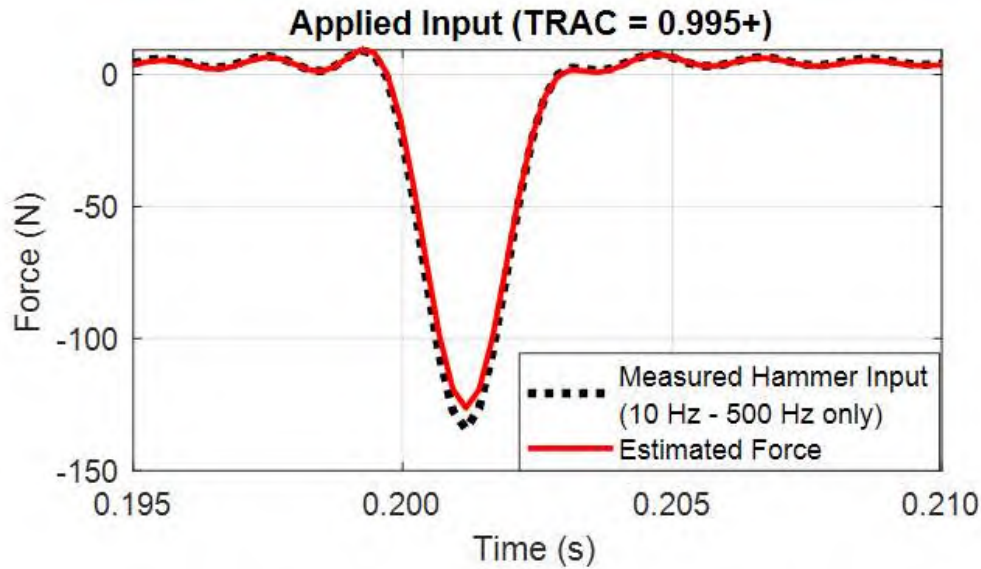
Patrick Logan, Deborah Fowler, Peter Avitabile

Experimental Characterization of Contact Force

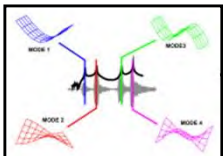


Experimental Characterization of Contact Force

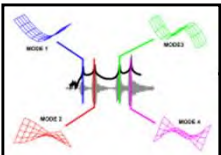
Localization within 1% (0.35 in)



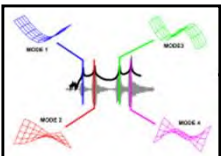
TRAC = 93%
Measured Max: 5.7 N
Estimated Max: 4.6 N



SPECIAL THANKS TO ALL THE STUDENTS

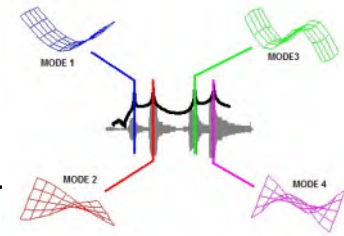


Special Thanks to the Students of MACL/SDASL





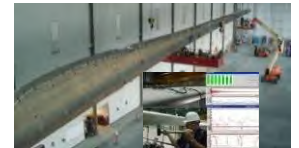
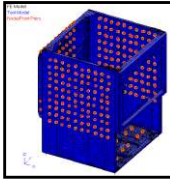
Dr. Peter Avitabile - Snapshot



Professor, Mechanical Engineering, Co-Director, Structural Dynamics and Acoustic Systems Laboratory
Vice President 2014 & President 2015/2016 for Society for Experimental Mechanics
Associate Editor - Handbook of Experimental Structural Mechanics

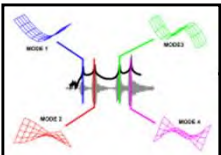
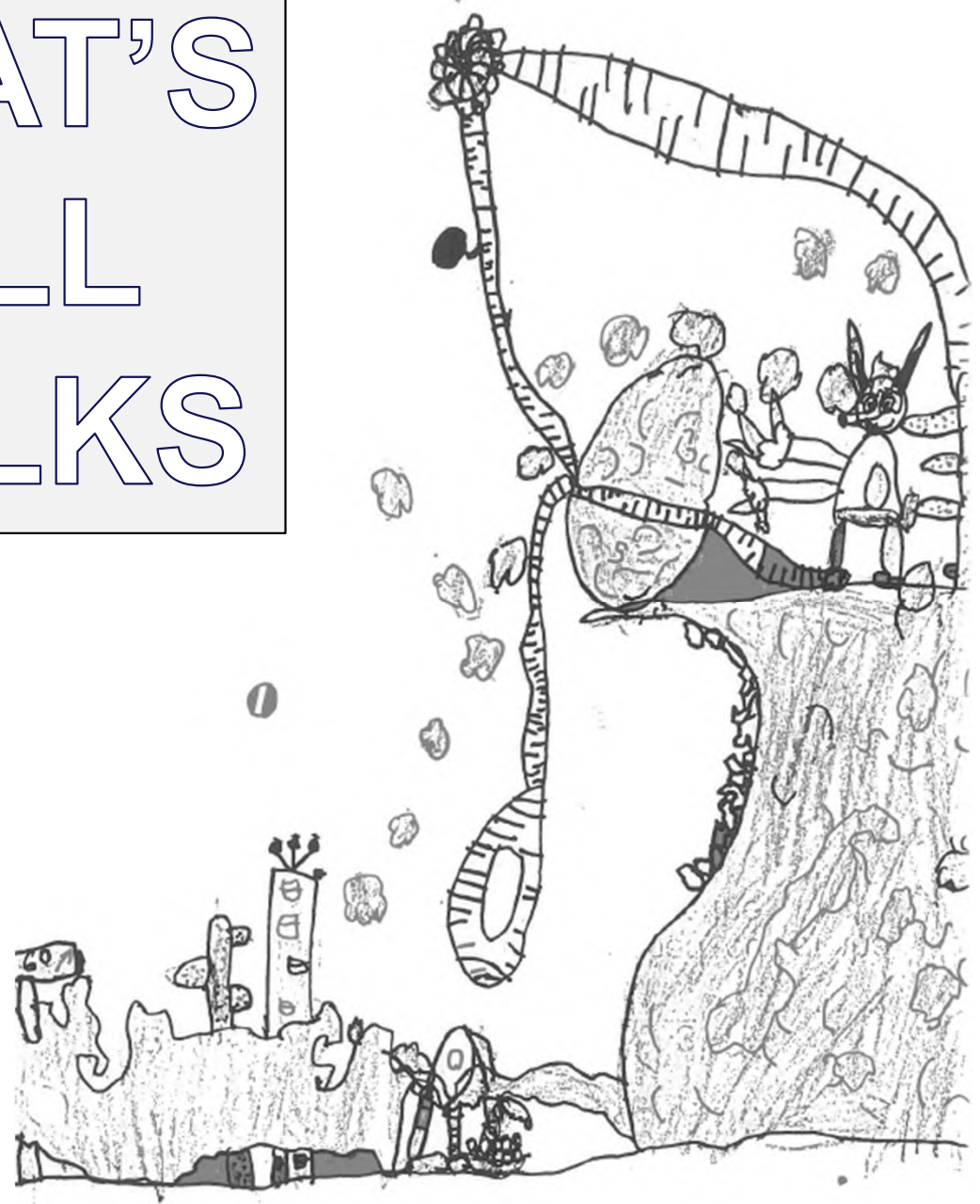
Pete has 50 years of experience in design & analysis using FEM and experimental techniques.
His main area of research is structural dynamics specializing in the areas of modeling, testing and correlation of analytical and experimental models along with advanced structural dynamic applications.

- Grants & Contracts: Numerous grants and contracts supporting the research in the Structural Dynamics and Acoustic Systems Laboratory including NSF Collaborative Research, Eglin AFB, DOT, DOE, ONR NUSC, Army Research Office, Pratt & Whitney, Bosch, DRS, MIT Lincoln Labs
- Publications: Journals since 2008 include 16 papers (3 under review and 2 in prep); conference papers published have been over 60 papers since 2008; over 40 were student presentations.
- UML Courses taught include a wide variety of courses: Mechanical Laboratory, Structural Dynamics, Dynamic Systems, Modal Analysis, Vibrations, Experimental Modal Analysis, Numerical Methods
- Actively involved in the Los Alamos Dynamics Summer School at the LANL since its inception in 2000.
- Provide a wide assortment of industry seminars for 30 years; extremely active in Society for Experimental Mechanics and the International Modal Analysis Conference for over 30 years; provided several Keynote Speeches for Conferences Worldwide
- Development of Reduced Order, Test Verified Component and System Models for Improved/Efficient Structural Dynamic Characterization; Expansion of Real Time Operating Data for Full Field Data
- Integration and Full Utilization of Measured Data from Optically Based Measurement Systems for Finite Element Modeling Applications
- Development of Highly Reduced Order Components for Linear and Nonlinear System Model Development along with Expansion for Full Field Dynamic Displacement and Full Field Dynamic Strain
- Compression of Massive Optical Data Sets with Advanced Processing



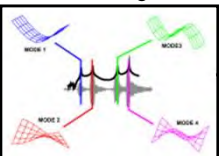


THAT'S ALL FOLKS



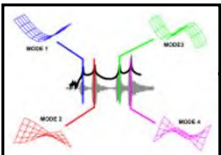
PhD Dissertations related to Model Reduction/Expansion, Nonlinear Response, Force Reconstruction, System Modeling

- **Fabio Piergentili**, *FRF Expansion for Unmeasured Translation and Rotational DOFs for Impedance Modeling Applications*
PhD, University of Massachusetts Lowell, Lowell, MA 1999; Advisor: John O'Callahan/Peter Avitabile
- **Hiromichi Tsuji**, *Application of Optimization Procedures using Phantom Connectivity Technique*
MS, University of Massachusetts Lowell, Lowell, MA 2002; Advisor: John O'Callahan
- **Pawan Pingle**, *Prediction of Full Field Dynamic Stress/Strain from Limited Set of Measured Data*
PhD, University of Massachusetts Lowell, Lowell, MA 2010; Advisor: Peter Avitabile
- **Javad Baqersad**, *A Non-Contacting Approach for Full Field Dynamic Strain Monitoring of Rotating Structures using the Photogrammetry, Finite Element and Modal Expansion Techniques*
PhD, University of Massachusetts Lowell, Lowell, MA 2015; Advisor: Chris Niezrecki/Peter Avitabile
- **Sergio Obando**, *Use of Expansion of Highly Reduced Order Models for the fir the Accurate Prediction of Full Field Dynamic Characteristics in the Forced Response of Linear and Nonlinear Systems and Components*
PhD, University of Massachusetts Lowell, Lowell, MA 2017; Advisor: Peter Avitabile
- **Patrick Logan**, *Force Estimation – Beyond Measured Points*
PhD, University of Massachusetts Lowell, Lowell, MA 2020; Advisor: Peter Avitabile
- **Debby Fowler**, *On the Use of Linear Dynamic Models with Limited Measured Data to Predict Nonlinear Response*
PhD, University of Massachusetts Lowell, Lowell, MA 2021; Advisor: Peter Avitabile
- **Brett Daniels**, *Strain Deformation Control of DUT in Consideration of Field to Laboratory Inconsistencies*,
PhD, University of Massachusetts Lowell, Lowell, MA (in progress); Advisor: Peter Avitabile & Alessandro Sabato
- **John Seymour**, *Development of a Blended Modal/Impedance System Modeling Approach for Dynamic Qualification*,
PhD, University of Massachusetts Lowell, Lowell, MA 2024; Advisor: Peter Avitabile



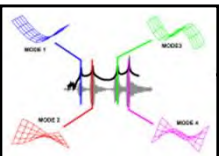
MS Theses related to Model Reduction/Expansion, Nonlinear Response, Force Reconstruction, System Modeling

- **Tracy Van Zandt**, *Development of Reduced Order Model Updating Strategies for Wing/Airframe Multibody Models*
MS, University of Massachusetts Lowell, Lowell, MA 2006; Advisor: Peter Avitabile
- **Nels Wirkkala**, *Impedance Based System Model Development for Target Mode Identification Reduced Model Updating*
MS, University of Massachusetts Lowell, Lowell, MA 2007; Advisor: Peter Avitabile
- **Adam Butland**, *A Hybrid Component Mode Synthesis Approach for System Modeling Applications*
MS, University of Massachusetts Lowell, Lowell, MA 2008; Advisor: Peter Avitabile
- **Dana Nicgorski**, *Investigation on Experimental Issues for Frequency Based Substructuring*
MS, University of Massachusetts Lowell, Lowell, MA 2008; Advisor: Peter Avitabile
- **Chris Chipman**, *Expansion for Real Time Operating Data*
MS, University of Massachusetts Lowell, Lowell, MA 2009; Advisor: Peter Avitabile
- **Lou Thibault**, *Development of Equivalent Reduced Order Model Technique for Linear Modal Models Interconnected with Nonlinear Connection Elements*
MS, University of Massachusetts Lowell, Lowell, MA 2012; Advisor: Peter Avitabile
- **Tim Marinone**, *Efficient Computational Nonlinear Dynamic Response using Modal Modification Response Technique*
MS, University of Massachusetts Lowell, Lowell, MA 2012; Advisor: Peter Avitabile
- **Julie Harvie**, *Computationally Efficient Response for Full Field Dynamic Strain Predictions*
MS, University of Massachusetts Lowell, Lowell, MA 2013; Advisor: Peter Avitabile
- **Tyler Doven**, *Strain Shape Expansion with Motion Magnification*
MS, University of Massachusetts Lowell, Lowell, MA 2019; Advisor: Peter Avitabile



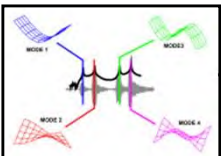
PhD Thesis related to FINE, Full Field Expansion, Strain Expansion, Optical

- **Jesus ReyesBlanco**, *Adjustment of Input Excitation to Account for Fixture-Test Article Dynamic Coupling Effects*, PhD, University of Massachusetts Lowell, Lowell, MA 2017; Advisor: Peter Avitabile
- **Tyler Doven**, *Strain Shape Expansion with Motion Magnification*, MS, University of Massachusetts Lowell, Lowell, MA 2019; Advisor: Peter Avitabile & Zhu Mao
- **Brandon Zwink**, *Dynamic Response Matching from Field to Laboratory Replication Methodology*, PhD, University of Massachusetts Lowell, Lowell, MA 2020; Advisor: Peter Avitabile
- **Ryan Schultz**, *Improving Efficiency of Multi-Shaker and Combined Shaker-Acoustic Vibration Tests*, PhD, University of Massachusetts Lowell, Lowell, MA 2020; Advisor: Peter Avitabile
- **Christopher Page**, *A Unified Substructuring Approach to Fixture Neutralization*, PhD, University of Massachusetts Lowell, Lowell, MA 2021; Advisor: Peter Avitabile
- **Daniel Rohe**, *Image Processing Techniques for Structural Dynamic Tests using Radiographic Images*, PhD, University of Massachusetts Lowell, Lowell, MA 2022; Advisor: Peter Avitabile & Zhu Mao
- **Brett Daniels**, *Strain Deformation Control of DUT in Consideration of Field to Laboratory Inconsistencies*, PhD, University of Massachusetts Lowell, Lowell, MA (in progress); Advisor: Peter Avitabile & Alessandro Sabato
- **John Seymour**, *Development of a Blended Modal/Impedance System Modeling Approach for Dynamic Qualification*, PhD, University of Massachusetts Lowell, Lowell, MA 2024; Advisor: Peter Avitabile
- **Troy Skousen**, *Impedance and Modal Based FINE Approaches Addressing Vibration Qualification Testing*, PhD, University of Massachusetts Lowell, Lowell, MA (in progress); Advisor: Peter Avitabile & Alessandro Sabato



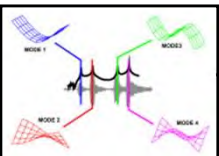
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- “Direct Expansion from Displacement to Strain Using Equivalent Transformation of Modal Coordinates”, B.Daniels, A.Sabato, R.Joshua, P.Avitabile, *Proceedings of the Forty-First International Modal Analysis Conference*, Austin, Texas, Feb 2023
- “Strain-based Damage Detection Using Cross-domain Hybrid Expansion Method”, B.Daniels, A.Sabato, R.Joshua, P.Avitabile, *Proceedings of the Forty-First International Modal Analysis Conference*, Austin, TX, Feb 2023
- “System Modeling Through the Coupling of Impedance and Modal Based Models”, J.Seymour, R.Joshua, P.Avitabile, *Proceedings of the Forty-First International Modal Analysis Conference*, Austin, TX, Feb 2023
- “A New Residual Compensation Method for Estimating Effects of Residual Modes for FRF Synthesis”, J.Seymour, R.Joshua, P.Avitabile, *Proceedings of the Forty-First International Modal Analysis Conference*, Austin, Texas, Feb 2023
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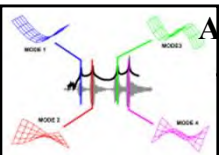
Papers related to FINE, Full Field Expansion, Strain Expansion, Optical

- **“Shape-Constrained Input Estimation for Multi-Shaker Vibration Testing”, R.Schultz, P.Avitabile, Proceedings of the Thirty-Eighth International Modal Analysis Conference, Houston, Texas, Feb 2020**
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